



Overview of BCT Microsats

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About Blue Canyon Technologies

Founded: 2008

First space hardware deliveries: 2012

- AFRL, NASA, JPL

Agile small satellites:

- Spacecraft
 - Cubesats (3U, 6U, 12U)
 - Microsats to >350kg
- Components/Subsystems
 - Star Trackers
 - Reaction Wheels
 - ADCS

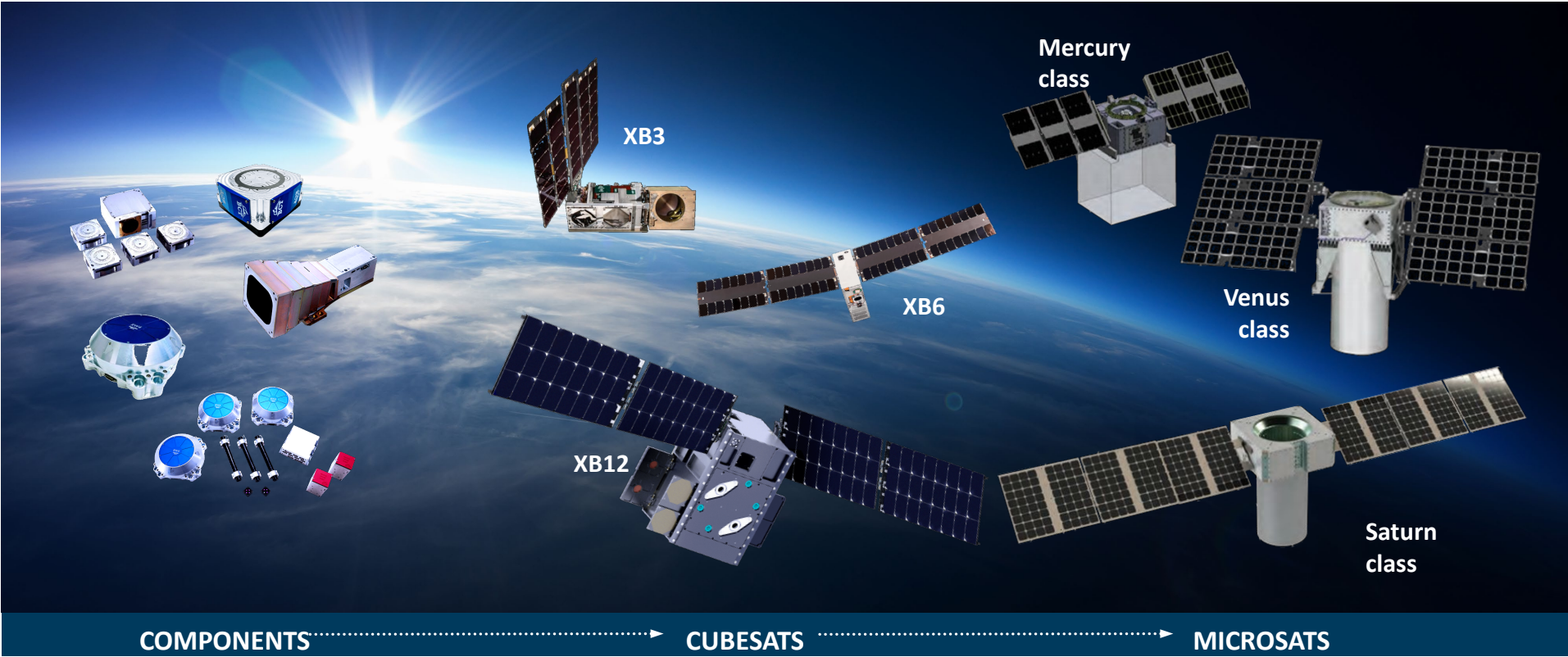
Staff & Facilities:

- > 325 employees
- > 120,000 sq-ft total facilities
- End-to-End Vertical Integration


