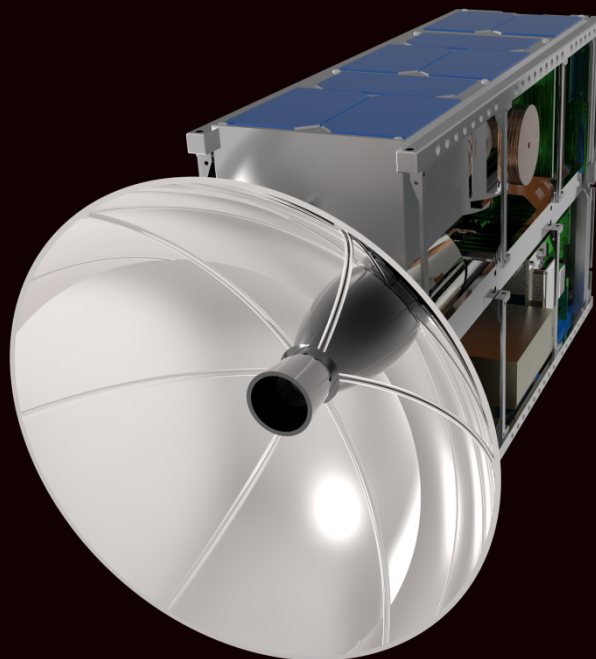


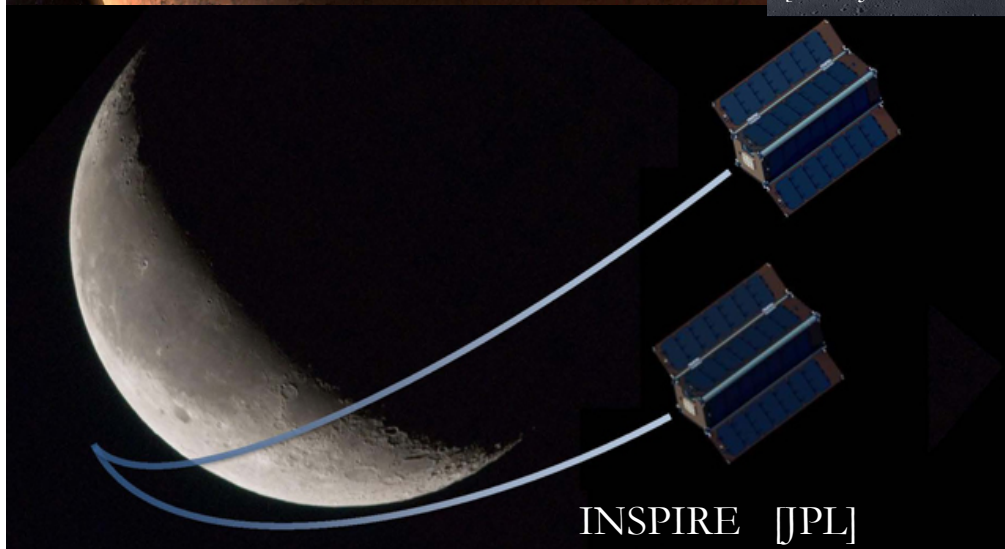
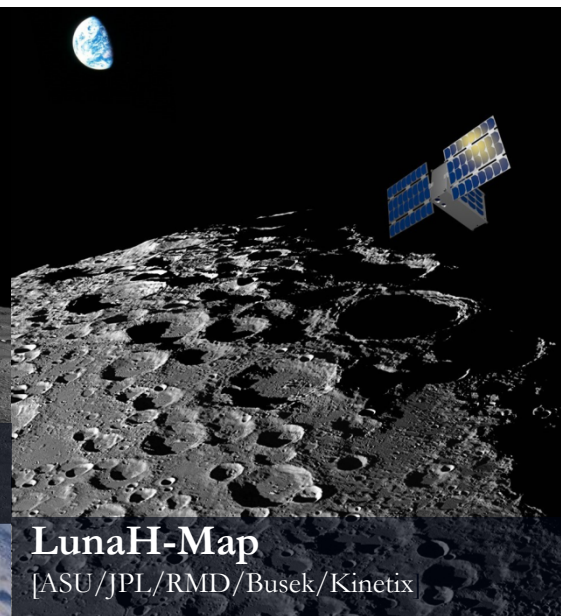
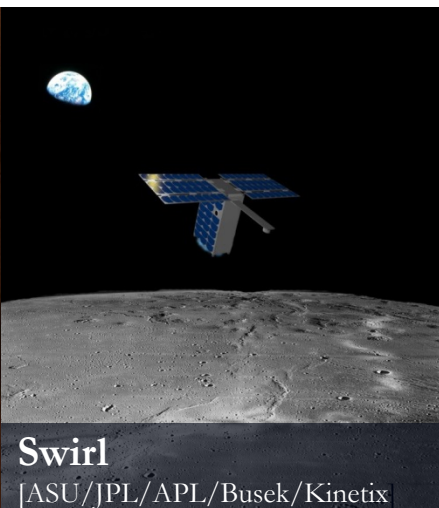
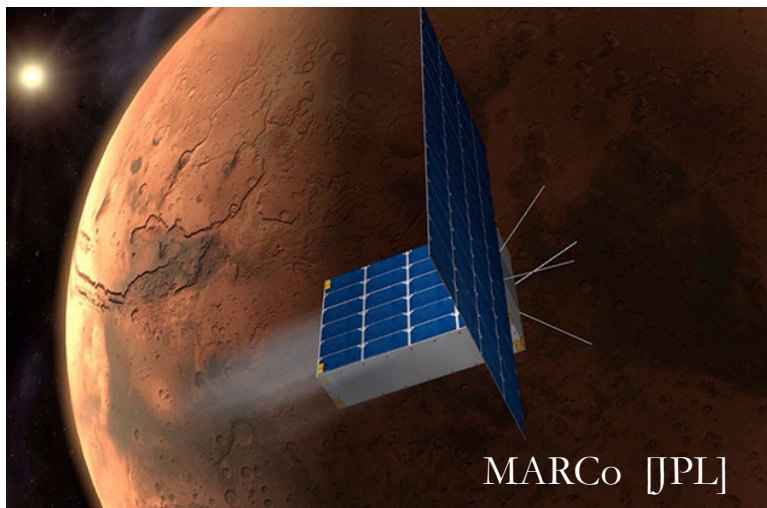
SpaceTReX

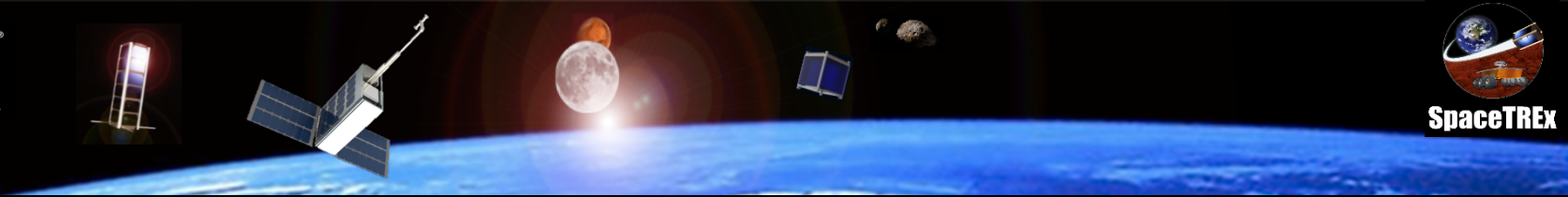


Solar Thermal Propulsion for Interplanetary Small Satellites

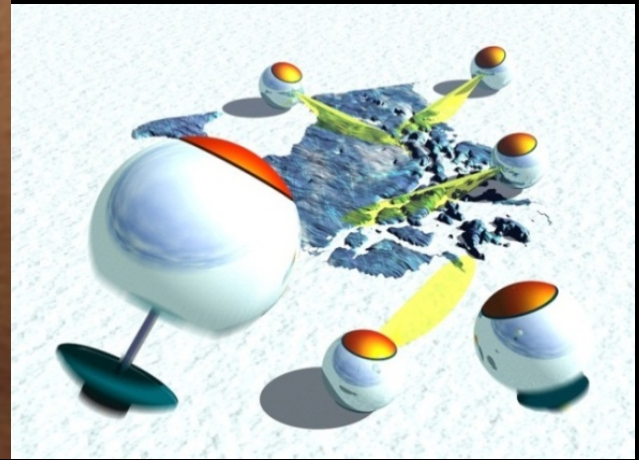
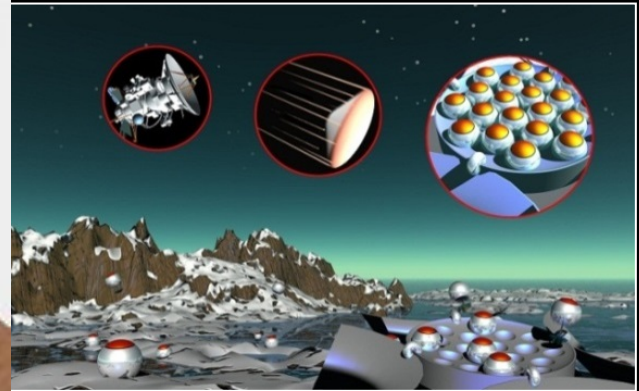
Salil Rabade, Nathan Barba, Laurence Garvie, Jekan Thangavelautham
School of Earth and Space Exploration
Arizona State University

