



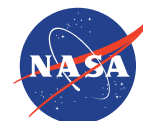
Enabling University-Operated Ground Support Development with the Morehead State University 21 m Ground Station

Interplanetary SmallSat Workshop

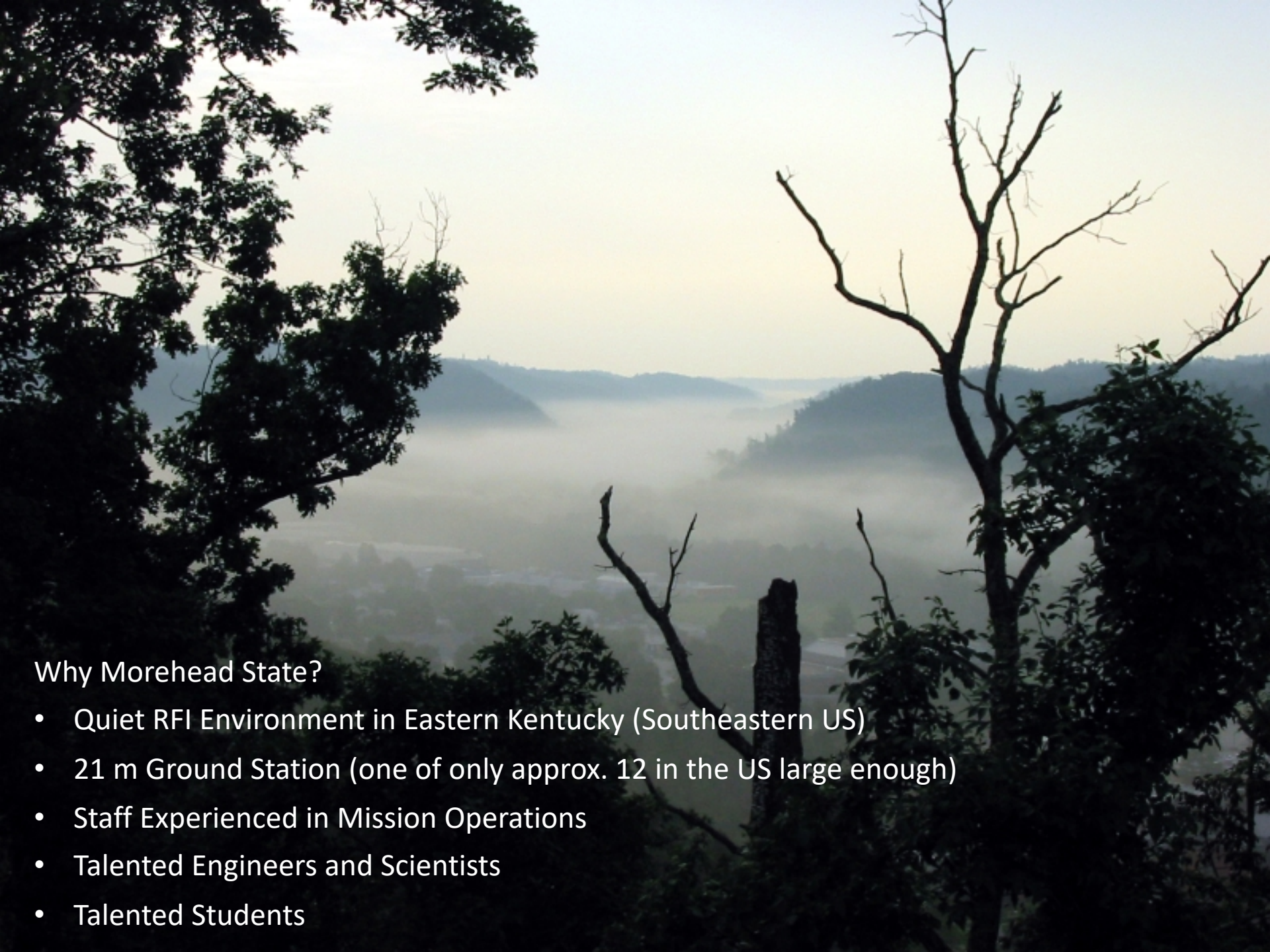
May 1, 2017

Jeff Kruth, Ben Malphrus (MSU)

Jay Wyatt, Tim Pham (JPL)



In Partnership with
Jet Propulsion Laboratory
California Institute of Technology



Why Morehead State?

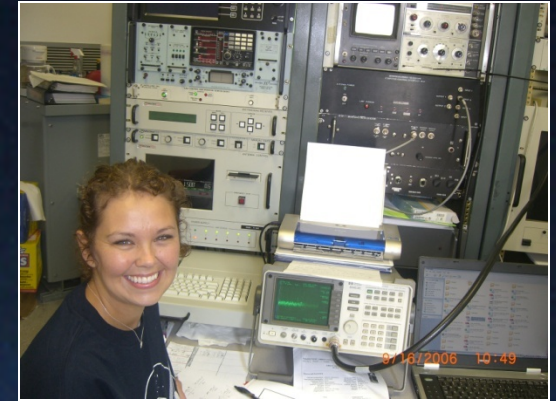
- Quiet RFI Environment in Eastern Kentucky (Southeastern US)
- 21 m Ground Station (one of only approx. 12 in the US large enough)
- Staff Experienced in Mission Operations
- Talented Engineers and Scientists
- Talented Students

21 Meter Space Tracking Antenna



- Specifications by MSU faculty with **NASA assistance**
- **Dual Purpose Instrument**
 - Ground Station for Smallsats
 - Radio Telescope for Astronomy Research
- Funded \$3.4 M -a variety of sources- Morehead State, Federal and State Funds, KSTC, NASA
- Built and Installed by VertexRSI (General Dynamics)
- Feeds Designed and built by VertexRSI, APL, and MSU
- Operational in 2006

Space Projects Create Opportunities for Students



- Undergraduate Research Experiences
- Instrumentation Experience
- Engineering Design
- Observational Astrophysics Research
- Ground Ops (TT&C)
- Project Management Experience
- Systems-level Engineering Experience