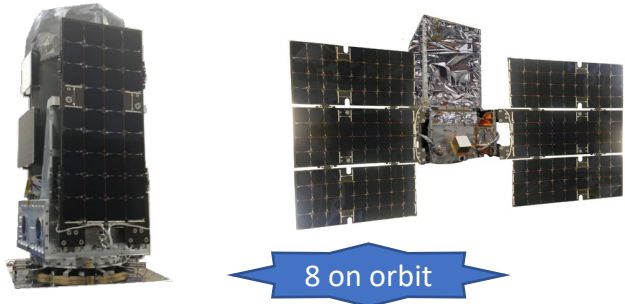
Recently acquired by Raytheon as a wholly-owned subsidiary !




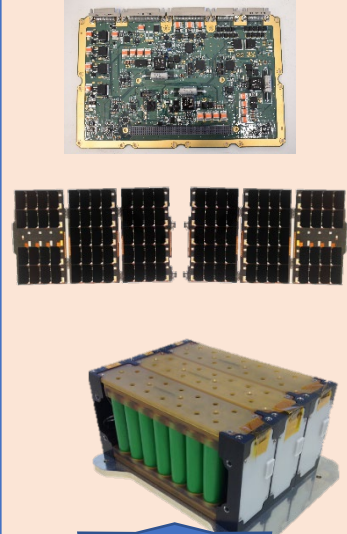



BCT Capabilities



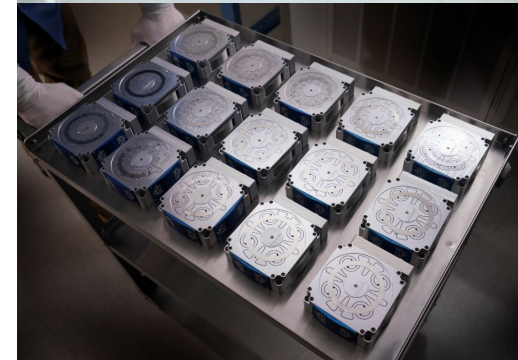
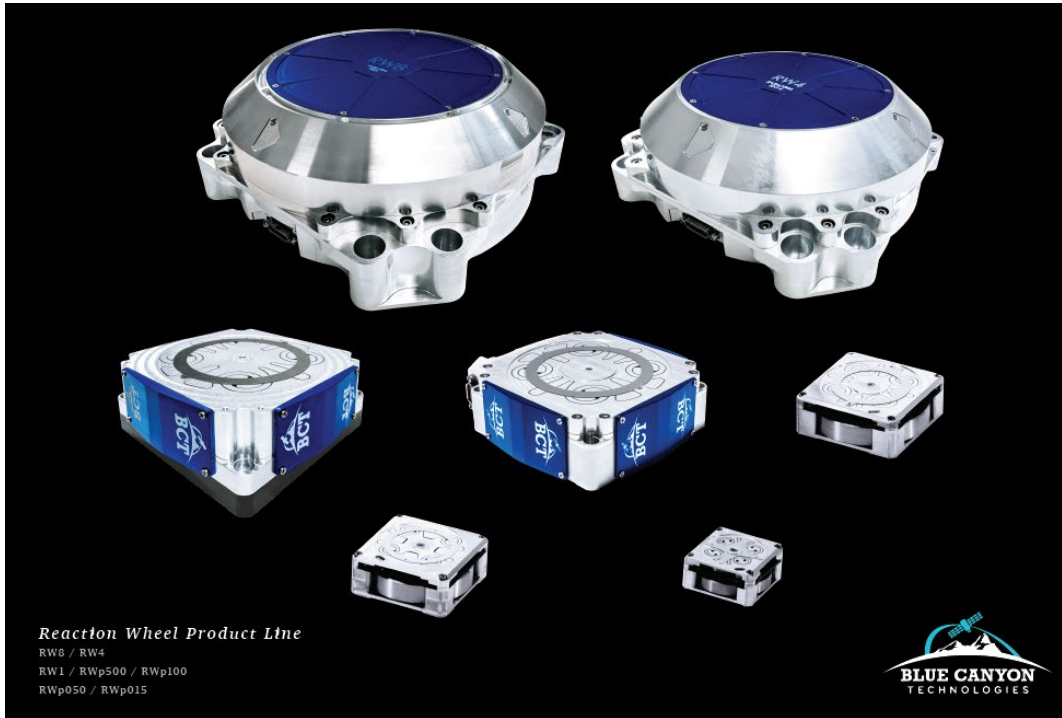
Vertical Integration Empowers Performance & Affordability

