

# Overview of High Efficiency, Compact Rotating Detonation Rocket Engines

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Presenter:

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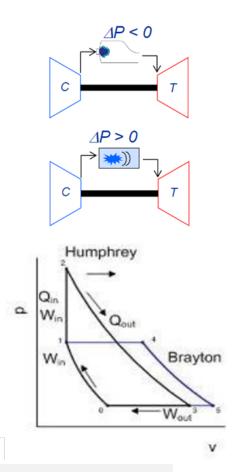
### **Introducing Supersonic Combustion**

**Rotating Detonation Engines** versus conventional:

Convention gas turbines combustion results in a pressure loss across the combustor (Brayton cycle)

# Pressure gain with constant volume combustion (Humphrey cycle)

 Deflagration or detonation pressure wave increases pressure and peak temperatures at turbine inlet reduced entropy production during combustion.





Improves fuel efficiency by up to 25%

**Source**: Richards, Geo. National Energy Technology Laboratory. New Development in Combustion Technology. 2014 Princeton CEFRC Summer School on Combustion

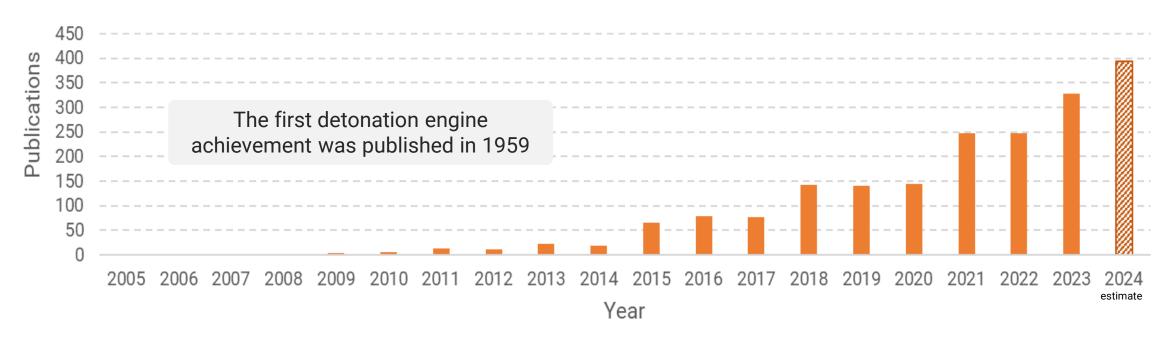


Supersonic wave travels around the annular chamber at 4,500 mph

**Source**: Oak Ridge National Laboratory. https://www.olcf.ornl.gov/2020/08/26/realizing-the-dream-of-rotating-detonation-engines-through-an-olcf-netl-ge-and-university-of-michigan-collaboration/

## **Detonation Engine Momentum!**





# NASA Validates Revolutionary Propulsion Design for Deep Space Missions

By Ray Osorio

As NASA takes its first steps toward establishing a long-term presence on the Moon's surface, a team of propulsion development engineers at NASA have developed and tested NASA's first full-scale rotating detonation rocket engine, or RDRE, an advanced rocket engine design that could significantly change how future propulsion systems are built.



Next-generation missile propulsion systems

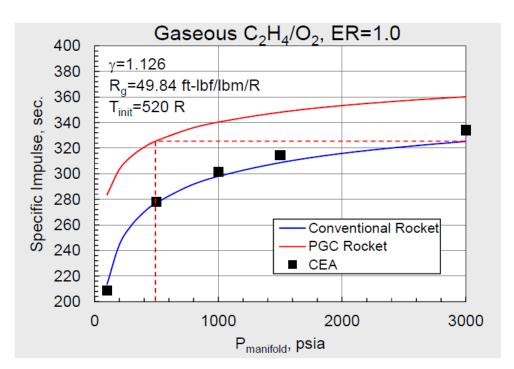




## **NASA Progress and Examples of RDRE Benefits**

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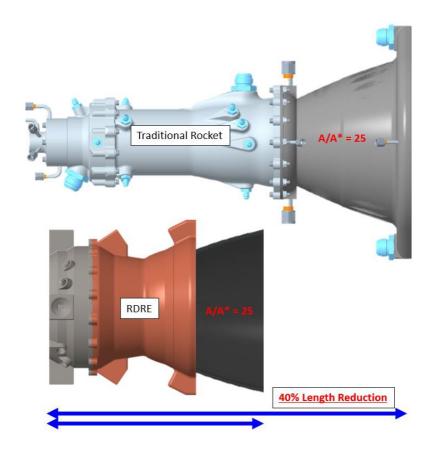
#### More efficient!



PGC Rocket at  $P_{\text{manifold}}$  of 488 psia Delivers Same  $I_{\text{sp}}$  as Conventional Rocket at  $P_{\text{manifold}}$  of 3000 psia

Smaller or Even No Pumps → Better T/W

#### More compact!



#### **Mission level benefits:**

$$\Delta v = v_{
m e} \ln rac{m_0}{m_f}$$

Rocket Equation and Aircraft Range equations have an exponential dependency on fuel efficiency

$$R = \frac{\eta}{SFC} \frac{C_L}{C_D} \ln \left( \frac{W_{initial}}{W_{final}} \right)$$

**Source**: Paxon, Daniel. NASA Glenn Research Center. *A Case for Basic Rotating Detonation Engine Research*. DARPA RDE Stakeholders Day. 26 May 2016

**Source**: Teasley, et al. NASA Marshal Space Flight Center. *Current State of NASA Continuously Rotating Detonation Cycle Engine Development*. 2023 AIAA propulsion and Energy Forum.

## **Supersonic Combustion Technology – it works!**













Nobel builds and tests supersonic combustion engines for commercial and aerospace

Phase II SBIR: Pulsed Detonator for Scramjet Ignition and Combustion Augmentation using On-board Resources



- Shell contract testing Hydrogen/Air RDEs for use in commercial energy production
- Long duration ( > 1-min, until it ran out of fuel) sustained detonation waves

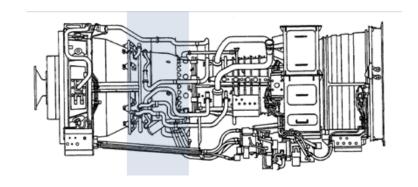




## Other Activities: Drop-in Combustor Replacements in Energy & nobel



**Detonation engine** simulations in digital twins of popular gas turbines











**CF-6 Turbofan Jet Engine** Introduced 1971 and used on Boeing 767/777 and Airbus A300/A330



LM2500 / LM6000 Repurposed as "aero-derivative" gas-fired powerplants



LM2500 Marine Gas Turbine Predominant turbine used in the U.S. Navy and 29 other navies

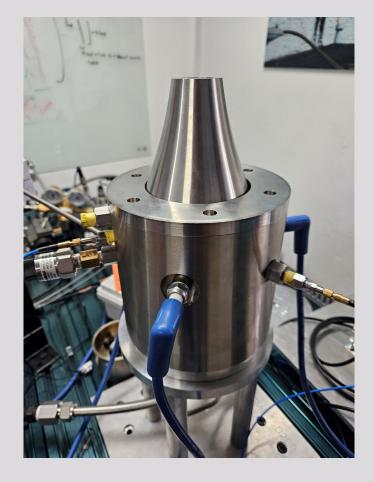
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## Nobel-UA Projects: Rockets and Other Things that Make Fire & nobel





Optimizing unique aerospike nozzles for rotating detonation engines



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## **Additional Information**

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