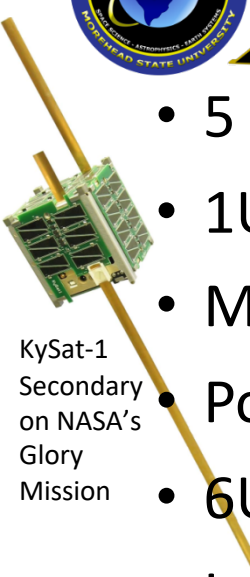




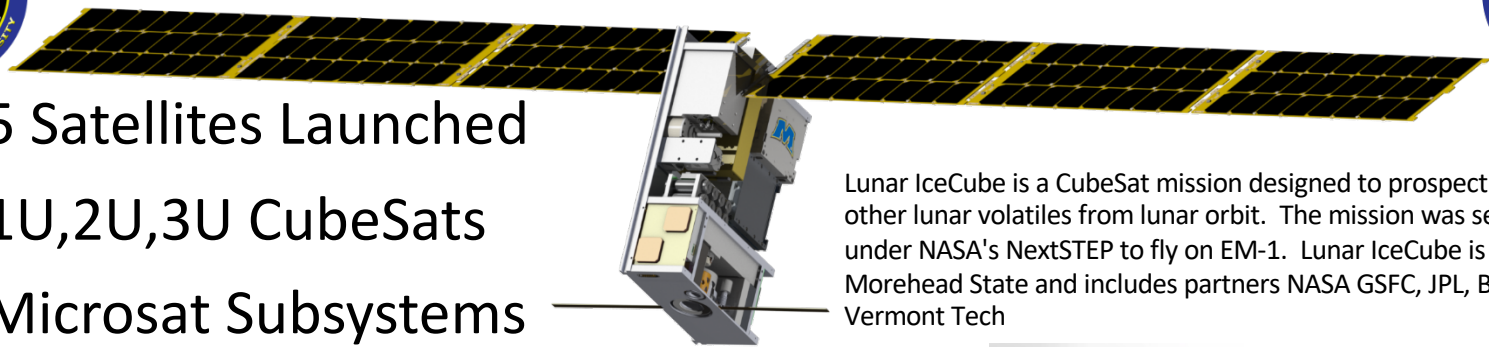
Morehead State SmallSat Missions



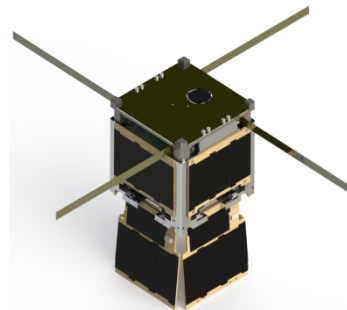
- 5 Satellites Launched
- 1U,2U,3U CubeSats
- Microsat Subsystems
- PocketQubs
- 6U Bus in Development
- Interplanetary Mission
- Variety of Customers



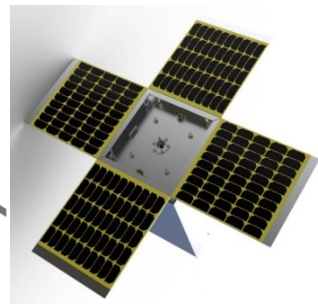
KySat-1
Secondary
on NASA's
Glory
Mission



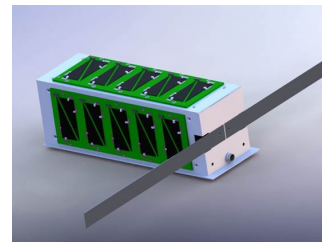
Lunar IceCube is a CubeSat mission designed to prospect for water ice other lunar volatiles from lunar orbit. The mission was selected under NASA's NextSTEP to fly on EM-1. Lunar IceCube is led by Morehead State and includes partners NASA GSFC, JPL, Busek, and Vermont Tech



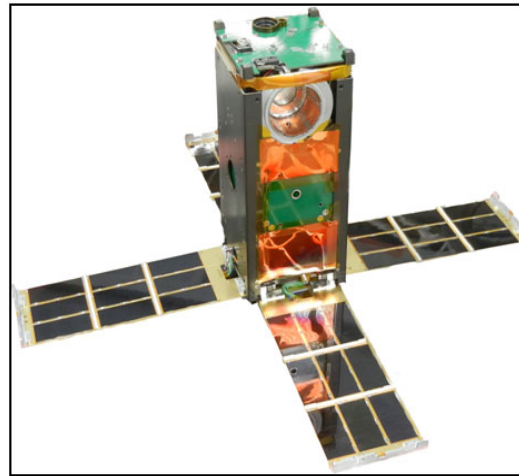
KySat-2 Launched in October 2013



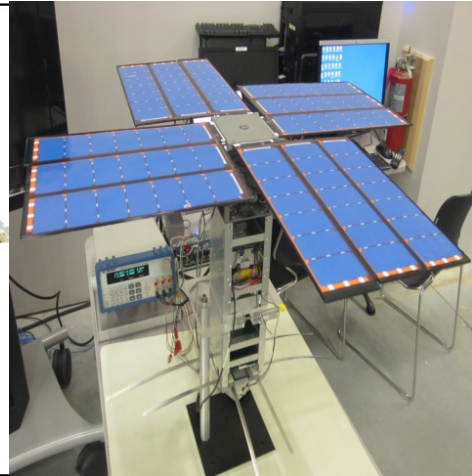
UniSat-5 w/ Univ. of Roma-GAUSS launched 2014



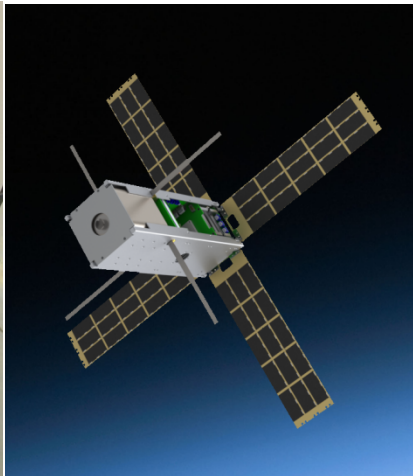
T-LogoQube (Eagle-1) Launched in October 2013



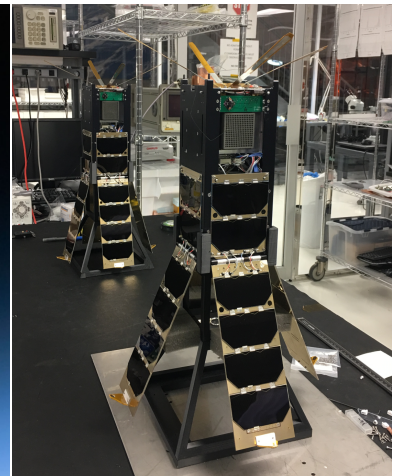
CXBN Launched in 2012



TechSat-1 In Developed for SMDC (w/ Radiance and Honeywell)