Spacecraft	Cubesats (3kg – 16kg)	Microsats (20kg – 350kg)
	 18 on orbit	 8 on orbit

Components	Reaction Wheels & Torque Rods	Star Trackers & Sun Sensors	Spacecraft Control Avionics	Power Systems & Solar Arrays	RF SDR
	 >400 RW on orbit	 >115 ST on orbit	 26 on orbit	 27 on orbit	 8 on orbit

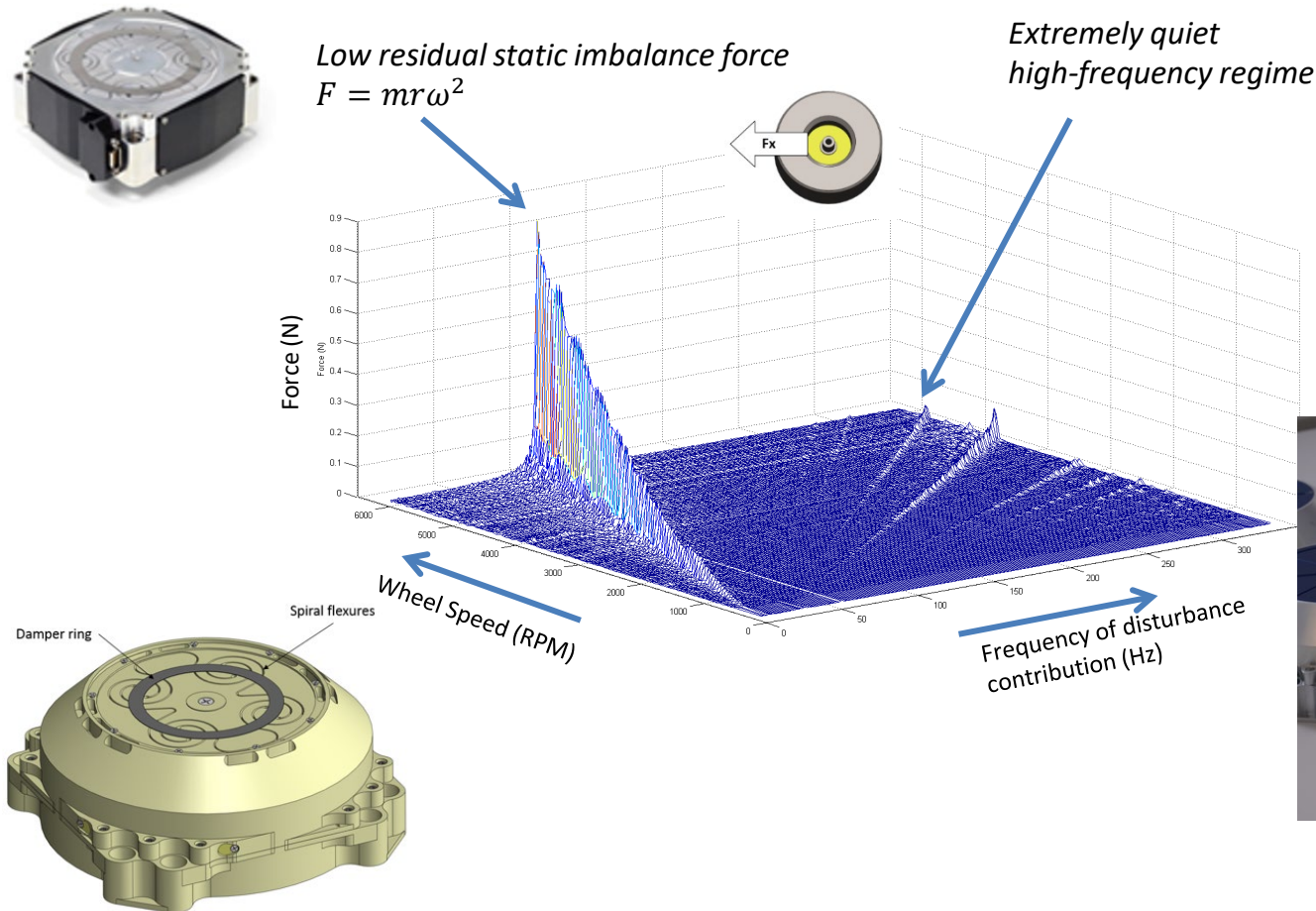
Reaction Wheel Sizes: 0.015 to 8.0 N-m-s