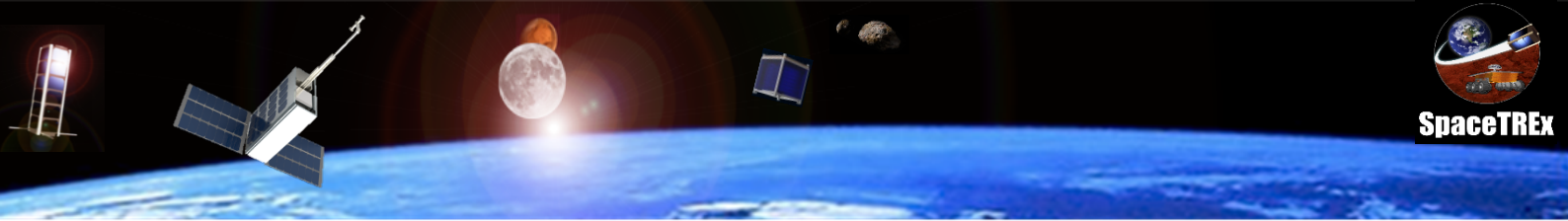
Motivation – Interplanetary Exploration





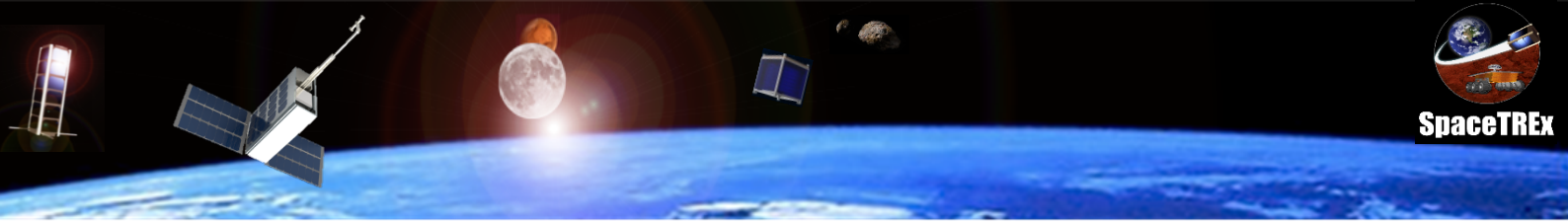
Planetary Surface Exploration/Prospecting





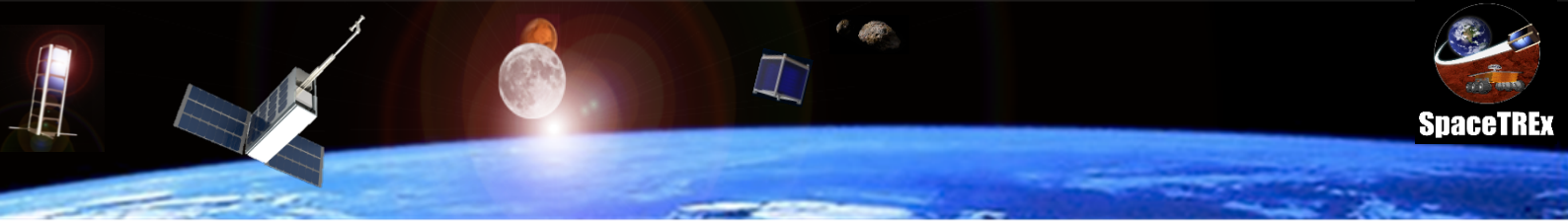
Need for a Sustainable Propulsion System

- High or good-enough delta V
- Long storage life, minimal thermal footprint
- High thrust – particularly for capture burns.
- Low-storage risk, safety
- Propellants that are compatible with ISRU methods
- Ease in refueling



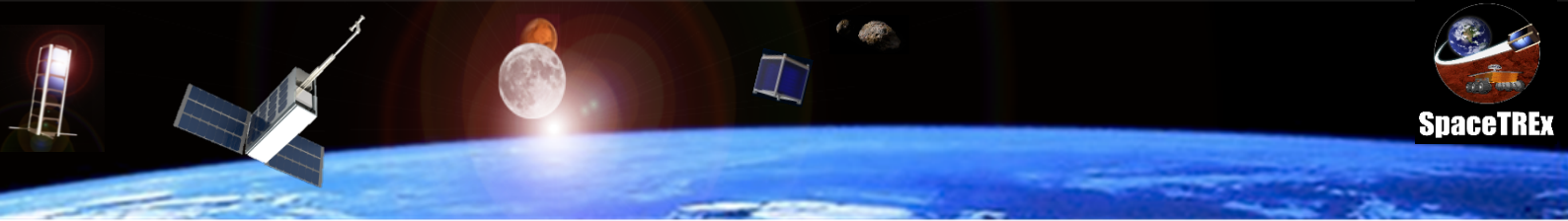
Water Steam vs. Water Electrolysis

- We are pursuing both approaches.
- Focus here will be water steam and for context we will compare against water electrolysis



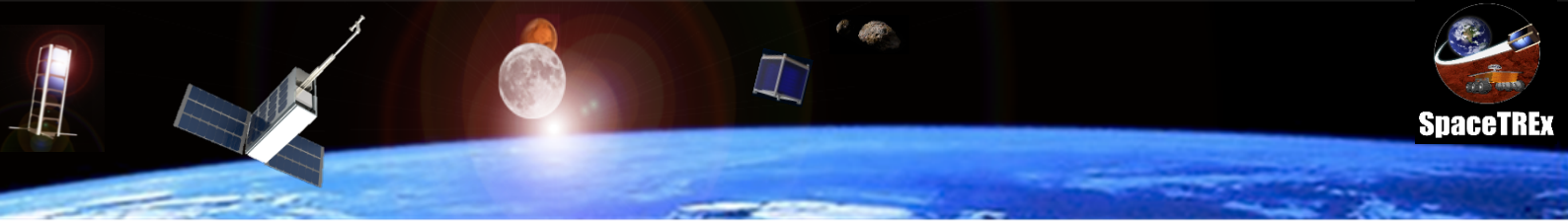
Water Steam Propulsion

- Properties of steam well understood
 - Minimal storage risk
- Isp up to 200 s
 - Generated from super heated steam, +1500 C
- Use of carbon nanoparticles to heat the water using solar thermal concentrator.
 - Free power sources, up to 99 % of sunlight can be converted to heat.



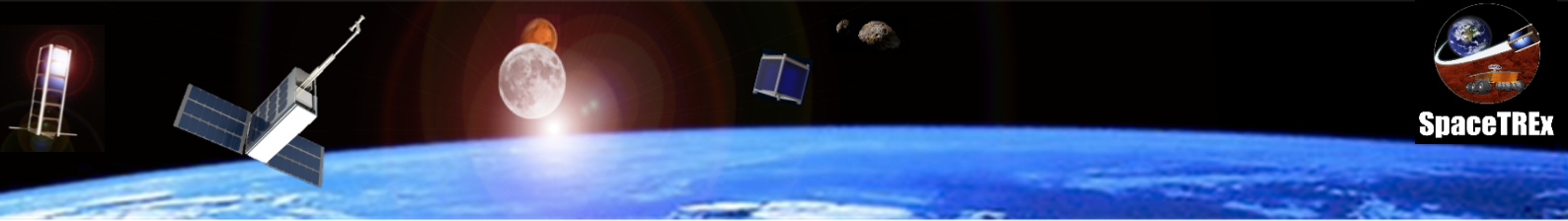
Steam Propulsion

- Achieve high temperatures with low-fidelity concentrator
- Compact, solid-state heating platform
- Comparably high thrust force compared to electrical propulsion methods.