Standard MSU 3-U Bus



CXBN-2 Launched April 2017

21 M Operations

- Satellite Ground Station for Ground Ops/Mission Support for our SmallSat Missions and Others
- Radio Telescope Mode for Research in Astrophysics
- Test-Bed for Experimental Communication Systems for Small Satellite Systems



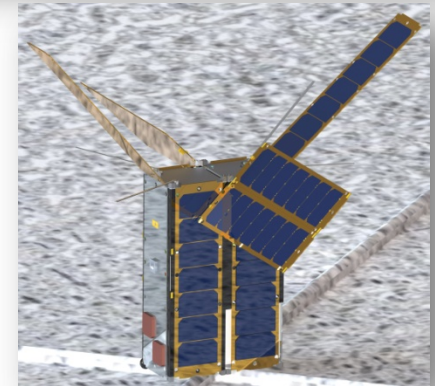
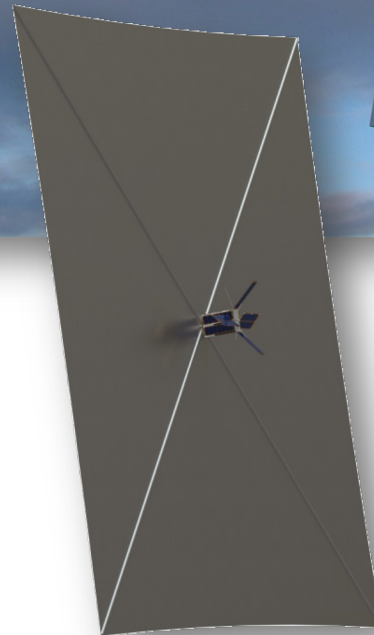
21 M Ground Station Mission Support

Support SmallSats Missions in:

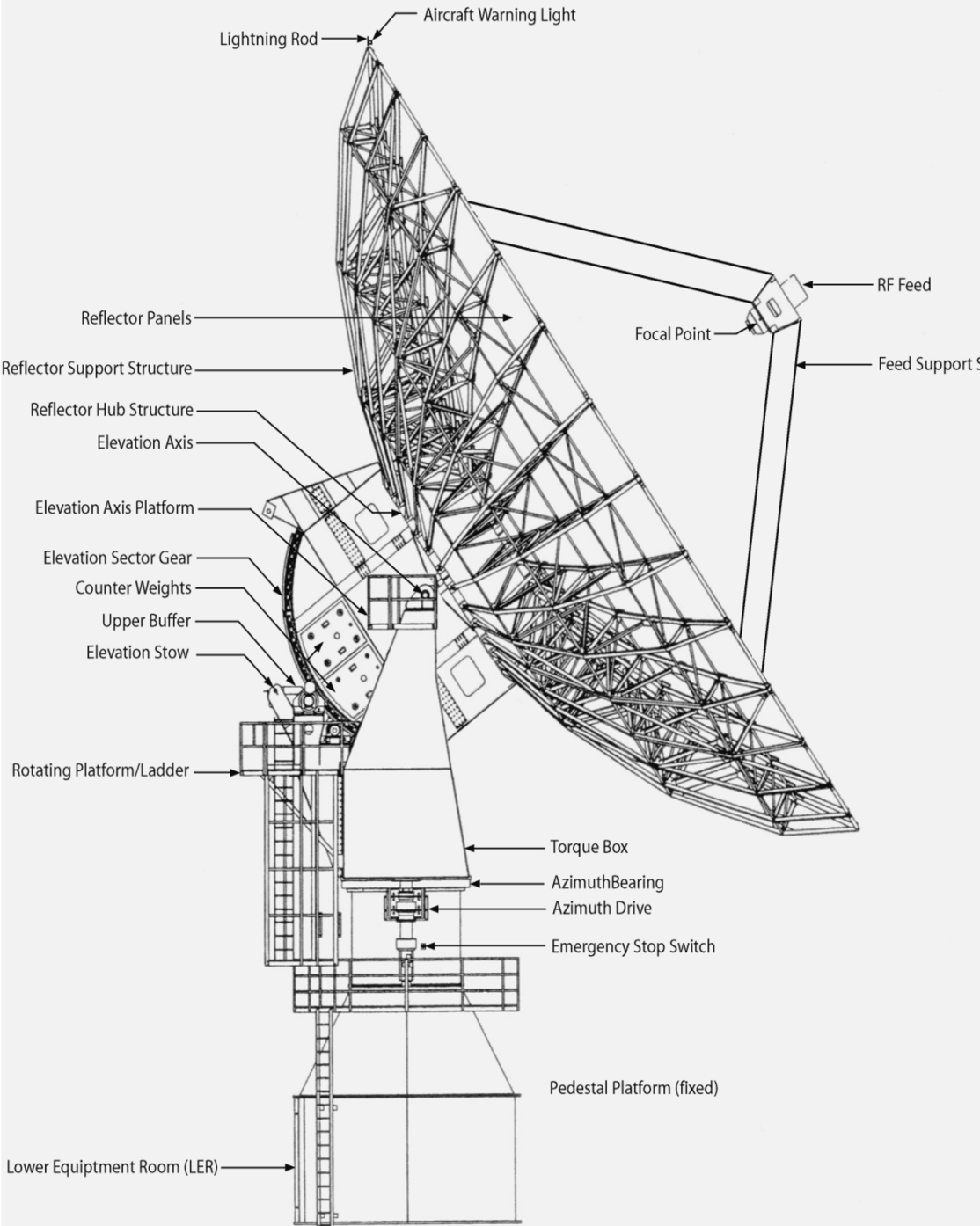
- LEO
- MEO
- GEO
- Lunar
- Near-by Asteroids
- Approaching Comets

Missions Operations For:

- Planet Labs Dove Constellation
- LRO (Mini-RF Comms)
- ISEE-3 Reboot
- CubeSats and MicroSats:
 - CXBN, KySat-2, Firefly, T-LogoQube, Eagle-2, UniSat-5, EduSat, UniSat-6, SERPENS