- High-volume manufacturing results in consistent quality and lower price
- Thousands produced
- > 400 on orbit
- Zero failures



Ultra-low Jitter Supports Precision Pointing

- BCT wheels are designed for long life, and extremely low jitter
- Most wheel sizes have been in life test for multiple years
- Low wheel disturbances result in low payload line-of-sight motion
- Plot is characteristic of all BCT wheel sizes



BCT Nano Star Trackers (NST)

- Reliable. High performance. Flight proven.
- Compatible across variety of spacecraft configurations and missions requiring high-precision knowledge
 - Tracks stars down to 7.5 magnitude
 - On-board star catalog (>24,000 stars) and lost-in-space star ID
 - Easy to integrate digital interface electronics
 - Compact packaging
 - >115 on orbit
 - Zero failures

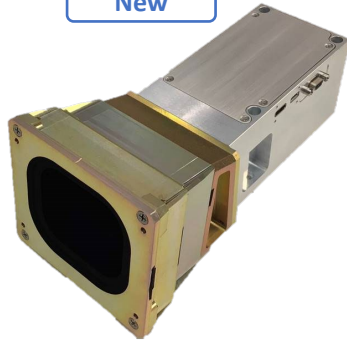
Standard

>50



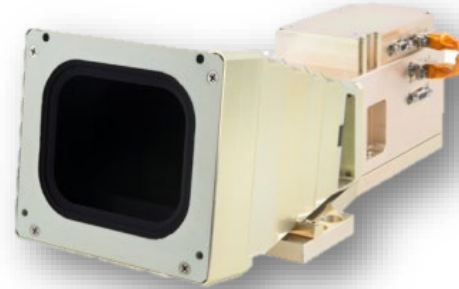
Mid Extension

New



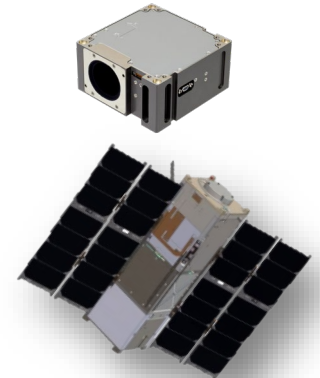
Extended

9



Integrated

>50

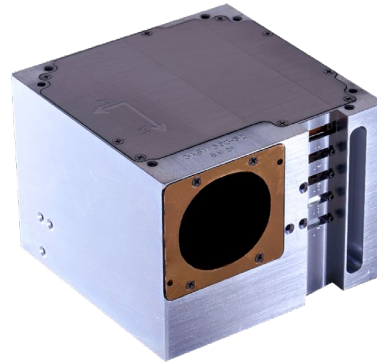


BCT Turn-key Attitude Control Systems

XACT-15



XACT-50



XACT-100

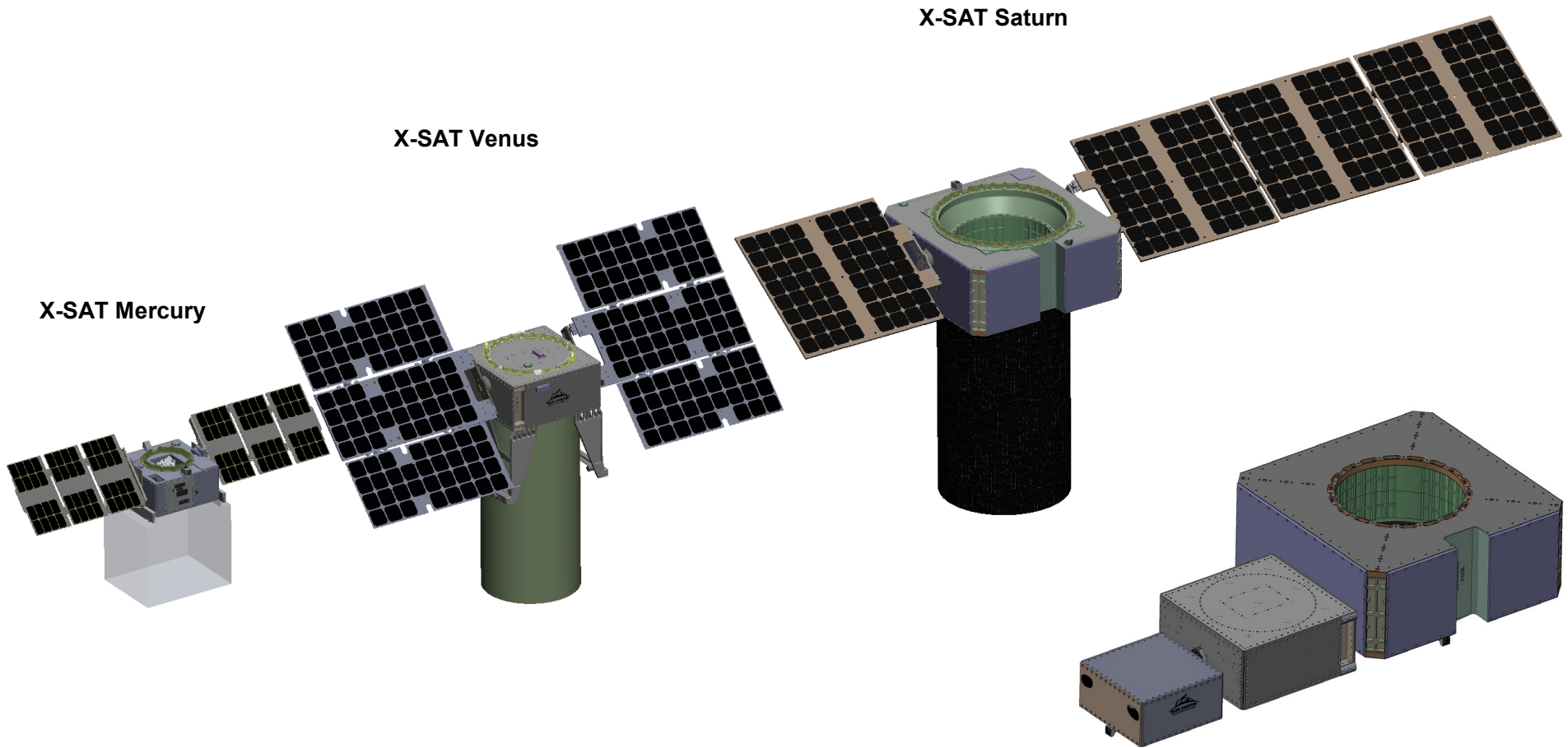


FleXcore

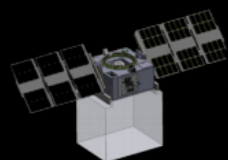


- XACT products support cubesats and smaller microsats
- FleXcore supports larger microsats
- >60 Attitude Control Systems flown
- Zero failures

