

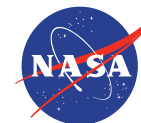
# **A Venus SmallSat Orbiter for Remote Sensing of Seismic Activity, the VAMOS Concept**

**Alan M. Didion, Systems Engineer**  
NASA Jet Propulsion Laboratory,  
Engineering and Science Directorate,  
Systems Engineering & Formulation  
Division

ISSC 2018, Pasadena, CA

As presented by:

**Alex Austin, Systems Engineer**  
NASA Jet Propulsion Laboratory,  
Engineering and Science Directorate,  
Systems Engineering & Formulation  
Division



# Introduction to the Planetary Science Deep Space SmallSats (PSDS3) Program

- ROSES 2016
  - Explore mission concepts for SmallSats (< 180 kg) at planetary targets, identify critical technology gaps
- Proposed concepts:
  - VAMOS
  - Cupid's Arrow
  - CUVE
  - CubeX
  - BOLAS
  - APEX
  - CAESAR
  - Chariot to the Moons of Mars
  - Aeolus
  - SNAP
  - JUMPER
  - ...and others!

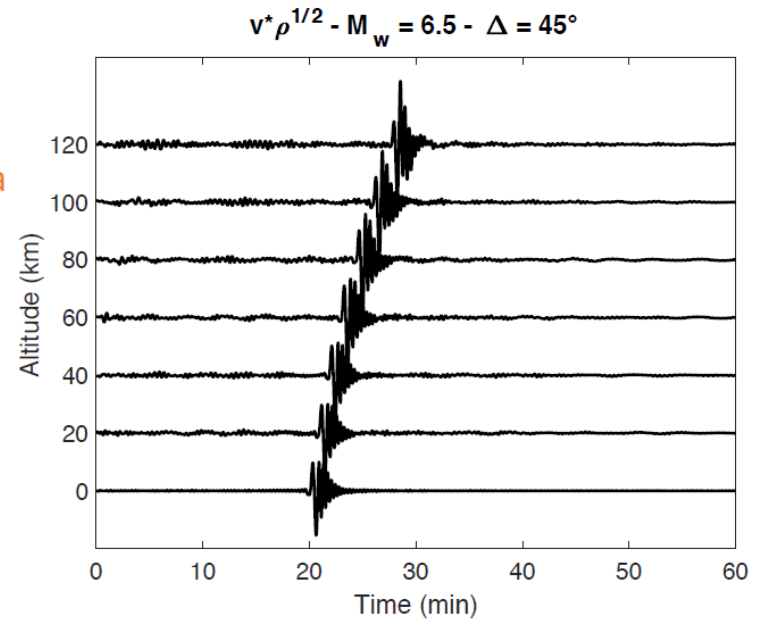
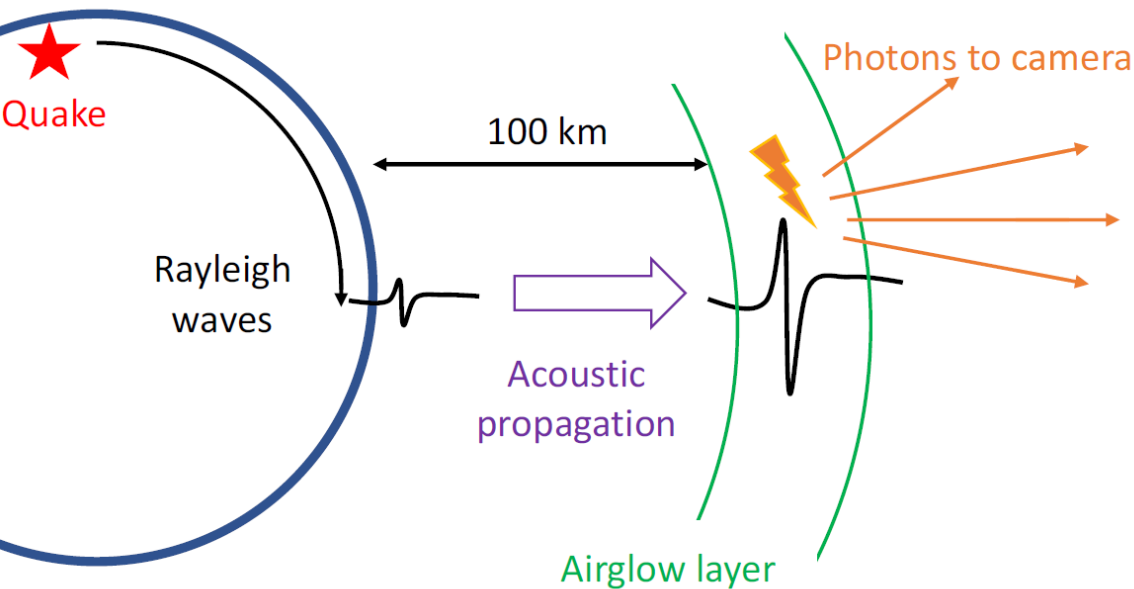
# Planetary Quakes on Rocky Bodies