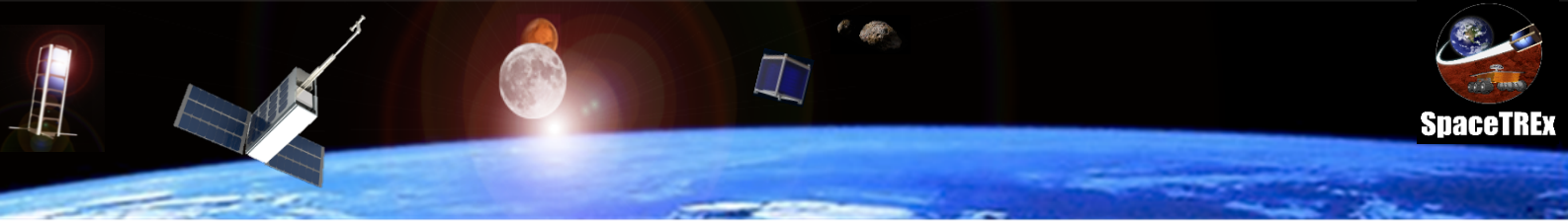
Steam Propulsion

- Comparable in Isp to mono-props
- Compatible with ISRU methods to extract water from Deimos/Phobos, asteroids, Moon.
- Not a solution for Mars ISRU

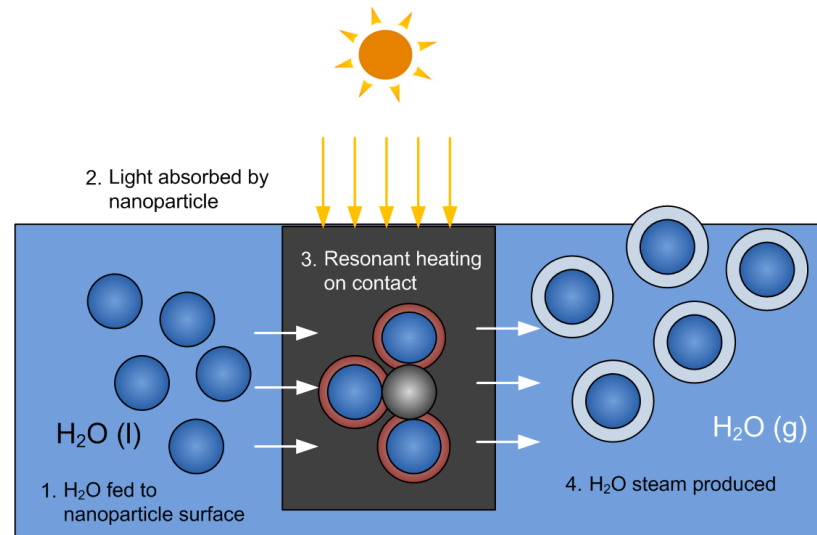


Steam Propulsion

- Steam propulsion is not a new concept.
- Old concepts required fission reactors, were meant to be large tug-boats of cis-lunar space.

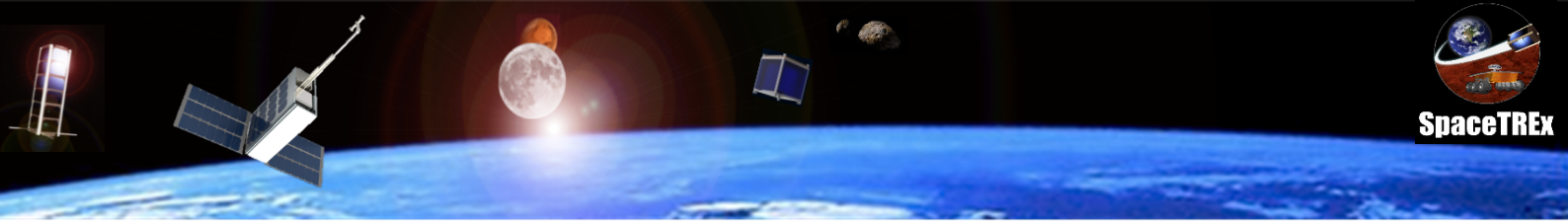


Steam Generation Using Carbon Nanoparticles



Source: "Solar Vapor Generation Enabled by Nanoparticles" Halas, et al. 2013

- **Concentrated light absorbed due to sub-wavelength geometry.**
- **Particles resonate, collect and transfer energy as heat.**



Steam Generation Using Carbon Nanoparticles

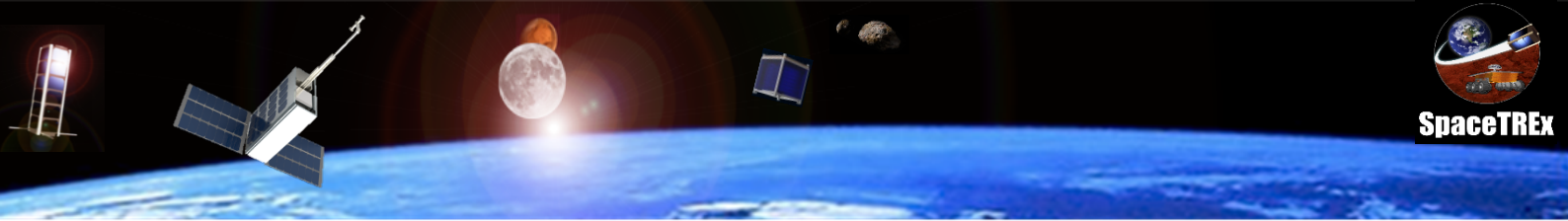
- Carbon Black N115, 82-86 % absorptivity
- Vanta Black, 99.5 % absorptivity