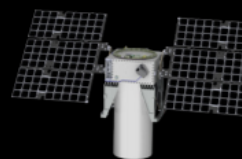
BCT Standardized MicroSat Product Line



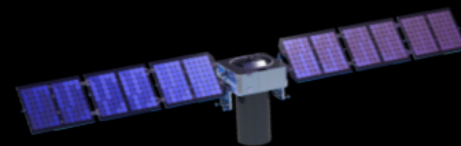
X-SAT Class Comparison



X-SAT
MERCURY CLASS



X-SAT
VENUS CLASS



X-SAT
SATURN CLASS

CLASS	8" launch vehicle interface (optional 11.32" available)	15" launch vehicle interface	24" launch vehicle interface
PAYLOAD VOLUME	14.0" X 17.0" X 17.0" (launch dependent)	20.5" X 16.4" X 27.0" (1 array) 17.0" X 16.4" X 27.0" (2 array) Larger volume available depending on launch vehicle	30.0" X 30.0" X 40.0" (typical) Larger volume available within rideshare envelope and in dedicated launch vehicle fairings
POINTING ACCURACY	±0.002° (1-sigma), 3 axes, 2 Trackers		
ENERGY STORAGE	10.2 Ah	10.2 Ah	40.8 Ah or 54.4 Ah
SOLAR ARRAY POWER	48W/wing, 98W max	THEA, two wing: 384W THEA, one wing: 192W	Hyperion 15, two wing: 1000W Hyperion 15, one wing: 500W
ORBIT ALTITUDE / ORBIT LIFETIME	LEO (≥ 5 years), GEO (≥ 2 years), Deep Space (≥ 2 years)		

'By the Numbers'

- >110 cumulative BCT spacecraft orders
- >50 different mission payloads

- >20 X-SAT spacecraft under contract (spanning all three classes)
- Many X-SATs going to GEO and beyond

- >30 BCT spacecraft and ADCS supporting missions in GEO, Lunar, and beyond
 - Includes XACT, cubesats, and X-SATs
 - ~10 different propulsion systems accommodated
 - Two BCT XACTs guided the MarCO cubesats to Mars
 - 10 of 13 Artemis-1 cubesats are utilizing BCT hardware

- Software features that support GEO, Lunar, and interplanetary missions are common to all BCT spacecraft and turn-key ADCS (cubesat to X-SAT)

MarCO: First Interplanetary CubeSats

MarCO XACT Systems



- Accompanied the Mars Insight Lander
 - Successfully relayed real-time EDL data to JPL (November 26, 2018)
 - No dropped data during the EDL portion (except expected Insight outage)
- BCT XACT attitude control system for each spacecraft
 - First XACT units delivered for flight from BCT
- BCT propulsion algorithms controlled thrusters for momentum control and trajectory correction burns
- Won the 2019 AIAA Small Satellite of the Year award