Deep Insight via Global Observations



# Planetary Quakes on Rocky Bodies

Deep Insight via Global Observations

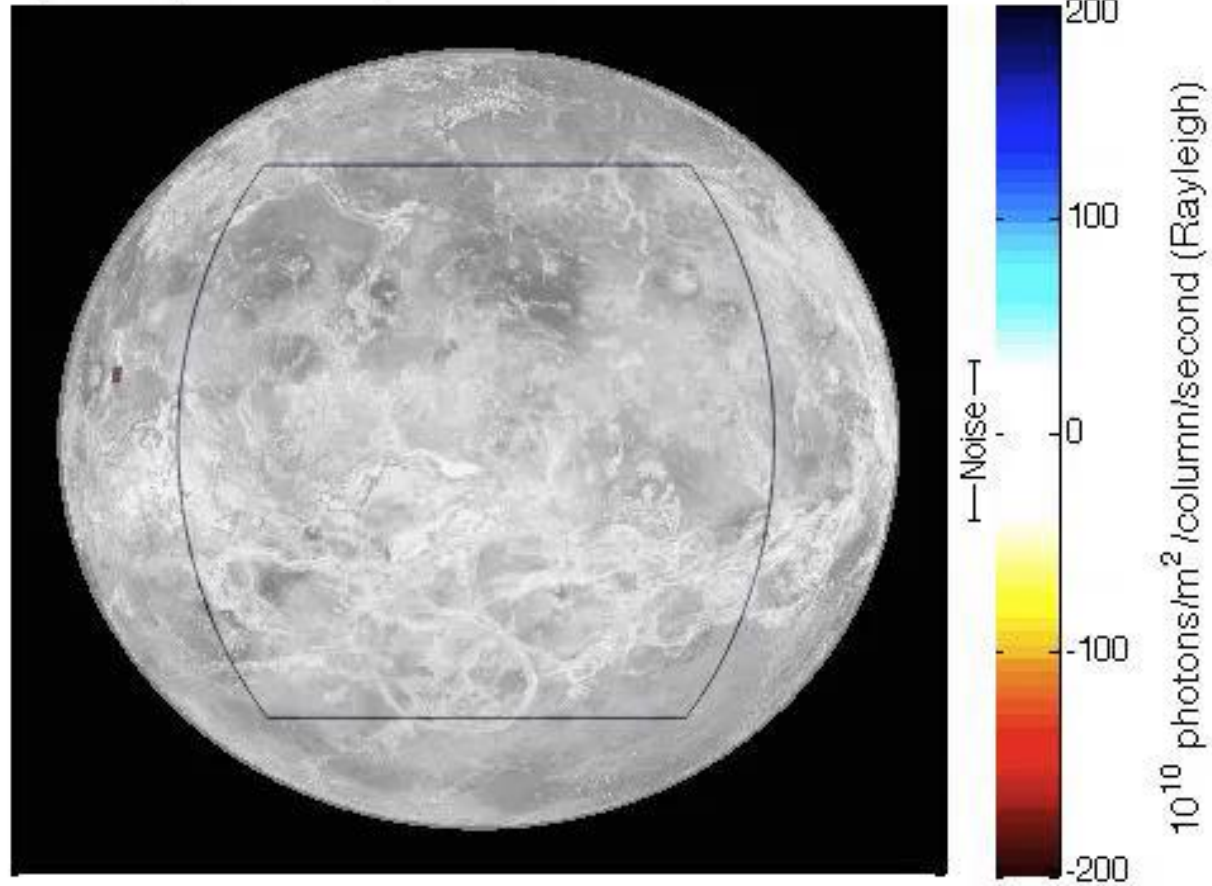


**Fig. 5** Synthetic seismograms at different altitudes in the atmosphere; the product of vertical velocity by the square root of density is shown.