21 M Overview



Parameter	Measured Values
Axis Slew Velocity	
Azimuth	> 3.0 °/sec minimum
Elevation	> 1.6 °/sec minimum
Polarization	> 0.7 °/sec minimum
Axis Acceleration	
Azimuth	1.0 °/sec²
Elevation	0.6 °/sec²
Travel Range	
Azimuth	± 269.8°
Elevation	1.0° to 90.3°
Polarization Range	± 90°
Pointing Accuracy	0.005° RMS
Tracking Accuracy	0.0004° RMS
Aperture Efficiency, η (L/Ku)	0.653/0.563
Surface Tolerance @ 35 mph wind	< 0.020" RMS

Interplanetary SmallSat Ground Ops: Morehead State 21 M Ground Station- **Current State**

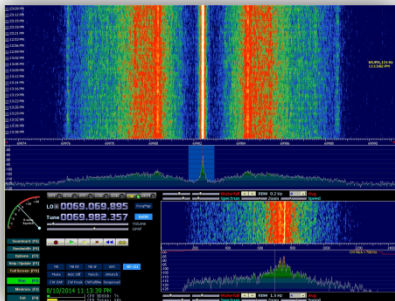


Morehead State University
21 M Ground Station

- Fully Operational, Full-Motion, 21 M Antenna
- Operational Experience: LRO, ISEE-3, Planet Labs, KySpace
- High Gain, Pointing and Tracking Accuracy
- Station is ideal for Inner Solar System Experiments
- Full Remote Control of All Systems
- X-Band Downlink Currently- Uplink capability planned
- NASA NEN Compatible
- Software-Defined TT&C Processor (SoftFEP) and High Data Rate Digitizer for Experimental Missions
- Extensive use of Student Operators (STEM Engagement)
- Plans for DSN Compatibility Upgrades with JPL assistance



Student Operators in the MSU 21 M LER



ISEE-3 Carrier During Lunar Fly-by Sept 2014



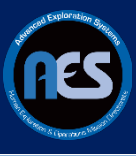
Student Operators in the MSU Mission Ops Center

MSU 21 Meter Current RF Capabilities

Radio Band	Frequency Range	Gain	Uses of Band
UHF	400-480 MHz	30 dBi	Satellite Telecom
S-Band	2.2-2.5 GHz	52.8 dBi	Both Satellite Telecom and Radio Astronomy
X-Band	7.0-7.8 GHz	62.0 dBi	Primarily Satellite Telecom
Ku-Band	11.2-12.7 GHz	65.50 dBi	Primarily Satellite Telecom



Morehead State University 21m Upgrade to DSN Compatibility



Goldstone, California



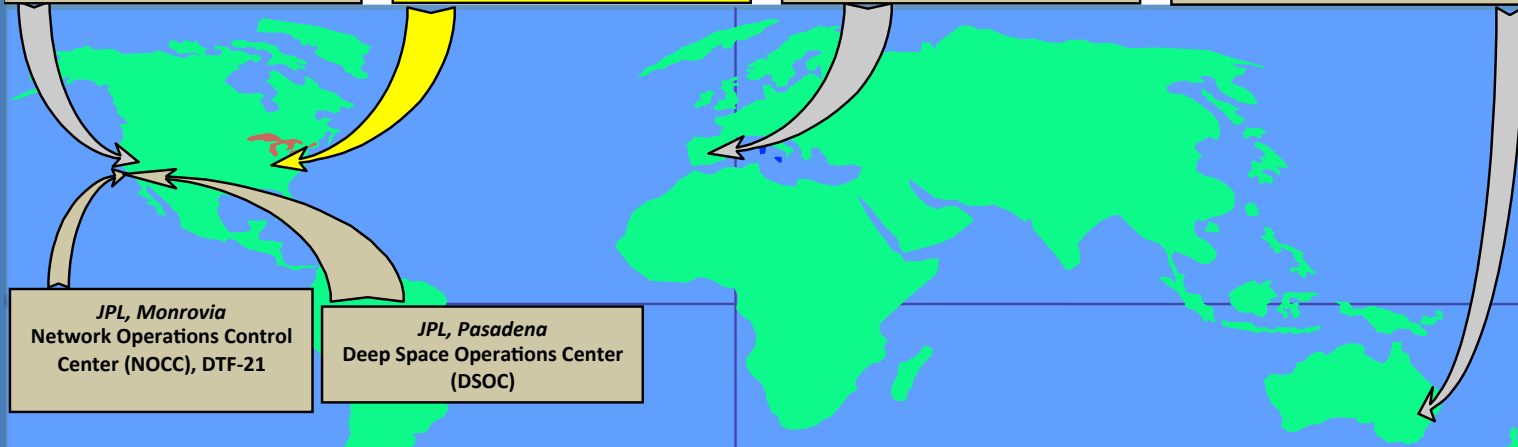
Morehead State Univ.



Madrid, Spain



Canberra, Australia



Objective:

- Demonstrate a cost-effective process for expanding DSN capabilities by utilizing non-NASA assets to provide communication and navigation services to small spacecraft missions to the Moon and inner solar system, thereby enabling interplanetary research with small spacecraft platforms.

Benefits:

- Serves as a test-case to define a path for other non-NASA ground stations to provide auxiliary deep space navigation and tracking support for small spacecraft missions.
- Develops an operational capability to support EM-1 Cubesat missions in the 2018 timeframe

Technical Approach:

- Develop and implement a strategy to transfer Deep Space Network (DSN) processes and protocols to the MSU 21 m antenna system and to upgrade the antenna hardware and software systems to enable integration into the DSN as an auxiliary station to support small spacecraft missions.
- The project is focused on the implementation of deep space communications, tracking and navigation techniques as well as adoption of CCSDS data interface standards such as the Space-link Extension service.
- Implement systems upgrades, conduct tests/demonstrations, and transition to an operational capability.