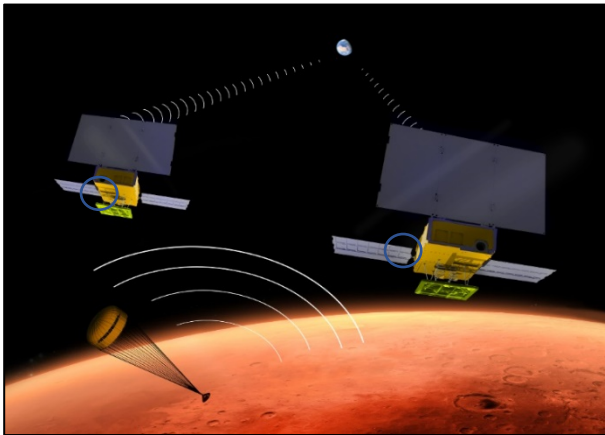


Image Credit: NASA/JPL - Caltech

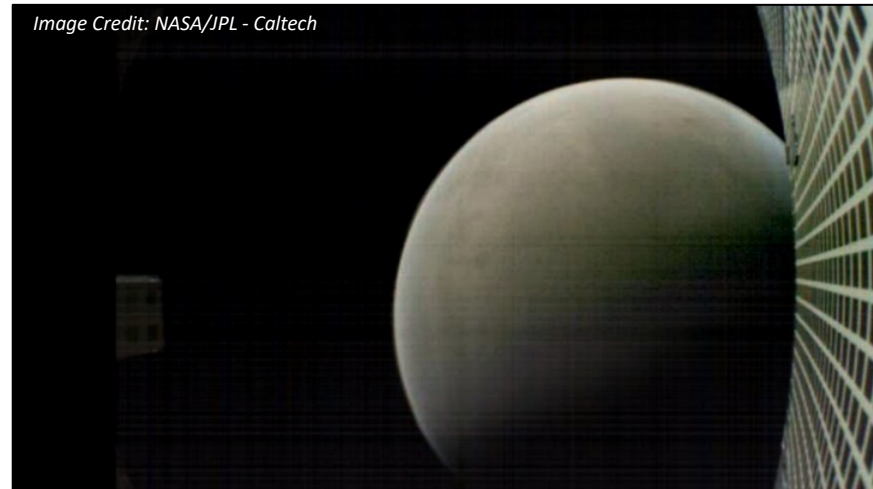
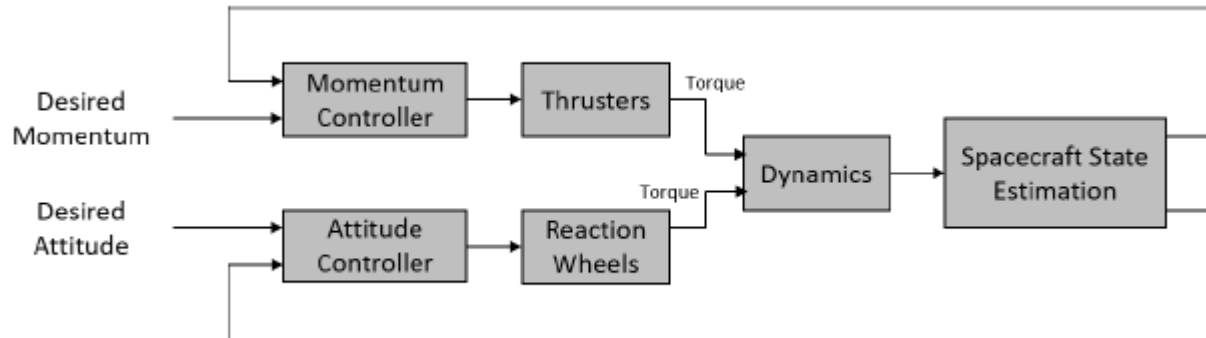