# Atmospheric Airglow

## Ionospheric Modulation Reveals Surface Displacement

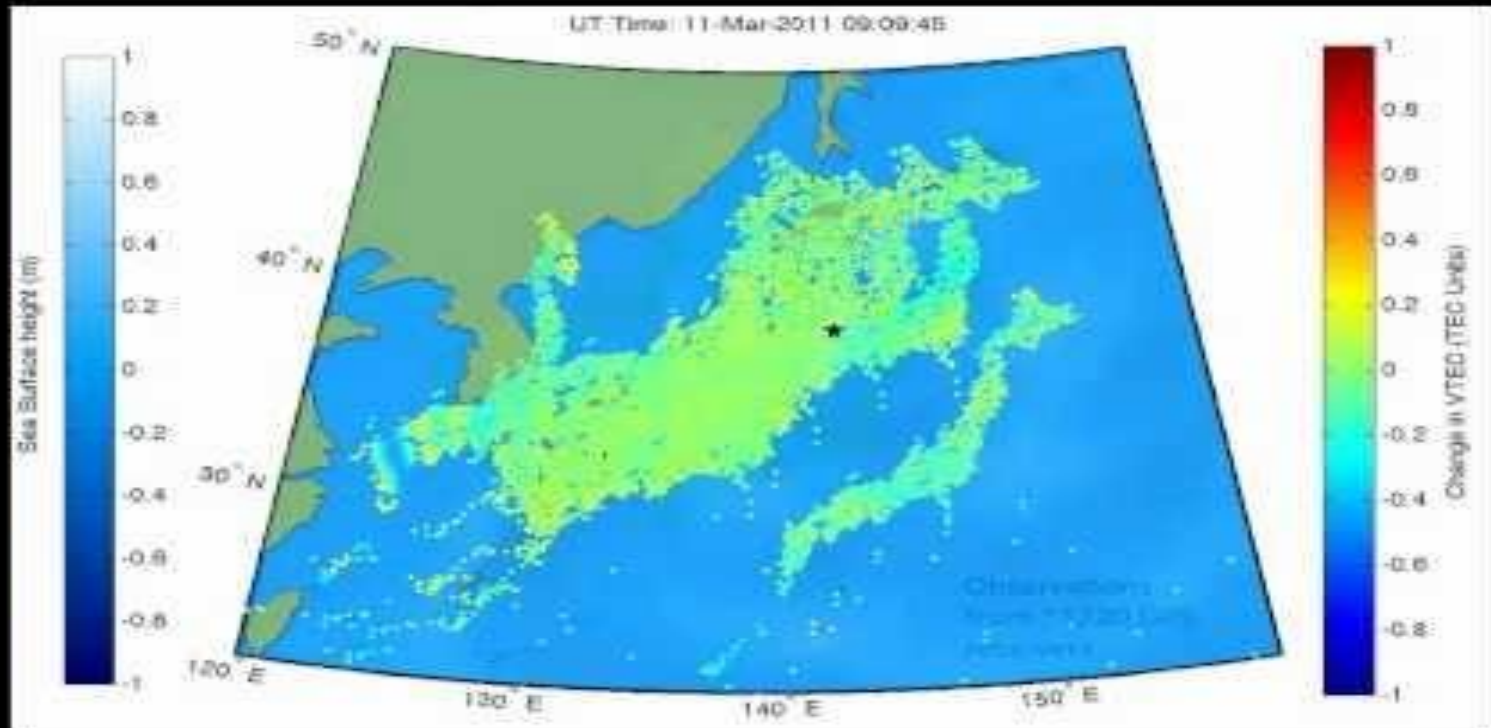
1.27  $\mu\text{m}$  airglow intensity fluctuation, Time: 0 min 10 s





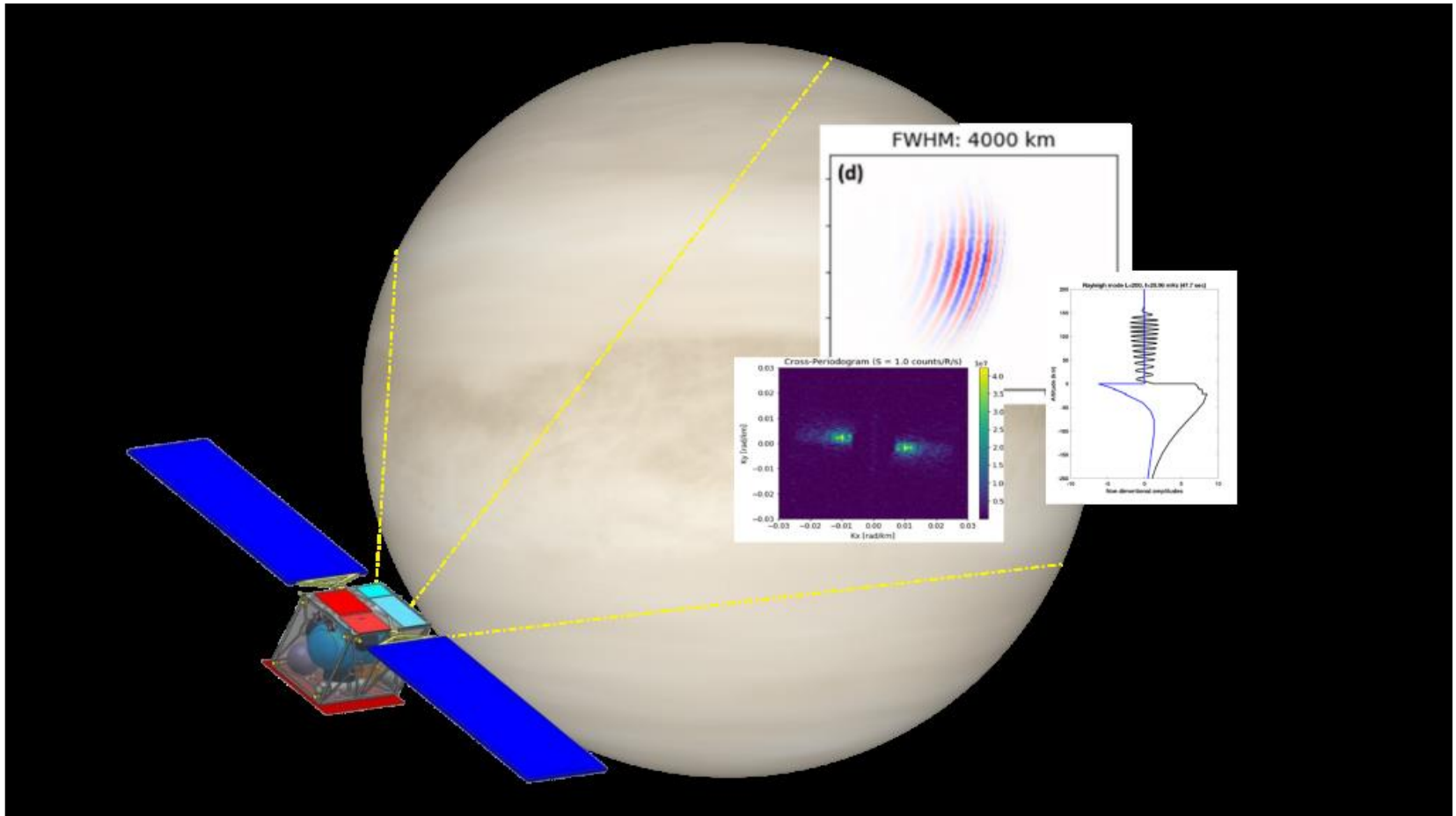
# Atmospheric Excitation by Planetary Quakes