Contacts

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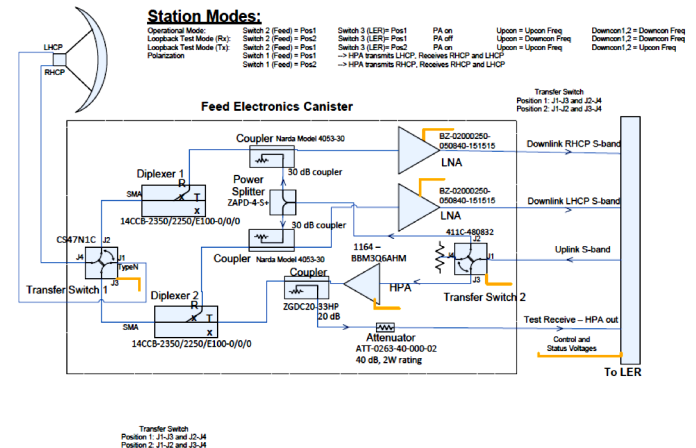
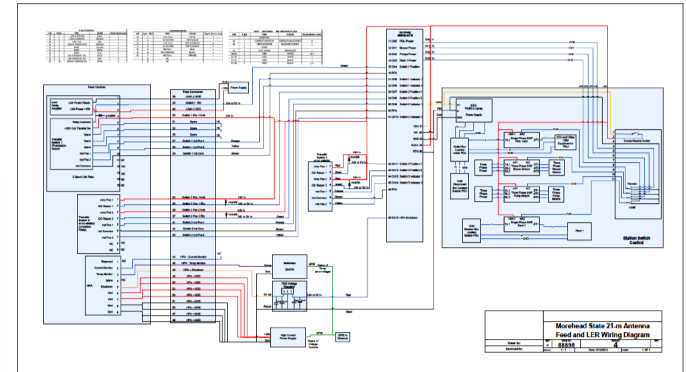
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 NASA Jet Propulsion Lab
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Enabling Interplanetary Smallsat Ground Support- Toward DSN Compatibility



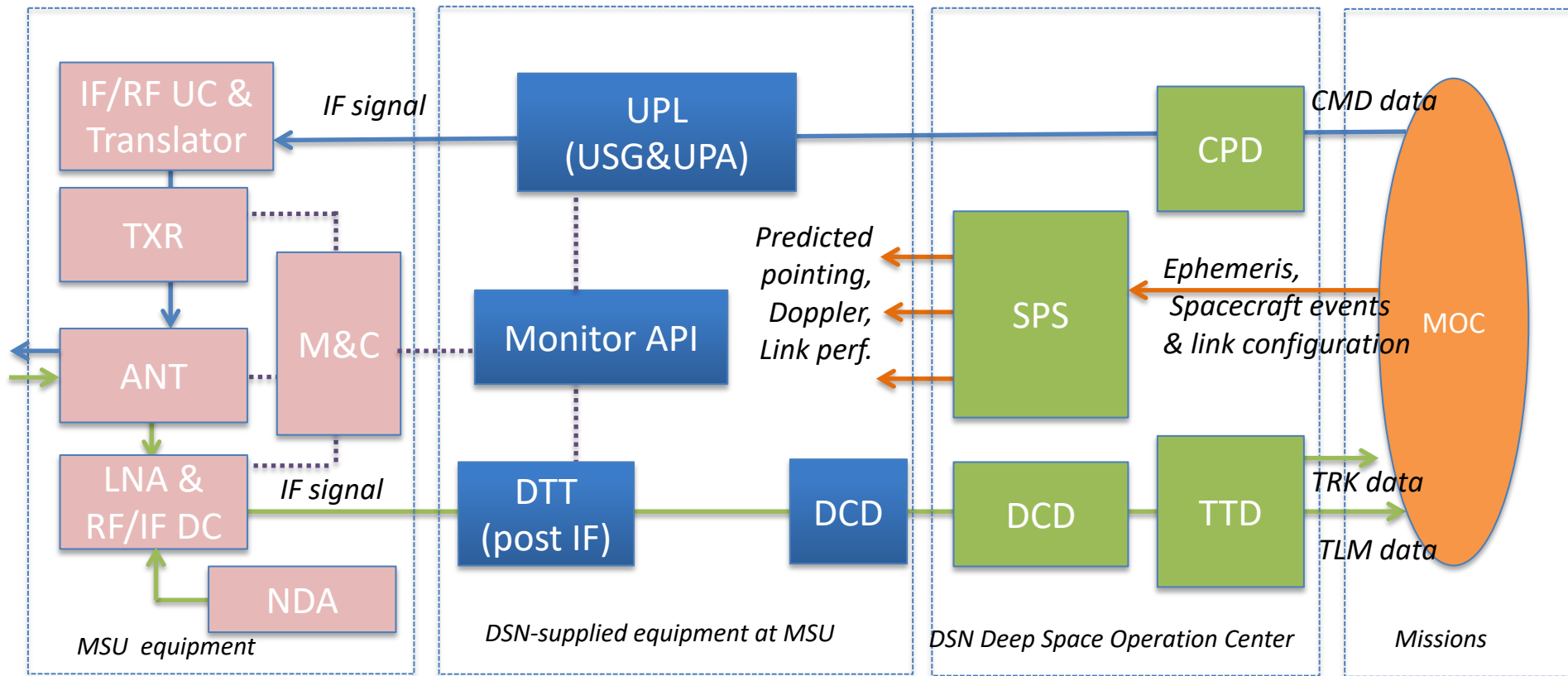
- Is Extremely Challenging:
- Only Three Stations in the World and Some (ESA and JAXA providing Cross Support)
- No Commercial Off-the-Shelf Equipment Exists
 - No Tracking Receivers
 - No Telemetry Receivers
 - No Uplink Transmitters
 - No Data Collection Systems
 - GPS Not Good Enough for a Time Standard
 - Atomic Clocks Not Good Enough for a Time Standard
- Existing X-Band Feed Not Adequate
- IF Processors Not Adequate
- All Equipment Must Be Custom Built
- IT and Physical Security Must Be Upgraded





