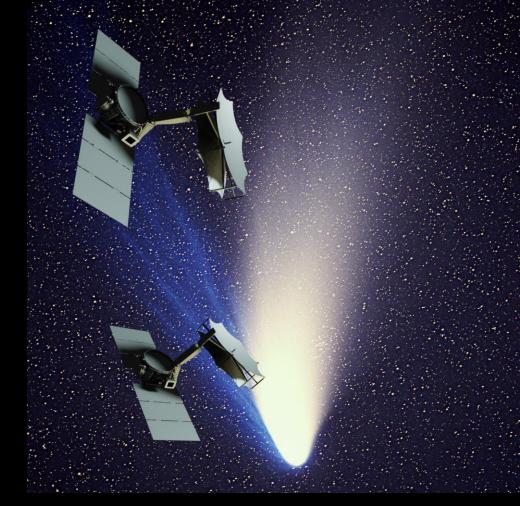
Xenia

SmallSat Rapid Response Flyby to Oort Cloud Comets and Interstellar Objects

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Authors

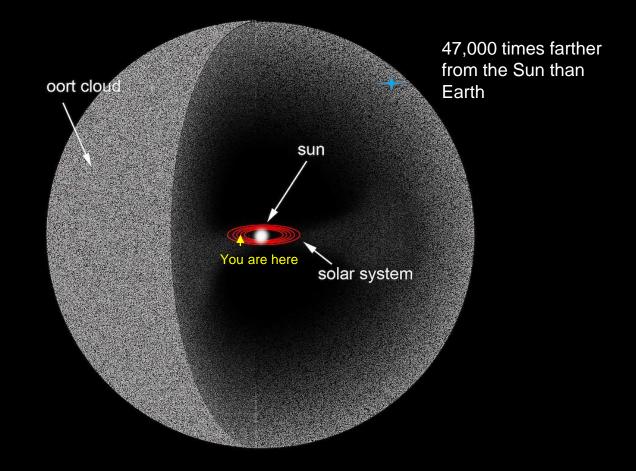
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Note on Terminology

- Oort Cloud Comet (OCC) Comet that originated in the Oort Cloud, a region of the solar system ~2,000-200,000 AU characterized by the abundance of icy small bodies
- Long Period Comet (LPC) Comet that originated in the Oort Cloud and eventually developed a shorter period through interaction with the giant planets. An OCC is a "fresh" LPC.
- Interstellar Object (ISO) A small body that originated from outside our solar system, i.e. on a hyperbolic solar trajectory. Can be actively outgassing (like 2I/Borisov) or asteroid-like (like 1I/Oumuamua)



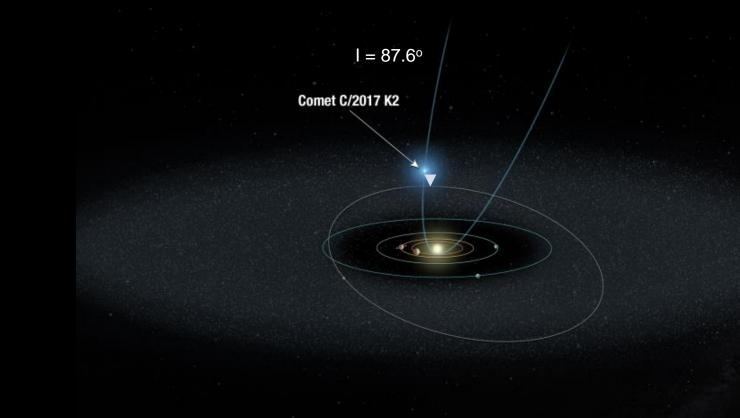
Why do we Want to Explore OCCs?

- Bound the temperature of the protoplanetary disk from volatile isotope markers and noble gasses
- Determine the makeup of the early solar system including volatiles, organics, and refractory material
- Understand the nature of the diversity among comets

Example Comet: C/2017 K2 – A Gift from the Sky

- C/2017 K2 (C2K2) is a "great" comet with nucleus between 50-80 km
- Perihelion in Dec. 2022 at 2.1 AU from Earth, 2.3 AU from Sun
- C2K2 will visit once in our lifetime but there will be more comets like C2K2
 - No guarantee those would be accessible
 - C/2010 U3, another record comet had perihelion at 8 AU (Feb 2019)
- All comets are interesting!
 - Concept discussed today can be applied to many future targets





Qualifies as "Great" comet (50-80 km large) ~1-3 million year orbit (inbound)