An Earth Analog- Japan Earthquake & Tsunami: March 11, 2011



# Mission Concept Overview

A Vigilant Small Spacecraft in High Circular Venusian Orbit



# Instrument & Spacecraft Description

A Simple Infrared Telescope with Dual Detectors on a SEP SmallSat

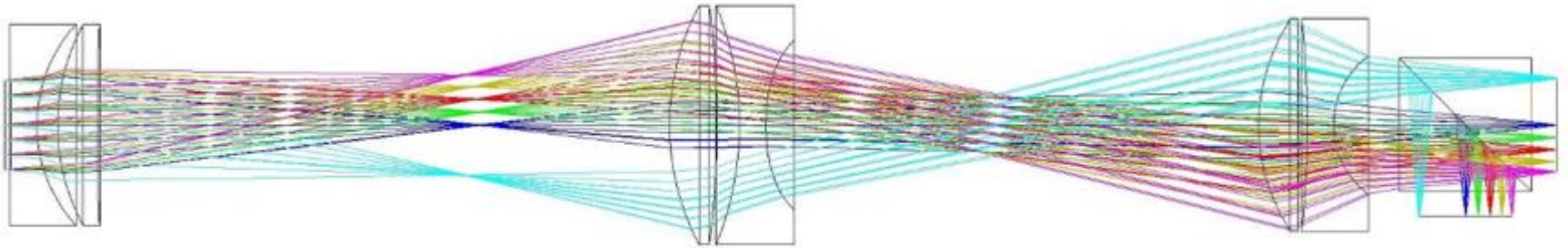
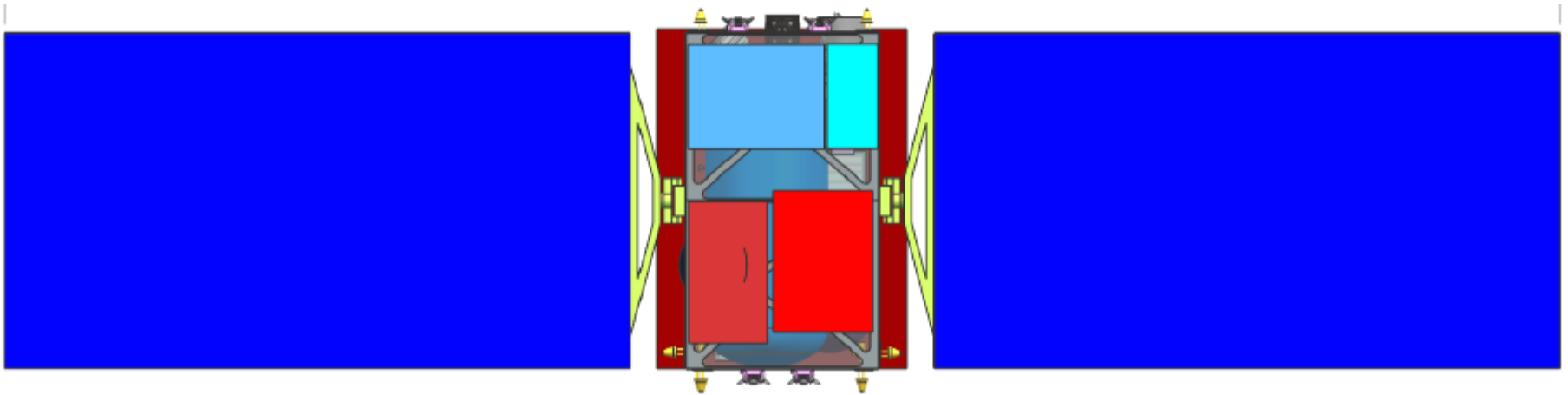