Carbon Black N115

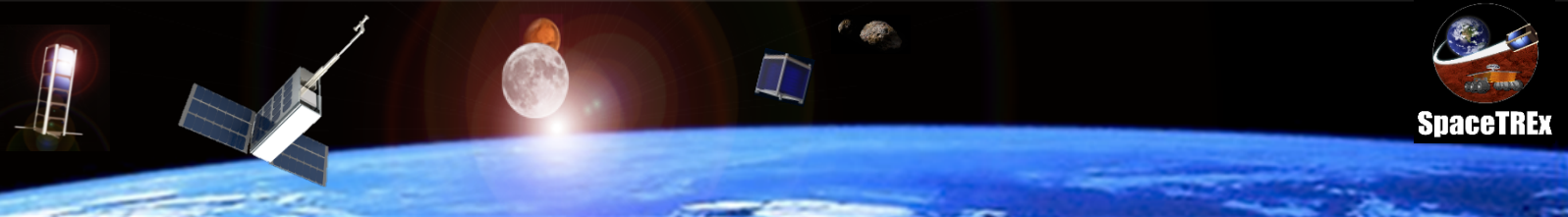


Vanta Black

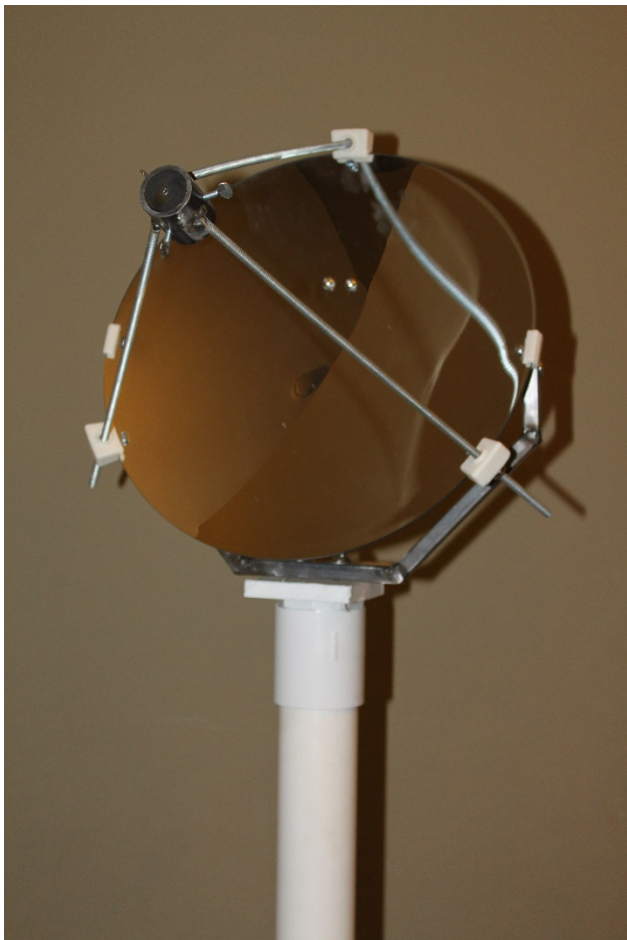


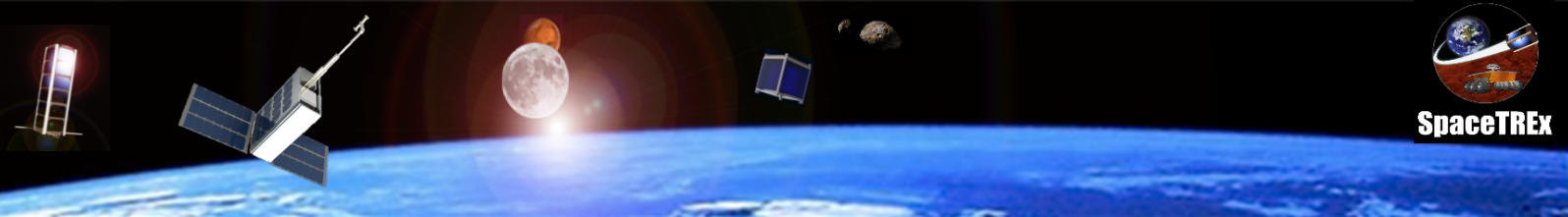
Steam Generation Using Carbon Nanoparticles

- These nanoparticles may naturally occur on asteroids.
- Heating occurs on a molecular scale. Very precise and local.
- Some of this may be a challenge for in-space propulsion.

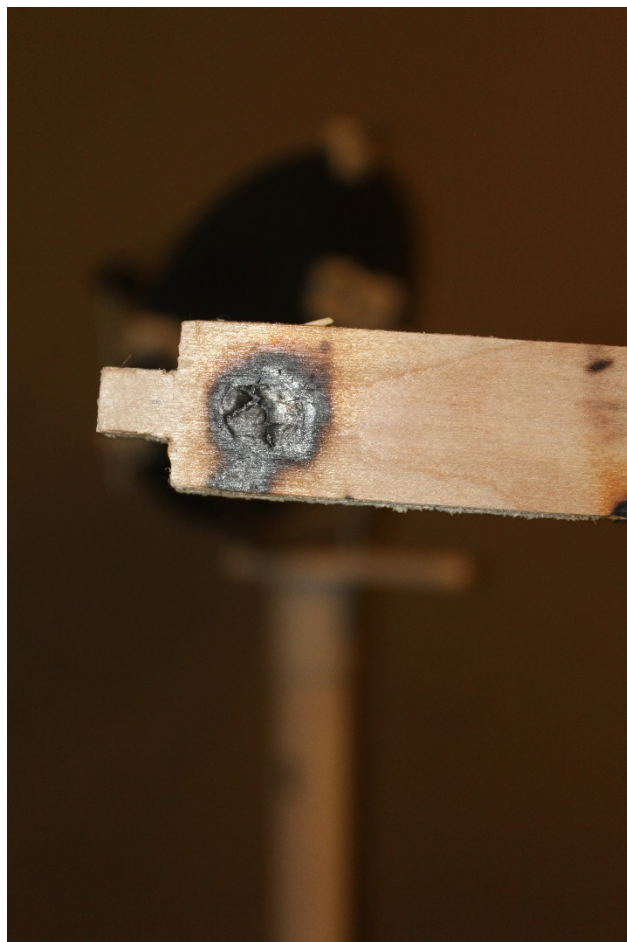


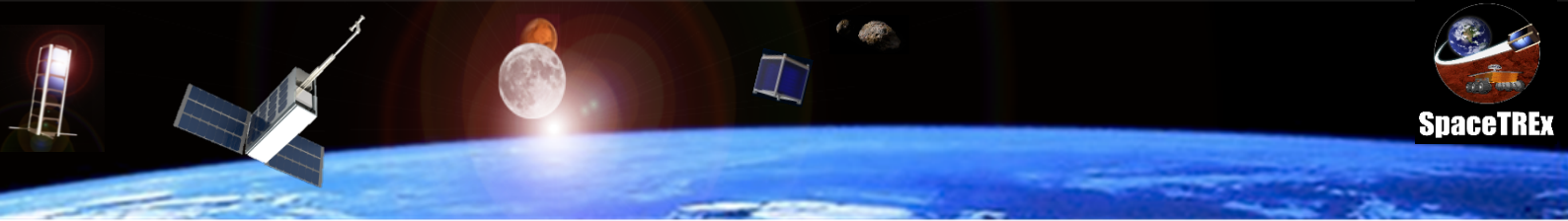
Laboratory Experiments





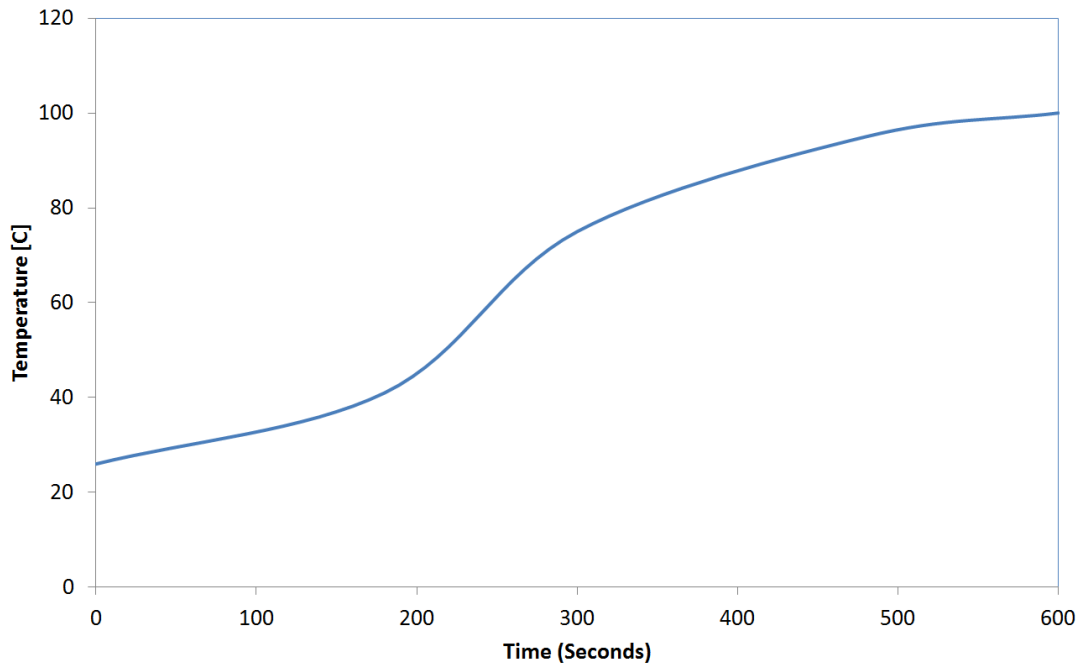
Initial Tests

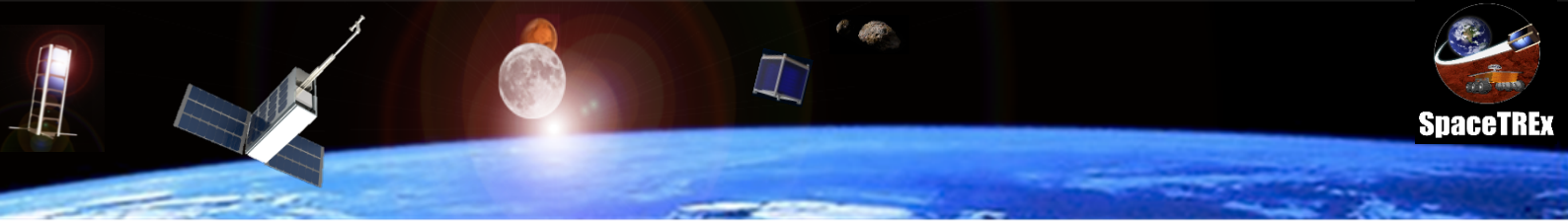




Concentrator Test Results

- Time for wood to catch fire: ~5-7 seconds
- Time for 15 ml of distilled water to boil: 10 minutes got upto 100°C
- Steam was evident ~5.5min