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Future Visitor Exploration



Borisov, 2019 Image credit: Jewitt, 2019 115,000 km

C2K2, 2017 Image credit: Jewitt, 2017





Hale-Bopp, 1995 Image credit: Kolmhofer, 1997

Ouamouamoua, 2017

Image credit: Meech, 2017

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Xenia Mission Concept

- Scientists and engineers at JPL worked to develop the Xenia mission concept
 - Rapidly build, test, and launch mission to encounter C2K2
 - <18 months from project start to ATLO
- Conclusion: technically, development is feasible but requires new technology development and parts availability
 - MarCO demonstrated rapid development process
 - Availability of COTS parts streamlines the design and build process
 - Enabling technology is actively in development, if not yet flown
- Programmatically, there was no place for a rapid response mission
 - Too large for SIMPLEx and cadence is not conducive to reactivity

Science

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- Constrain the temperature of the edge of the protoplanetary disk

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 - First "reactive" mission, with many applications down the line
 - Scout is important for assessing risk (~38 km/s, 10⁵ kg/s dust ejected)

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 - Public will see the comet with their naked eye from Earth and up close from Xenia



Credit: Cherie Benoit at Flickr



Credit: Wally Pacholka



Credit: Paul Sutherland



Credit: Philipp Salzgeber

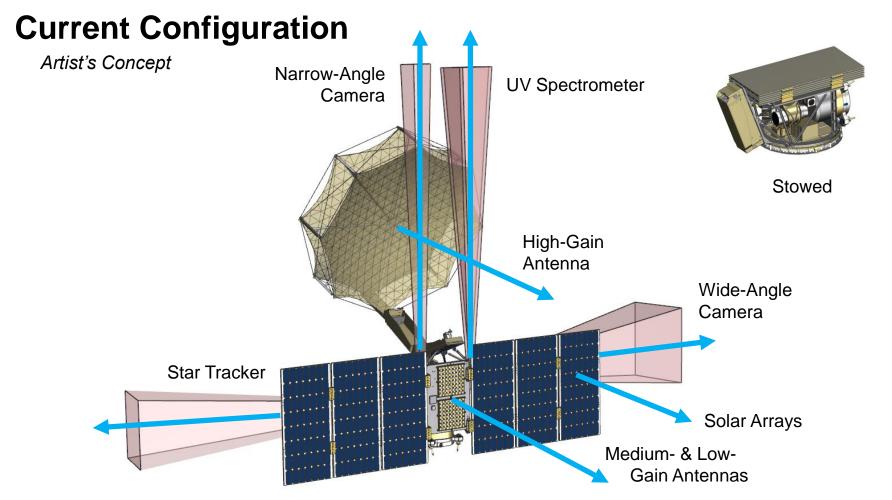
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- Technology demonstration opportunity
 - Autonomous science data return
 - Deep-space SmallSat propulsion

Xenia mission goals are enabled by using small spacecraft

Driving Requirements

Short Name	Requirement	Drives
Mass	The mass of each spacecraft shall not exceed 75 kg (150 kg total)	Flight system and payload
Launch Energy	The launch vehicle shall directly inject the two spacecraft on an intercept trajectory with an energy of >57 km ² /s ²	Flight system and navigation
Launch Date	The spacecraft shall launch no later than February 2022	Schedule (all)
Post-Encounter Data Return	The spacecraft shall return at least 1 Gb of data from the encounter before end of mission	Flight and ground systems

SmallSat



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Paradigm Shift in SmallSat Exploration | Constraints

- Extremely rapid turn-around time (~24-month project from start to launch)
 - Need all subsystems to be ready for integration
- Large amounts of launch energy required to encounter (C₃ of >57 km²/s²)
 - These types of opportunities almost never land in launch energy-friendly places
- High-risk, high-reward exploration to time-limited visiting body
 - Schedule slip means missing the opportunity!

Enabling Capabilities for Rapid SmallSat Exploration

- Efficient SmallSat deep space propulsion (Isp > 200)
 - Several companies and institutions who are trying, but no existing, flight proven systems
- Flight-proven, stowable HGAs and autonomous data return
 - Need capability to return science data after fast fly-by
- Rapid turnout for COTS subsystems and components capable of use in deep space
 - E.g. SDL IRIS radio
- No programmatic opportunity for rapid response missions
 - Existing AO scheme does not allow for missions to take advantage of fleeting opportunities like visiting a passing by comet

Summary (1)

- OCCs and ISOs hold some of the most pressing secrets about the formation of our solar system but are not easy to get to
- Missions of opportunity are often characterized by high energy and inclination orbits and quick time to respond
- Need the capability to respond to incredible science opportunities
- We are eager for our first flyby of an OCC or ISO!

Summary (2)

- Need to leverage commercial parts and rapid development cycles to respond to infrequent, fleeting opportunities with great science potential
- Reactive mission to OCC is technically feasible; design can be adapted for other far-away visitors like OCCs and ISOs
- Programmatic elements are not in place to quickly mobilize to respond to an opportunity like a passing OCC or ISO

We need to get ready to respond to an approaching OCC or ISO!



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