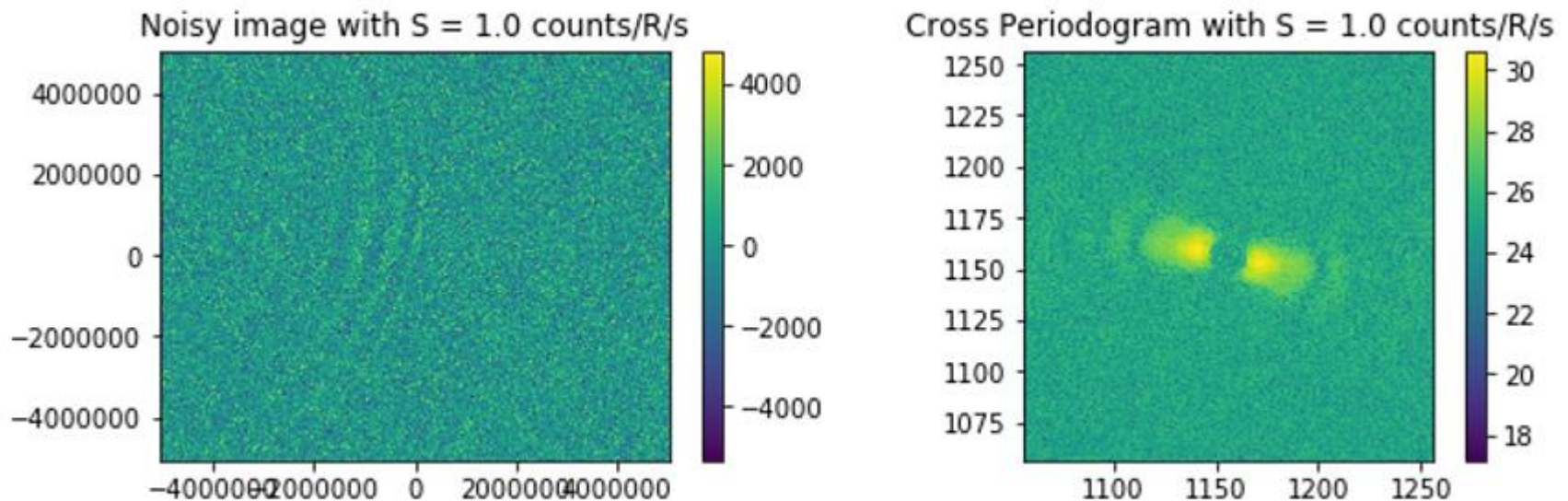
Figure 13- Baseline concept refractive design (aperture to the left, detectors behind a beamsplitter to the right)





# Event Detection Algorithms

## Real-time Wavefront Detection

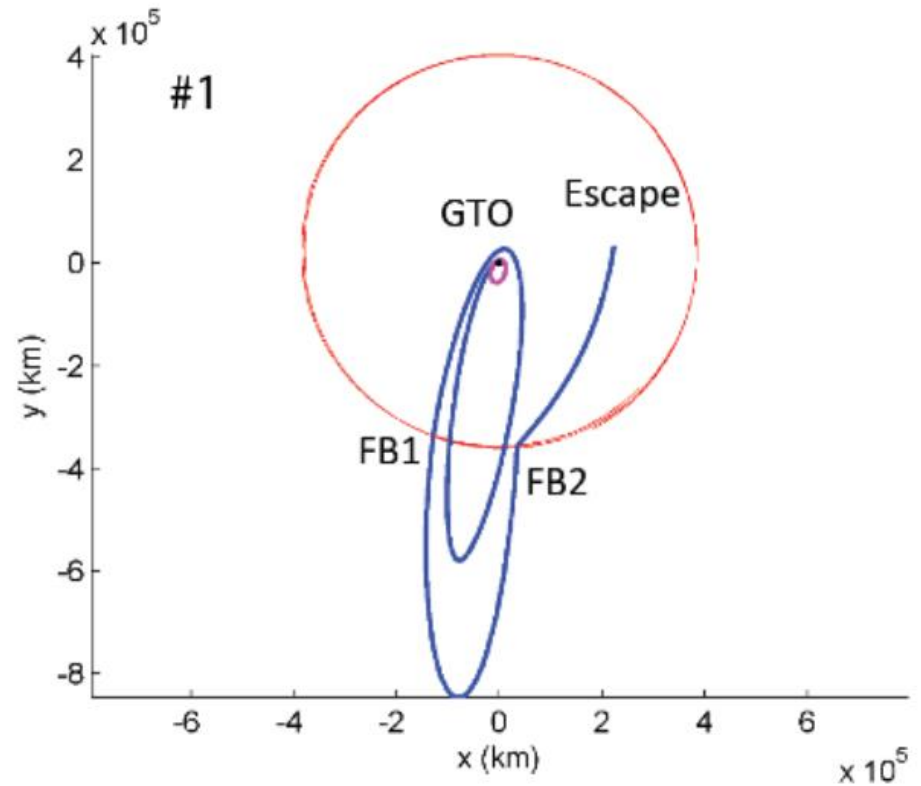
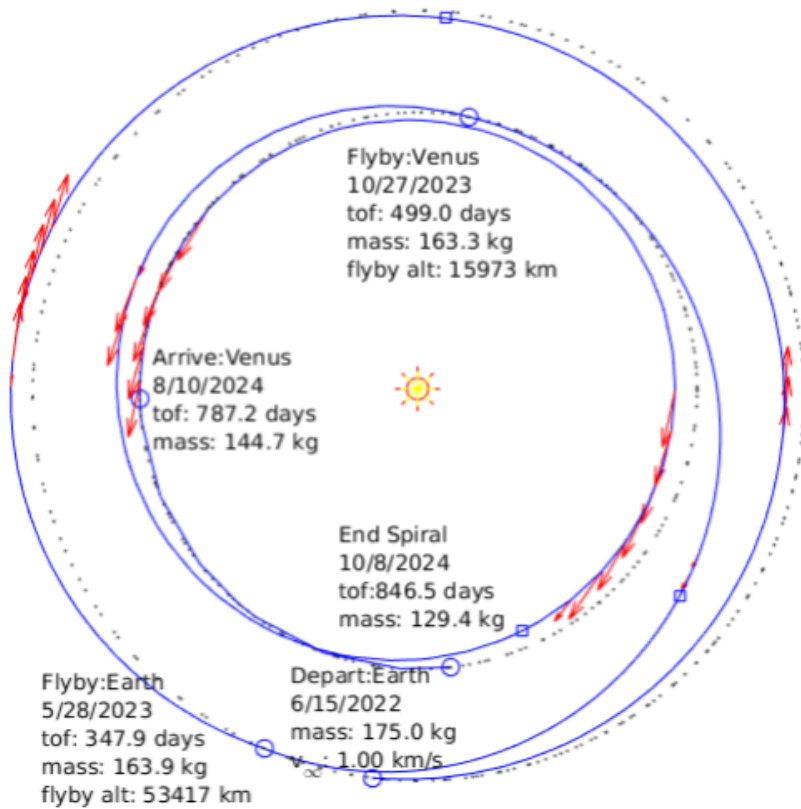