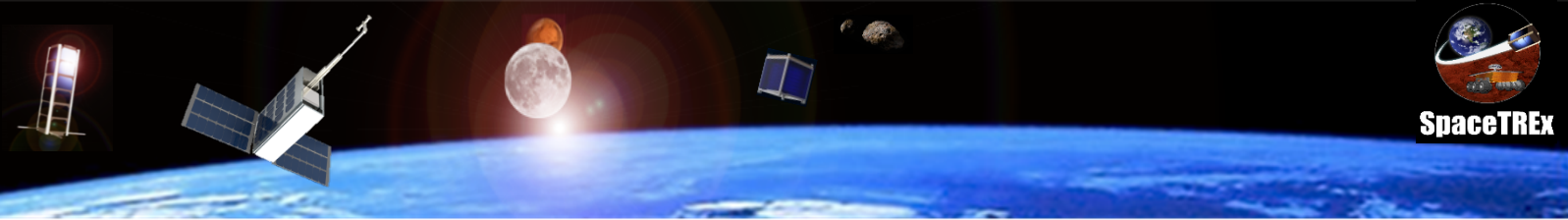


Early Results with Carbon Nanoparticles

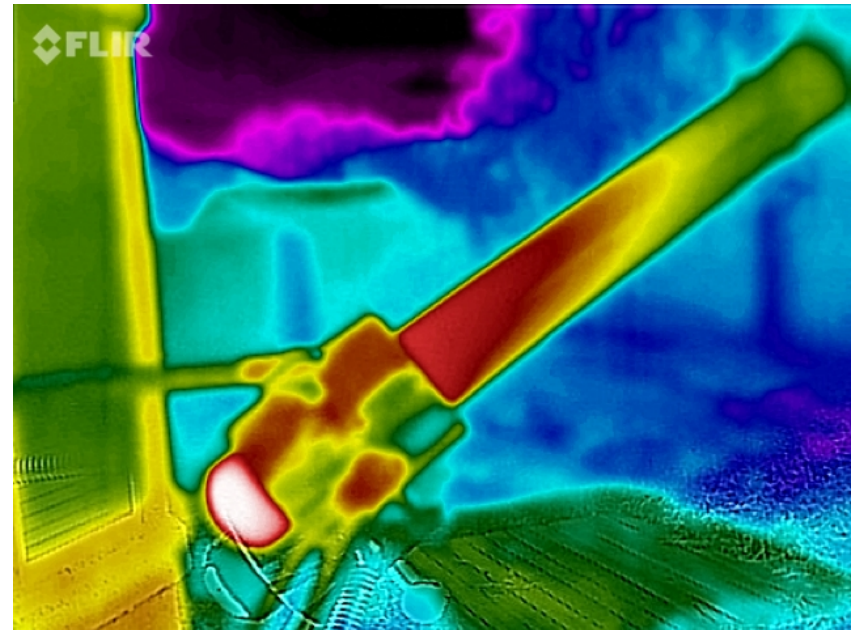
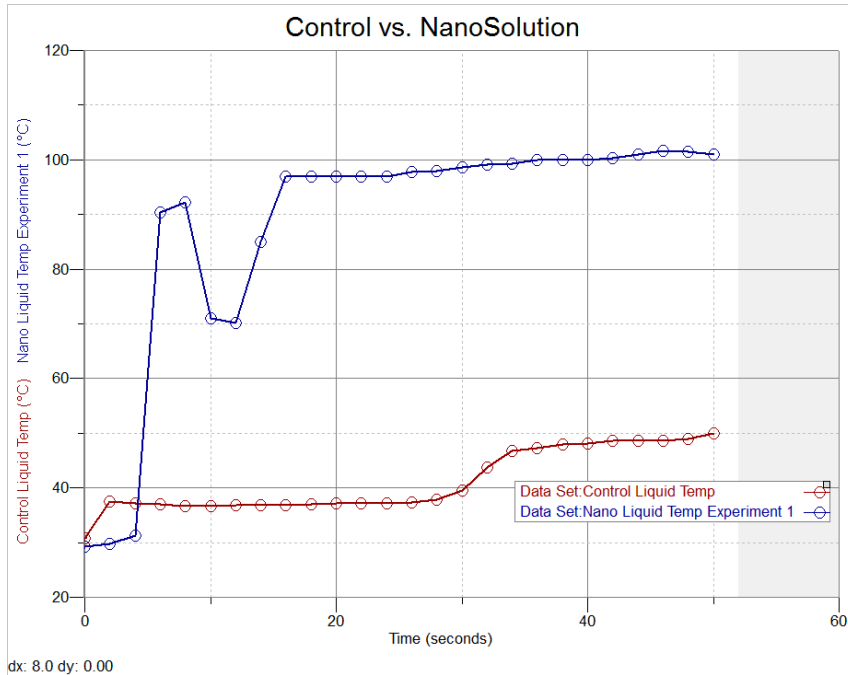
- Solar to heat conversion efficiency: 82 %

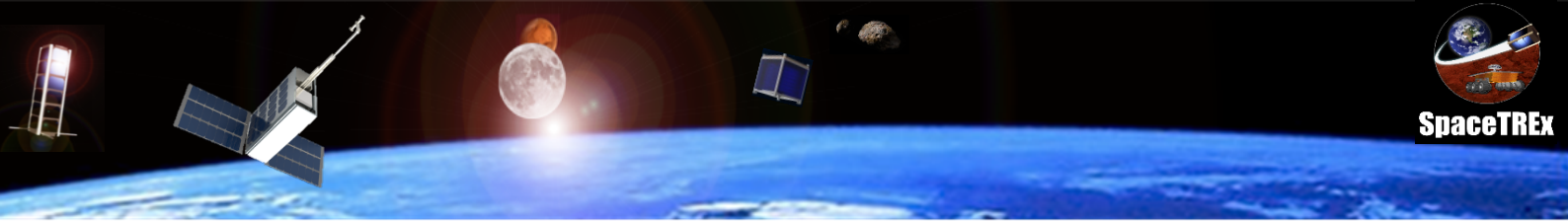
Condition	Time to Boil	Deviation
Distill Water	600 sec	10 %
Distill Water + Nanoparticles	40 sec	15 %

- Up to 15x shorter time to boil.