New System Architecture Required

System block diagram

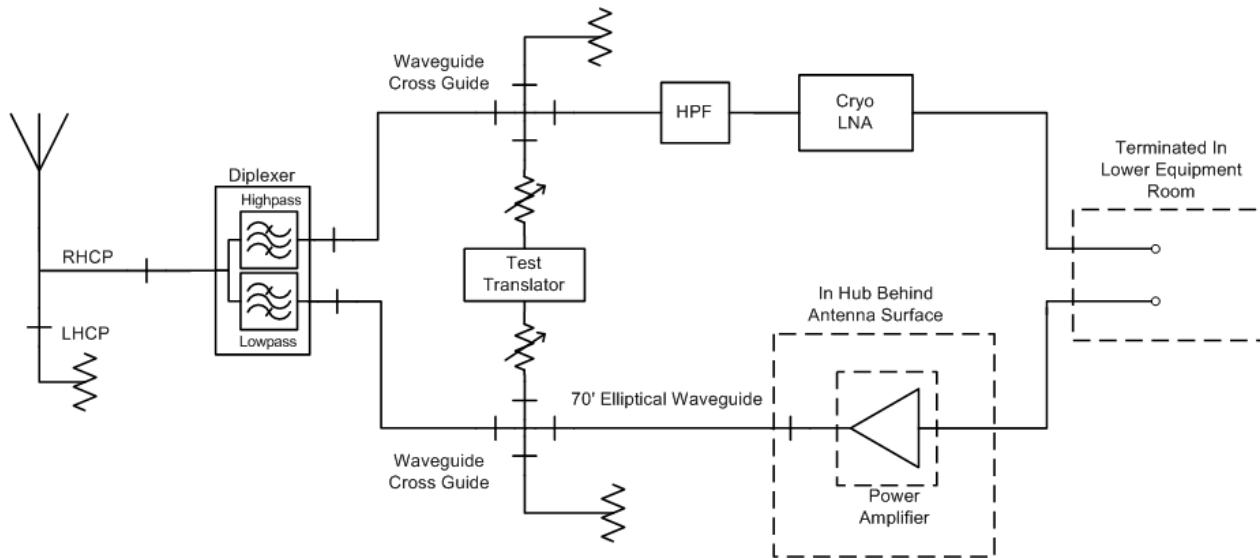
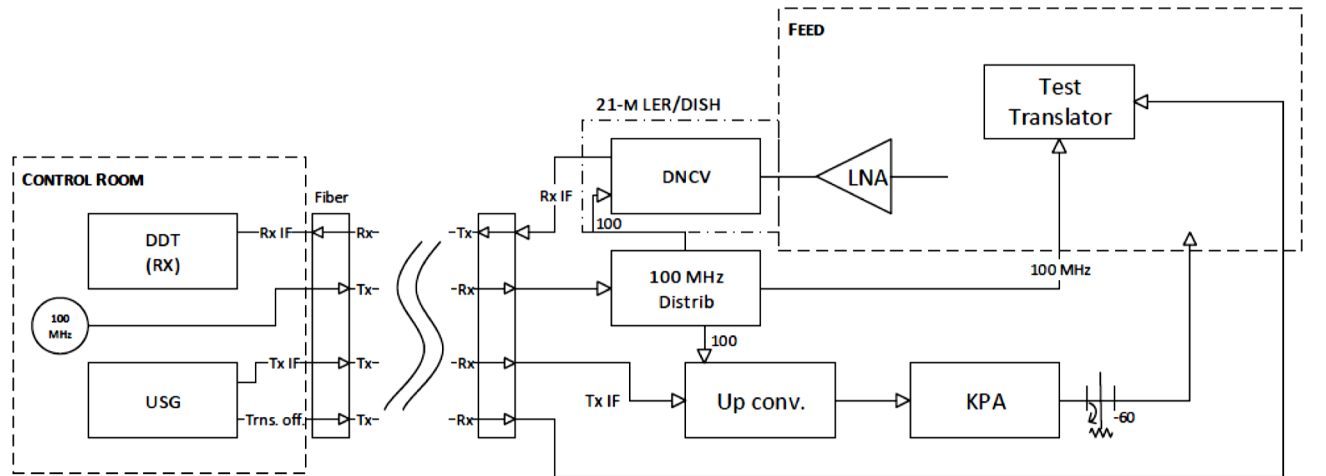