**Figure 2- Simulated image of raw data (left); periodogram showing detection (right)**

# Trajectory Design

## Solar Electric Propulsion- EEVV Transfer

Reference Trajectory



# Trajectory Design

## Rideshare Flexibility- GTO Escape from the Earth-Moon System

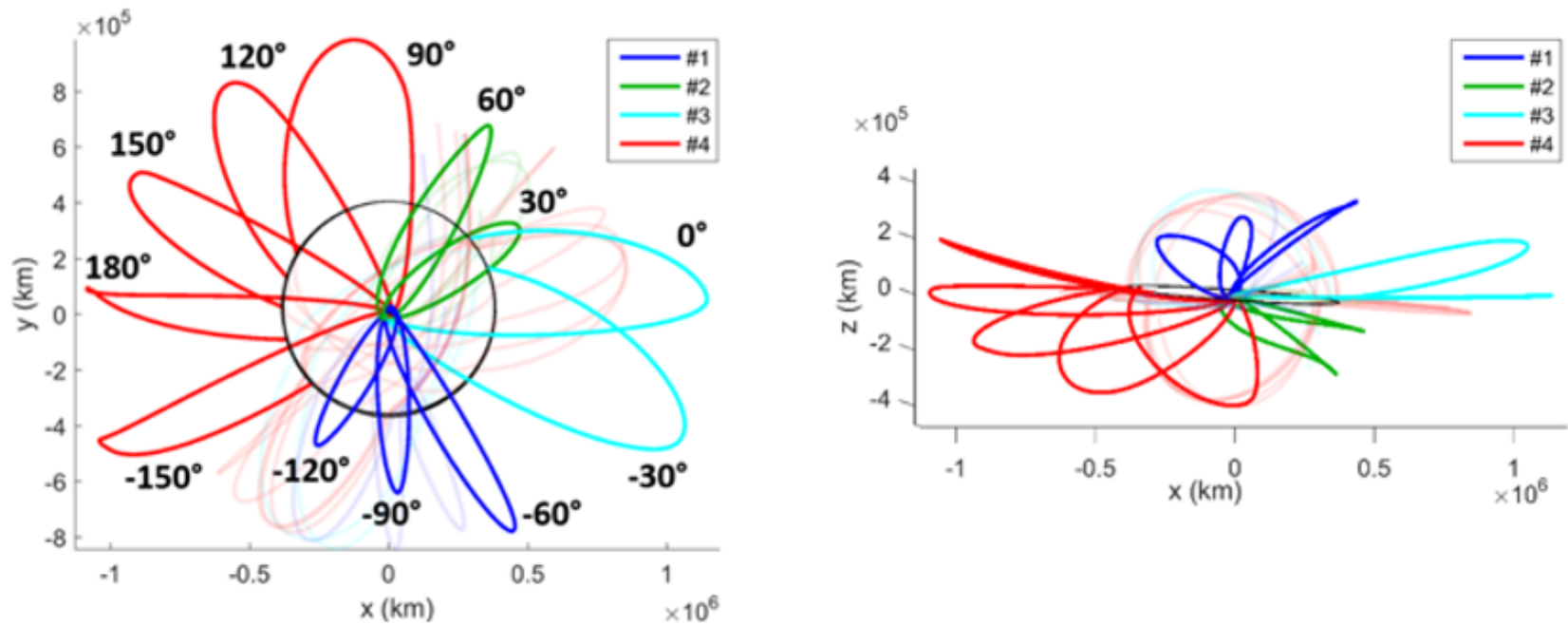
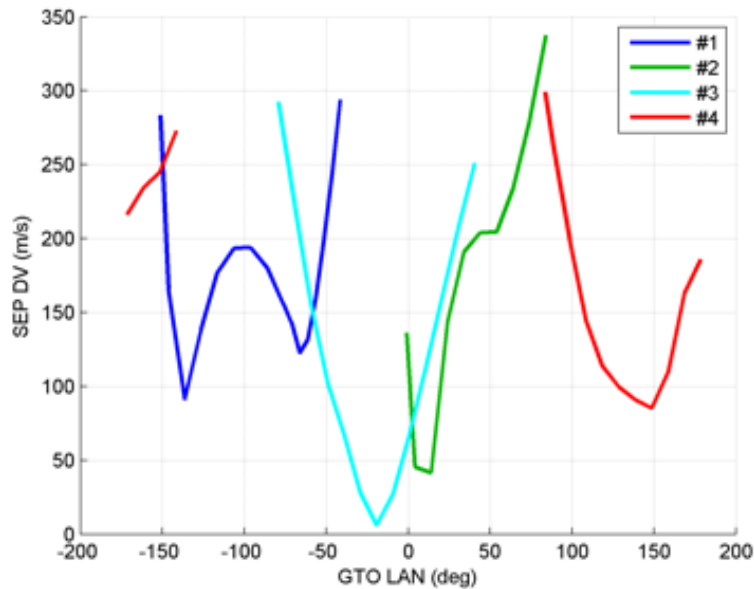