Early Results with Carbon Nanoparticles

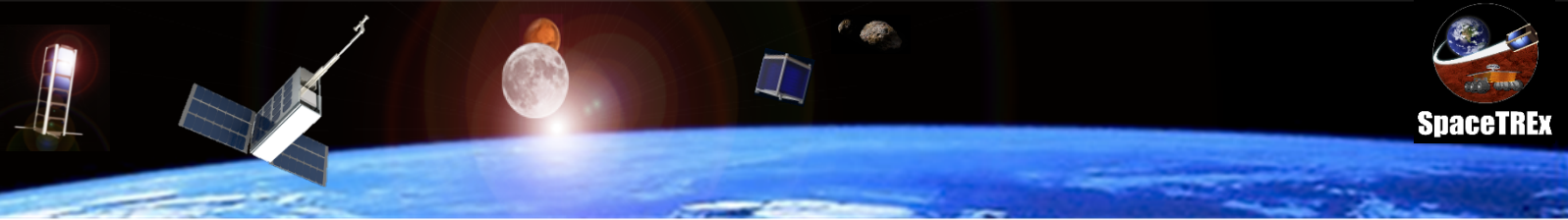




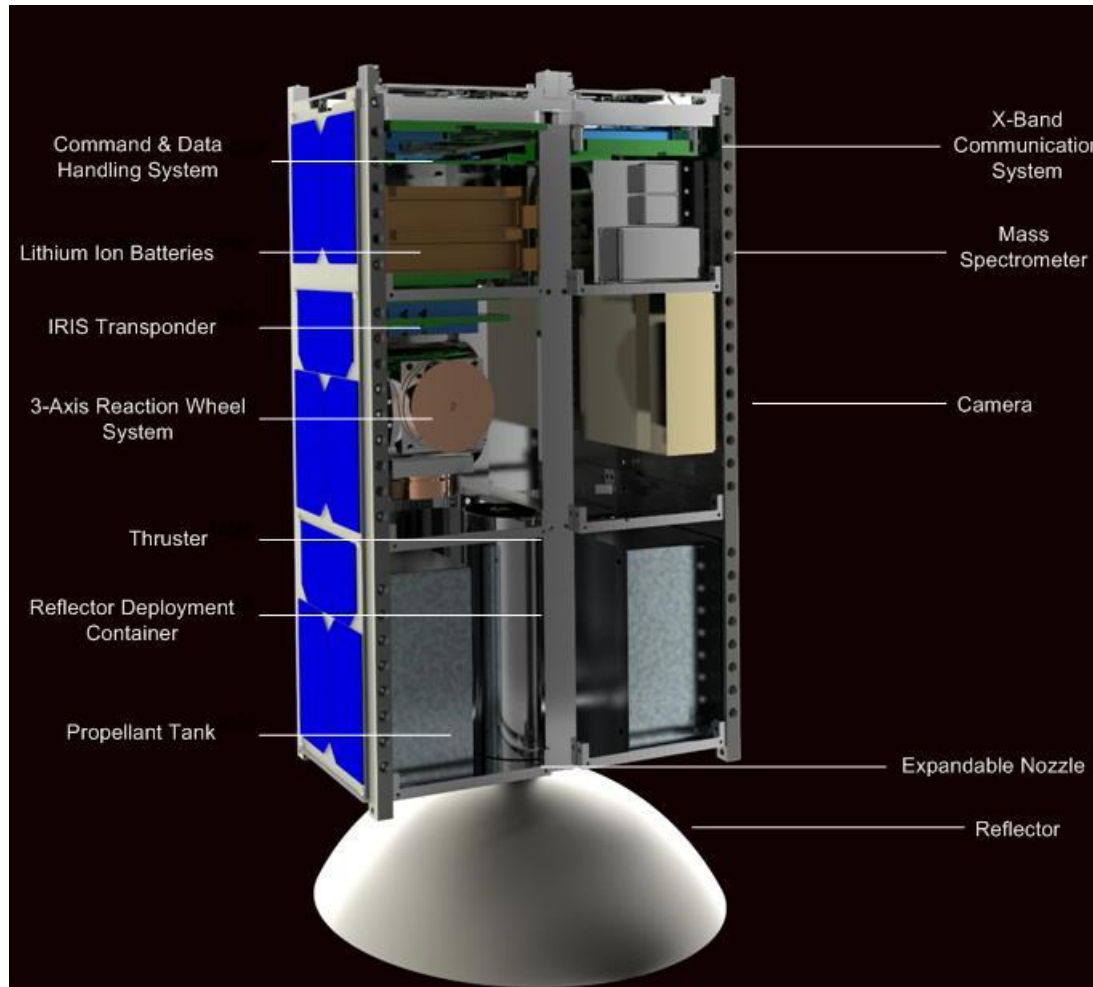
Solar Thermal Steam Propulsion Concept

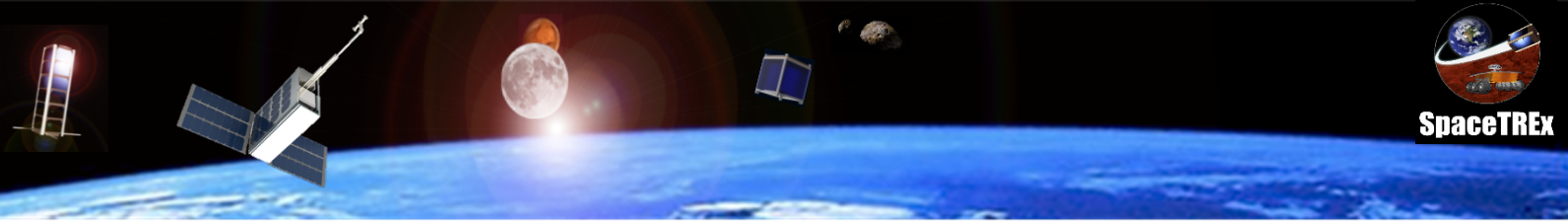
Components	Mass (kg)	Volume (L)
Propellant	7	7
Propulsion Dry Mass	3	1
Comms & ACDS	2	1
Structure	1	0.5
Power	1	0.5

- **Aggressive, optimized design.**
- **Delta V = 1400 m/s vs. 2500 m/s for Photovoltaic Electrolysis.**