Becoming a DSN-Compatible Station



New X-Band Feed Required with Cryogenic LNA and High Power TX Capability



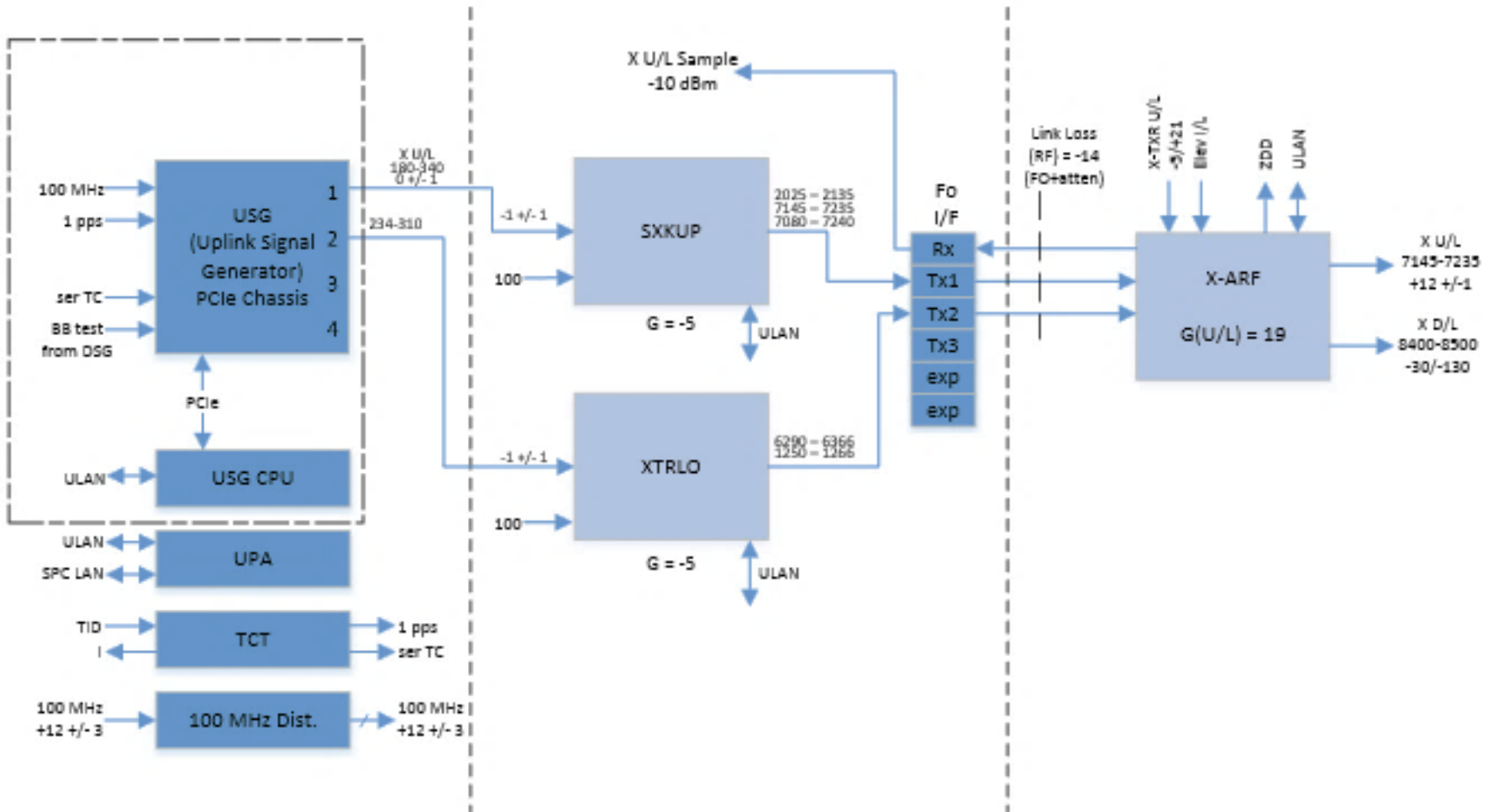


Becoming a DSN-Compatible Station



Command Capabilities Required

- USG-Based B6E/UPL
- 1 KW Power Amplifier

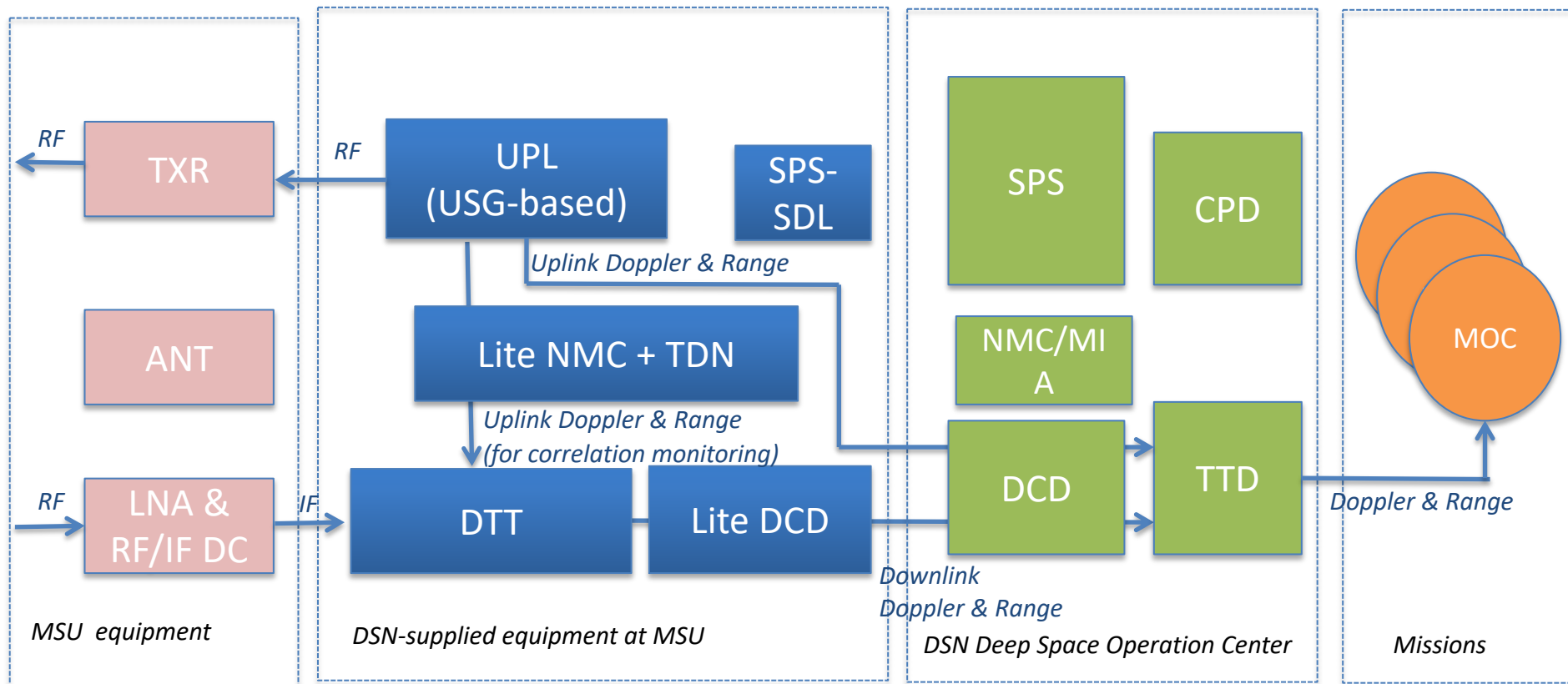




Tracking Capabilities

Tracking and Ranging Capabilities Required Additional Equipment and Processes

- UPL (Also for Commanding)
- NMC (Very Lite Version)
- DTT (Also for Telemetry Down)