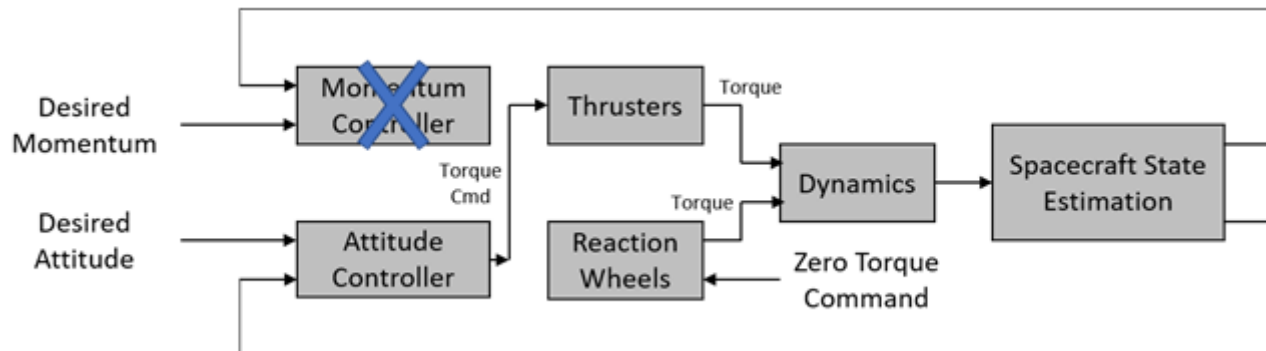
Image Credit: NASA/JPL - Caltech

Propulsion

- Propulsion is needed not just for orbit adjustment, but also for momentum management
- BCT spacecraft support hybrid thruster-wheel control for low-thrust systems



- And thruster-only control for high-thrust systems



Deep Space Trajectory Maintenance

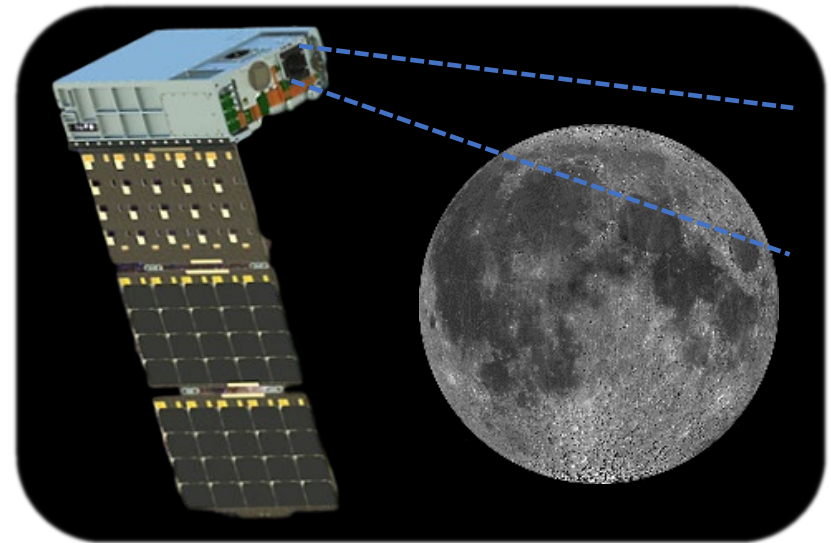
- LEO missions use a high-order onboard propagator, based on standard high-fidelity Earth gravitational models, along with high-fidelity models of position for sun, moon, Earth magnetic field, etc.
- Enhancements are necessary for lunar and deep space, including the ability to accept and evaluate standard polynomial trajectory definitions with respect to various celestial bodies
- Depending on mission need and complexity, options exist to provide many overlapping trajectory definitions to X-SAT, and to manage these trajectories via a sophisticated interface to add, delete, burn to nonvolatile memory
- The ability to store to nonvolatile memory is critical to mission autonomy

Autonomy

- Lunar and interplanetary missions often require increased autonomy, compared to typical LEO missions
 - Communication with spacecraft is often very limited and delayed
 - Certain propulsion burns must occur at the desired time to inject properly
- X-SAT supports sophisticated table-driven Event-Checks and macro command responses
- Key features:
 - Momentum fault detection and autonomous thruster-based momentum unloading
 - Autonomous ADCS commissioning: from tipoff to 3-axis stabilized operations in safe mode
 - High-fidelity celestial body modeling that eliminates the need for ground-based pointing attitude computation
 - Critical burn recovery: ability to autonomously complete mission-critical thruster burns, utilizing critical data in non-volatile memory and a real-time clock

Future Lunar/Deep-space Autonomous OD Capability

- NASA Tipping-Point Technology contract to develop and demonstrate lunar/deep-space autonomous orbit determination
- Zero SWaP software solution
- Utilizes BCT star tracker imaging, high-accuracy attitude, and advanced filtering to determine orbit
- Will help off-load DSN utilization



THANK YOU