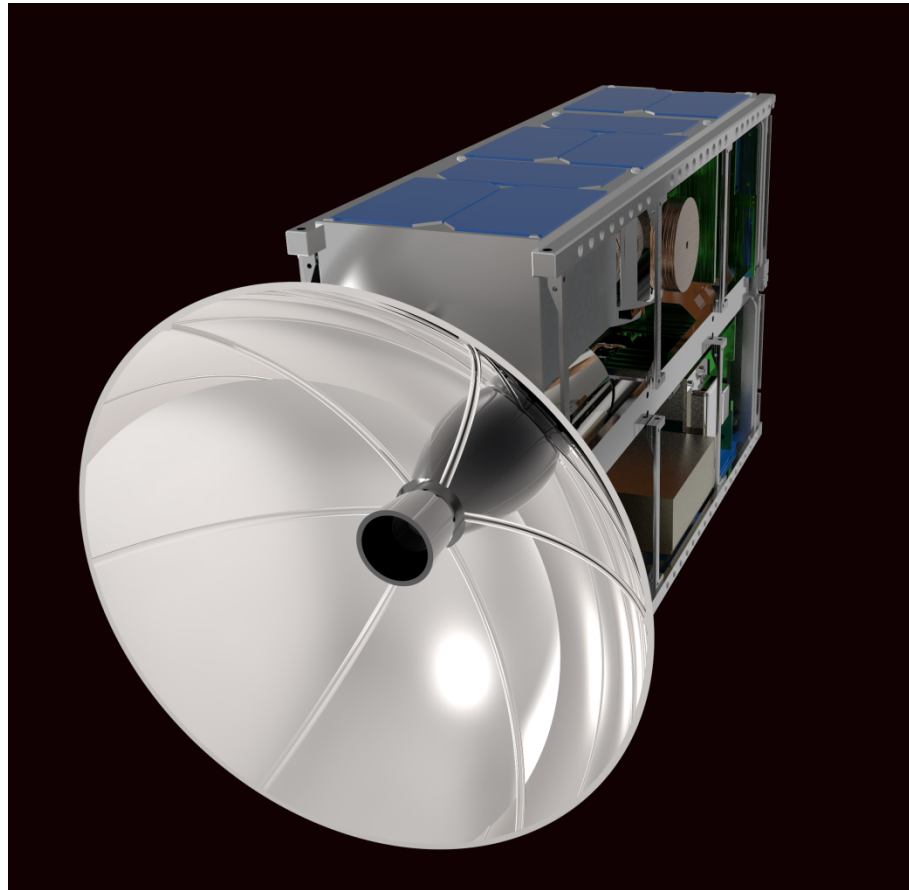


Solar Thermal Spacecraft Concept

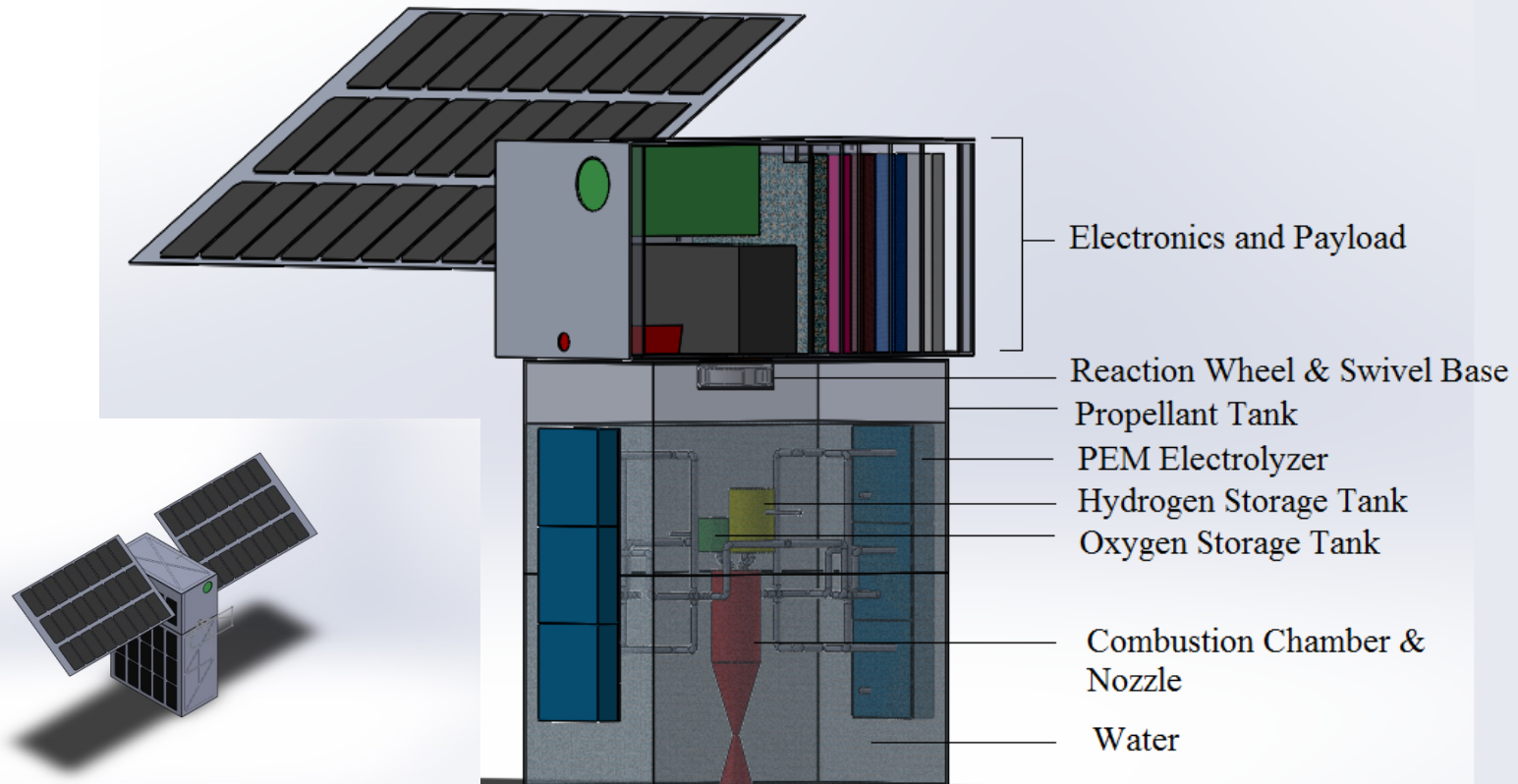


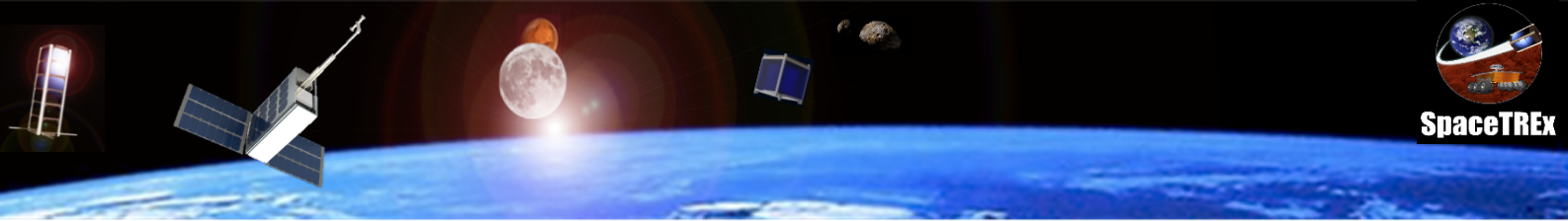


Solar Thermal Spacecraft Concept



PV Electrolysis Spacecraft Concept

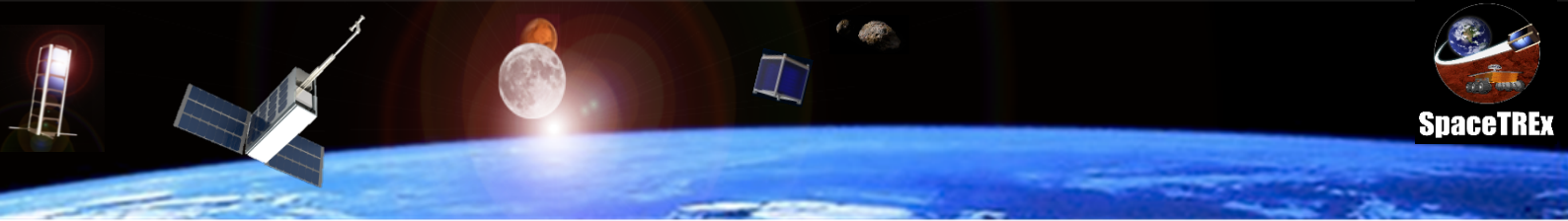




System Performance

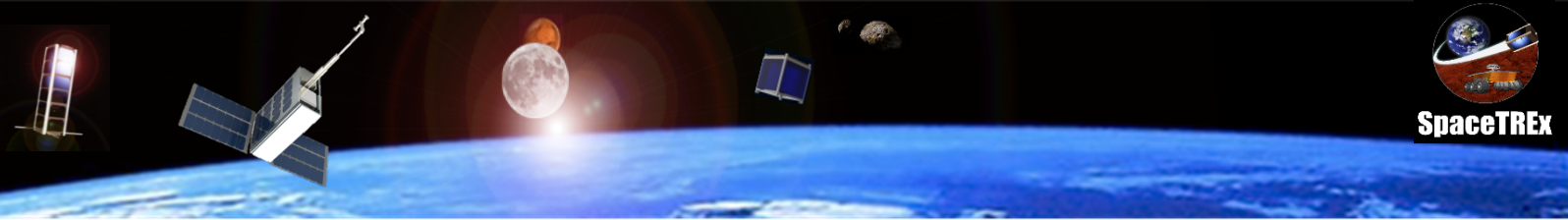
- For short pulses
 $I_{sp} = 360s$
- For a 6U CubeSat
 - Total Mass – 14 kg
 - Dry Mass – 4.5 kg
 - $\Delta V = 4000 \text{ m/s}$

To	Required ΔV
Low Lunar Orbit	4040 m/s
EML – 1	3770 m/s
EML - 2	3430 m/s

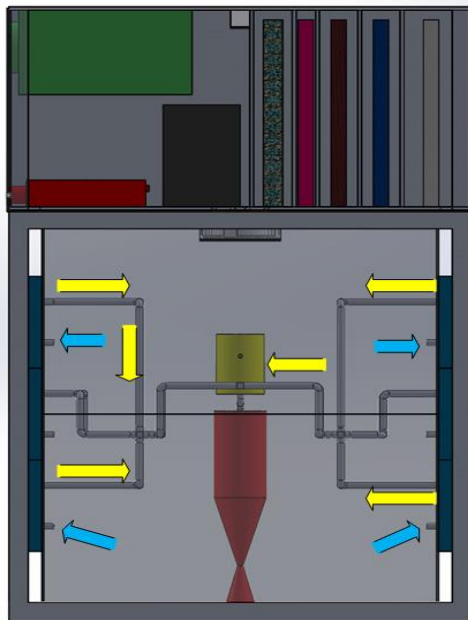


PV Electrolysis Spacecraft Concept

- Concept developed based on currently available state of the art CubeSat components
- PEM Electrolyzer with 85-90 % conversion efficiency
- Dual-body design, lower stage is centrifuge to separate water from the propellants.

