Cubesat Constellation Architecture to Support Space-Based Property Claims

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Challenges

- Property Ownership and Usage Rights in Space (Int'l Law and Military/Defense)
- Historical Norms

 Societal and Economic
- Registration (e.g., UN Registry for Objects in Space)

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Missions Involving Secondary CubeSat Deployments





Objectives

- **Deployment** of 'Beacons' to Small-Body (Asteroid) Orbit (Orbital Insertion) with Sensory Instrumentation
 - Geolocationing (Precision) and In-Space Activity Capture (Timestamped Records with Accurate Location Matched with Intelligent Classification Software)
- Integration with Service Providers
- Collection Comprises a 'Universal Cadastre'
- *Cooperation* with Int'l & Domestic Regulatory Agencies



Architecture





Key Planning Steps and Events Sequence

I. Small-body, destination selection;II. Launch window selection;







Small-Body Selection

4660 Nereus (1982 DB)					
Δv (km/s)	4.985				
Taxonomy	C, X, E				
Spin Period (hours)	15.16				
Synodic Period (years)	2.225				
Orbit Condition Code	0				







Launch Windows for 4660 Nereus (1982 DB)





Key Planning Steps and Events Sequence (continued)

- I. Selection of small-body, probe destination;
- II. Select launch window;
- III. Launch to GEO;
- IV. Payload deployment;
 - V. Cubesat travels toward rendezvous with small-body;
- VI. Cubesat positioning in orbit around small-body;
- VII. Data acquisition;
- VIII. Relays to larger satellite receiver; and
 - IX. Continuous data acquisition, encoding, and relays.



Asteroid's Trajectory Itinerary



4660 Nereus	(1982 DB)		
SPK-ID	2004660	Orbit Condition Code	0
Absolute Magnitude	18.2	Size	330 m
Semi-major axis	1.486 AU	Eccentricity	0.359
Inclination	1.45°		

Trajectory Itinerary

	Date	ΔV			
Earth Departure	e Jan-13-2022	4.07 km/s		C3 = 19.3 km ² /s ² DLA = 6°	
1.31-yr transf	er				
Asteroid Arrival	May-08-2023	586 m/s			
1.31-yr total mission		587 m/s 4.66 km/s	post-injection ΔV total ΔV		
Solar range:	0.98 - 1.98 AU	Earth ra	nge:	0 - 2.94 AU	



Full Timeline

2020	2021	2022	2	023	2024	2025	2026	2027
	Design + Build	Outbound Cr	ruise	Asteroid	Operations			
		Í	Î	Î				_
	 Launch: Jan 2022 Asteroid acquisition Maneuvers to reach 	and approach maneu asteroid orbit	ivers					

- In-orbit observations and comms
- In-orbit servicing (*ongoing, as needed*)



Hardware

- Dispenser, Housing for Cubesat Constellation
- Locker
 - Cubesat Bus
 - Single-Board Computer and Memory
 - Thermal Radiator and Radiation Shielding
 - Solar Panels
 - Electric Propulsion System
 - Iris V2 CubeSat Deep-Space Transponder (IRIS), Omni-Directional UHF Antenna, and High Gain Reflectarray Antenna
 - Van Atta Reflectors
 - Sensors
 - CMOS Cameras
 - Laser Altimeter and Star Tracker







Mission Software

- Off-the-shelf:
 - Dispenser Timing/Precision Deployment
 - Power Management
 - Avionics, Attitude Control, and Propulsion
 - Station-Keeping
 - Relay Encoding and Timing
- In-house:
 - Data Capture, Handling, and Logging/Storage
 - Activity Classifier
 - Secure Encryption and Storage Distribution



Comms

- Secure Transmissions Over Long Distances
- Sacrifice Timeliness for Precision, Reliability
- Interoperable with Larger Satellite Communications Infrastructure
- Data Storage Distribution: Portion Kept in Space, Portion Sent to Earth-Based Servers



Regulatory Concurrence

- *Main Objective*: Legitimize In-Space Property Ownership & Usage Rights
 - Precise Location & Activity Data, Validating Legitimacy of Third-Party Claims
 - Service provider for State and Non-State actors
 - Digitally Connected with Appropriations Agencies at the International and Domestic Levels





Further Investigation

 Optimization algorithm for *constellation-scale* deployment (including lunar and Mars gravity assists and efficient rendezvous with multiple small-bodies)

- Comms hardware upgrades (e.g., IR)
- Sensor instrumentation upgrades
 (e.g., X-ray fluorescence imaging spectroscopy [see: <u>CubeX</u>, 2018])
- Modularity (pre-built extensibility for add-on modules)



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