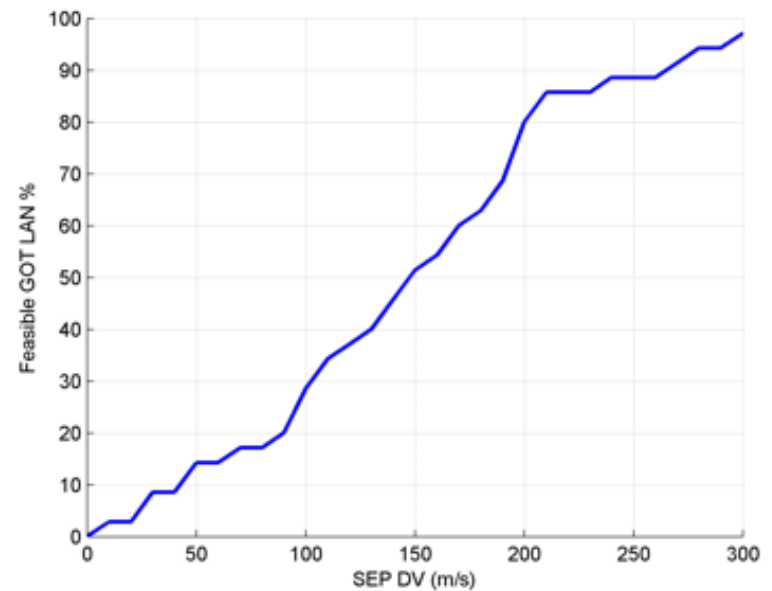
Figure 8- Trajectory solutions for 360 degree range initial GTO LAN in 30-degree increments. Left: top view; right: side view. For easier visualization, all trajectories are transparent after the first lunar flyby.