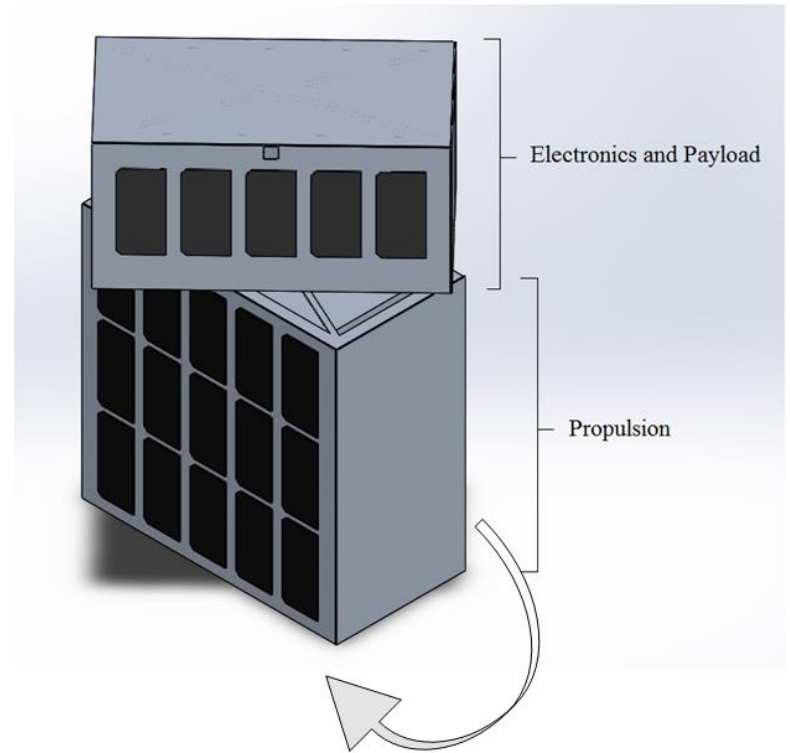


System Operation

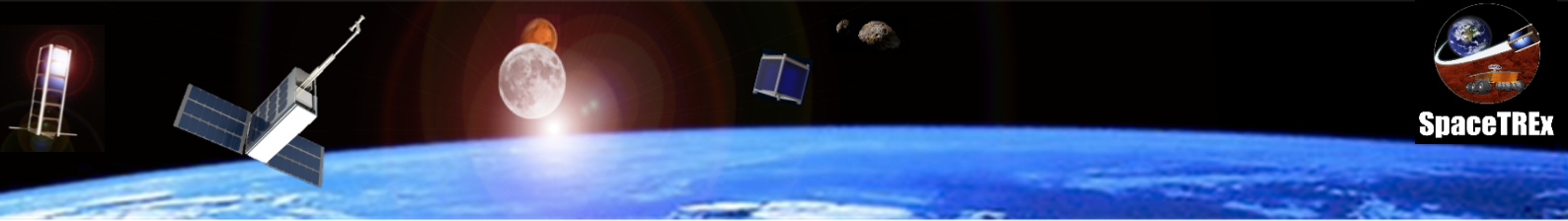


→ Water
→ Electrolyzed H₂

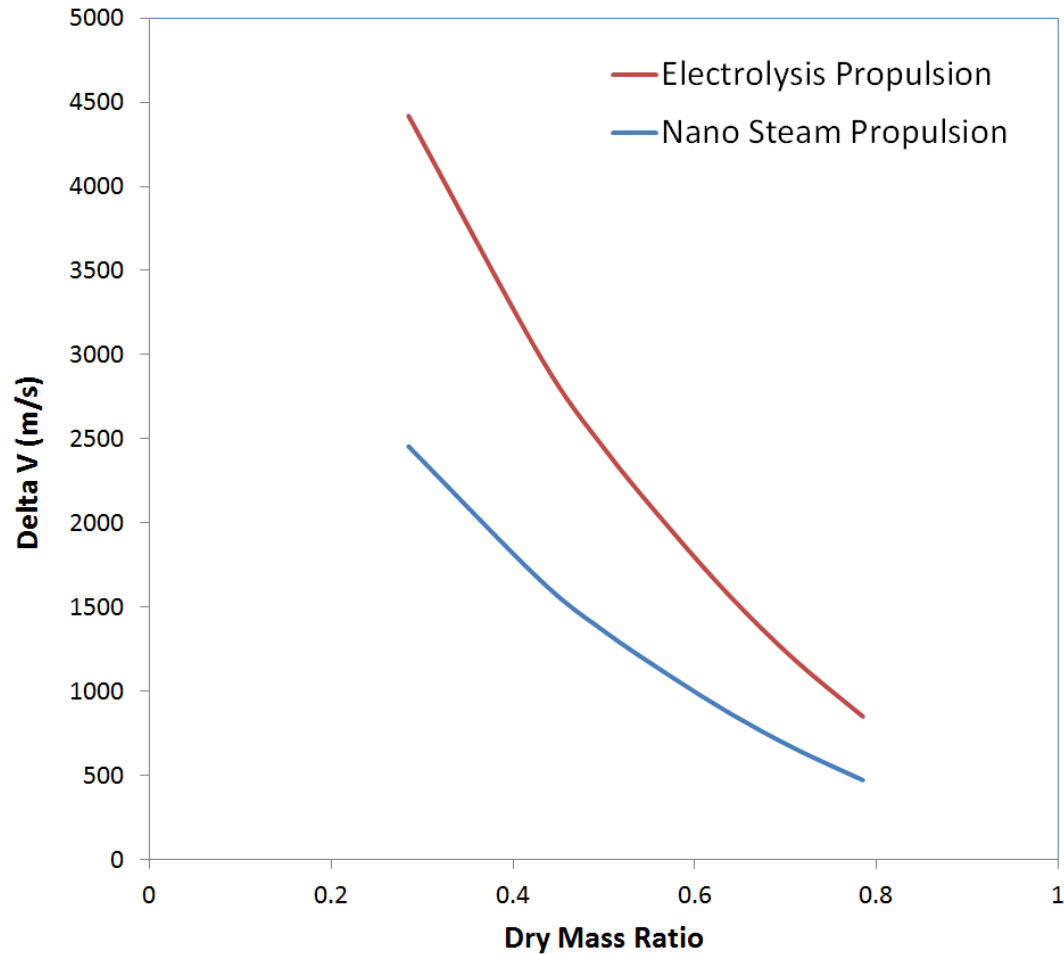
Side View of PVEPS

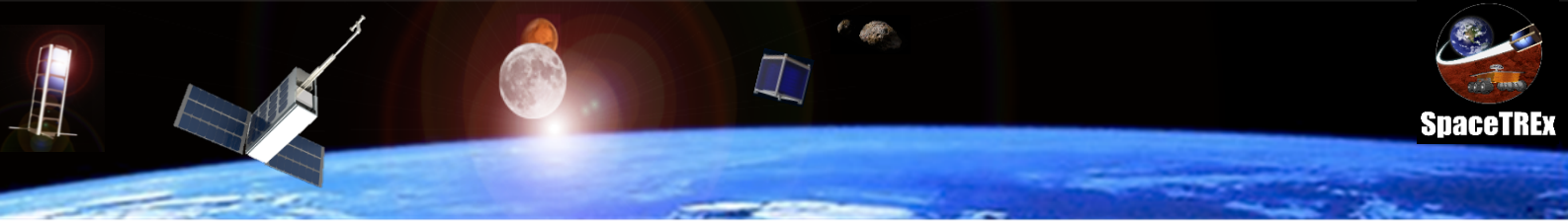


Lower segment rotates to separate water from reactants



General Case & Electrolysis Comparison

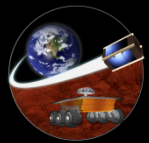


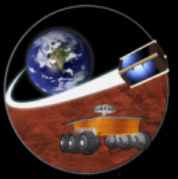


Discussion

- Overall a steam propulsion system is simpler, but at the cost of reduced performance by half
- Well aligned with a water-based cis-lunar economy concept.
- Water can be with impurities, particularly sulfur, carbon monoxide
- Easier adoption, until advancements make electrolysis options more practical.

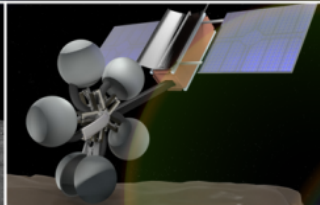
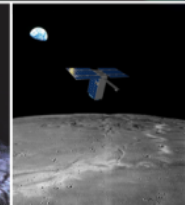
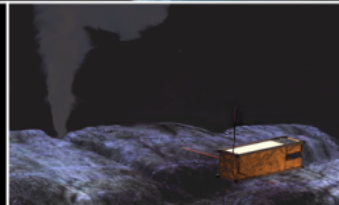
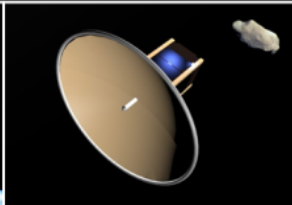
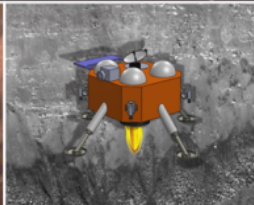
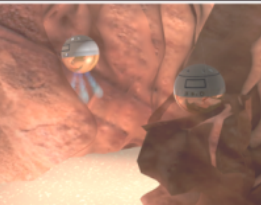
SpaceTREx Team

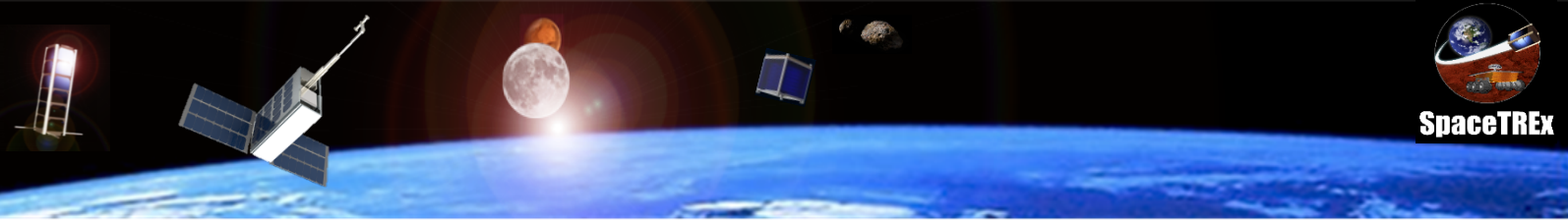




SpaceTrex

Adventure Awaits





Thank You!