# Trajectory Design

## Rideshare Flexibility- GTO Escape from the Earth-Moon System



**Figure 9- SEP  $\Delta V$  required for initial GTO LAN values ranging from -180 to 180 degrees**

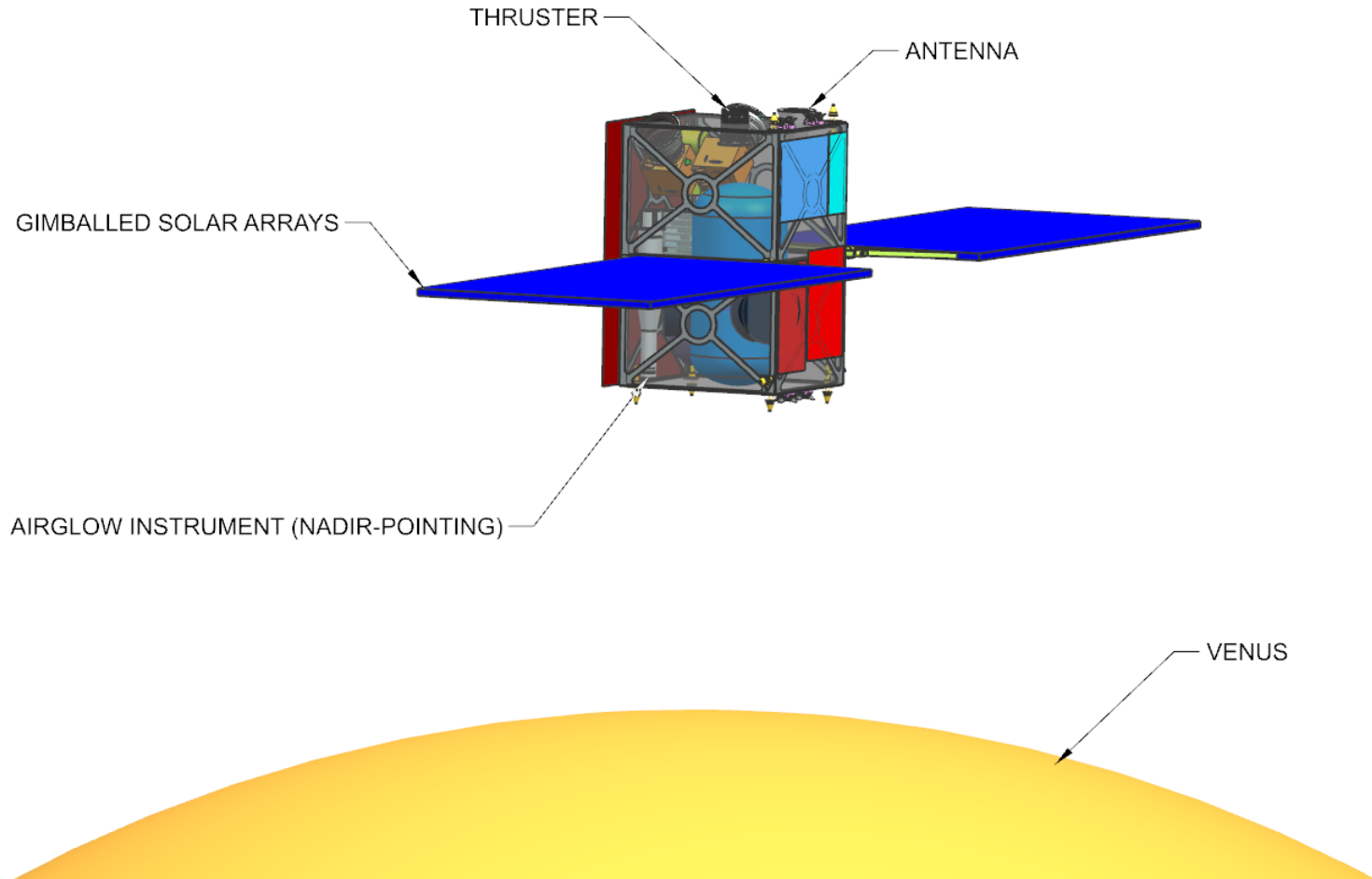


**Figure 10- Percentage of allowable GTO LAN as a function of SEP  $\Delta V$**

# Concept of Operations

## Constant Observation of the Venusian Disc

Not to scale!



# Option Comparison & Major Trades

- Chemical propulsion option for lower cost at the expense of suboptimal science orbit
- Advanced CubeSat/SmallSat CDHS vs. RAD750-based systems
- Possible commercial bus partnerships
- Direct-delivery via Venus-bound mission
  - Simplifies propulsion, power design, but drastically limits launch flexibility
  - Unique VAMOS orbit further unlikely to be offered



# Meet the Authors

Many thanks to them!

- NASA JPL
  - Alan Didion, Attila Komjathy, Brian Sutin, Barry Nakazono, Ashley Karp, Mark Wallace, Gregory Lantoine, Siddharth Krishnamoorthy, Mayer Rud, James Cutts
- IPGP
  - Philippe Lognonné, Balthasar Kenda, Mélanie Drilleau
- University of Illinois at Urbana-Champaign
  - Jonathan Makela, Matthew Grawe
- DLR
  - Jörn Helbert
  
- A. Didion, A. Komjathy, B. Sutin, et al., (2018) “Remote Sensing of Venusian Seismic Activity with a Small Spacecraft, the VAMOS Mission Concept”, IEEE Aerospace Conference, Big Sky, MT, March 3-10.

# Q&A, Acknowledgements

- Contact Information:
  - Alan Didion: presenter, systems engineer
    - Alan.M.Didion@jpl.nasa.gov
  - Attila Komjathy: principal investigator
    - Attila.Komjathy@jpl.nasa.gov
- This material is based upon work supported by the National Aeronautics and Space Administration under ROSES 2016 NNH16ZDA001N-PSDS3 issued through the Planetary Science Deep Space SmallSat Studies Program. Support to the French team has been provided by CNES. This work was conducted at the NASA Jet Propulsion Laboratory, a division of California Institute of Technology.