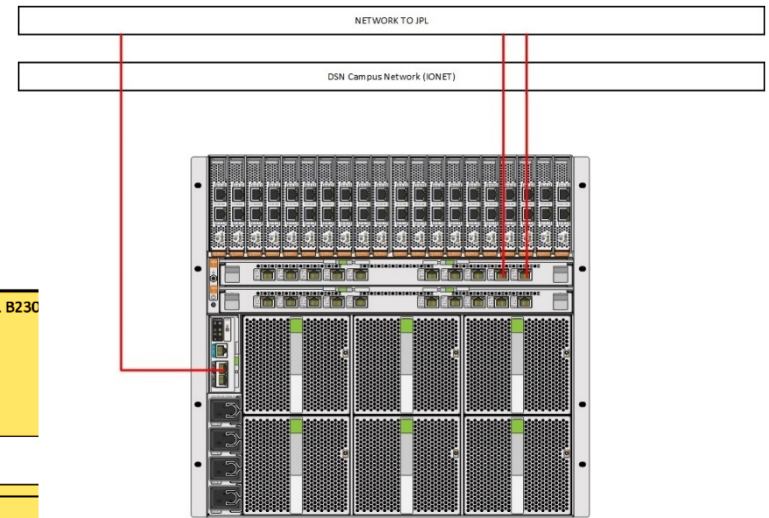


Becoming a DSN Station

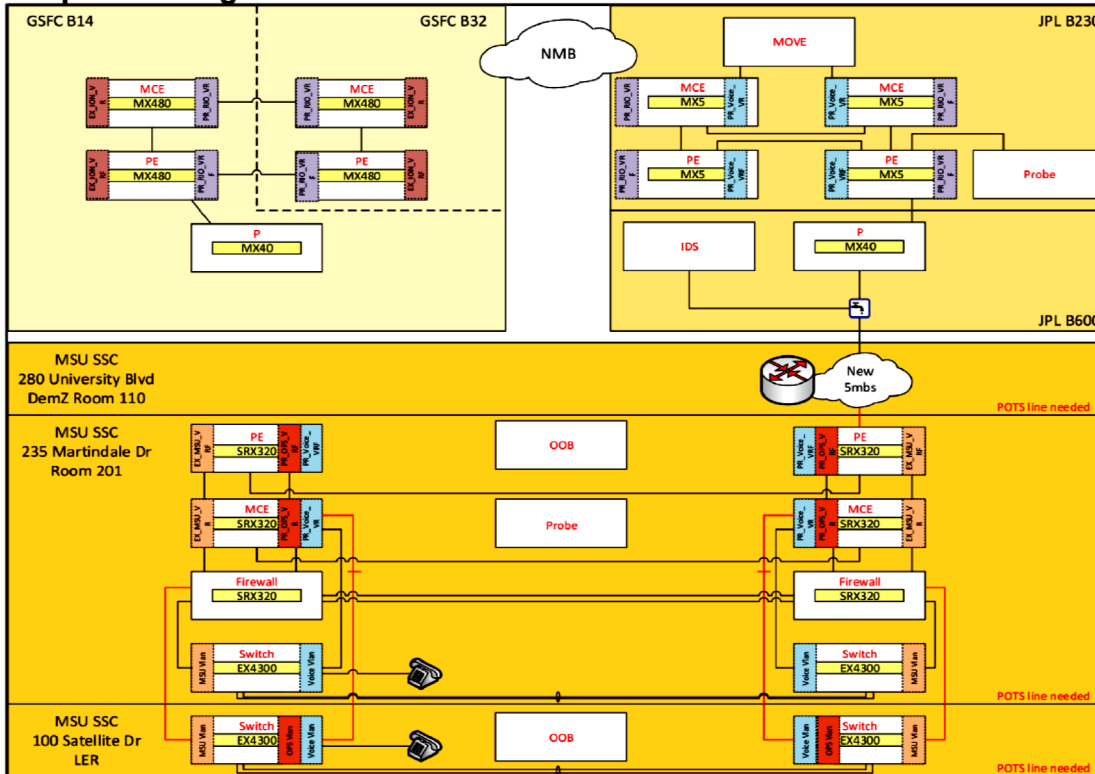


High Security IT and Network Connection Required

- Independent Network
- Architecture Designed with JPL and CSO
- Behind NASA Firewall
- Designed by NASA CSO
- Direct Connection to the NASA IoNet



Proposed Design



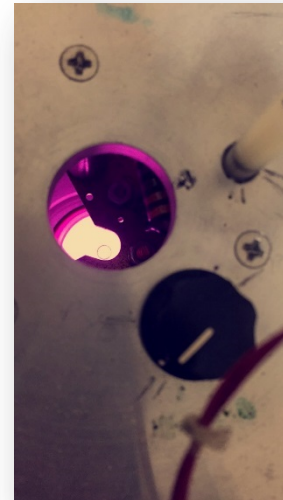
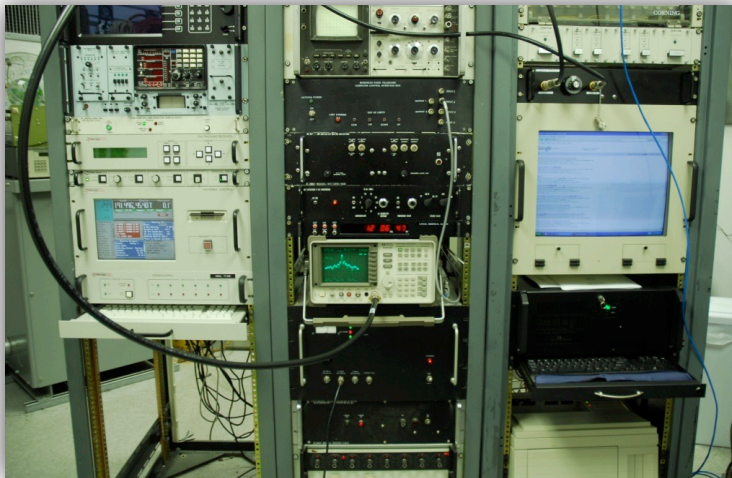


Becoming a DSN-Compatible Station



Status of Project

- System Architecture Design Complete
- DSN “Lite” Equipment in Fabrication at MSU and JPL
- Hydrogen MASER in Operation (on semi-permanent Loan from MIT)
- X-Band Feed Designed and in Fabrication
- Cryogenic LNA Under Test
- Antenna Control Systems Tested



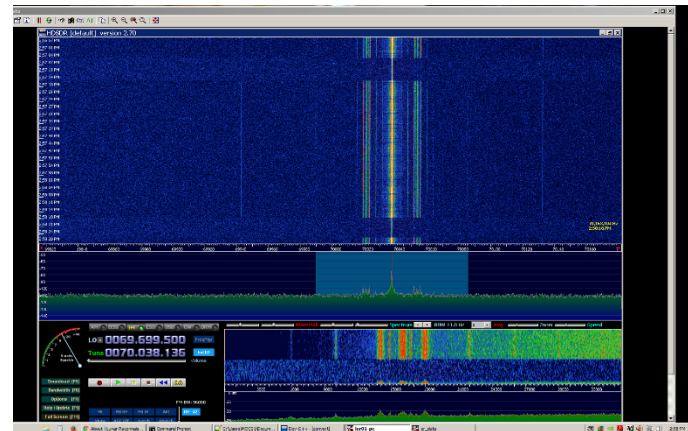
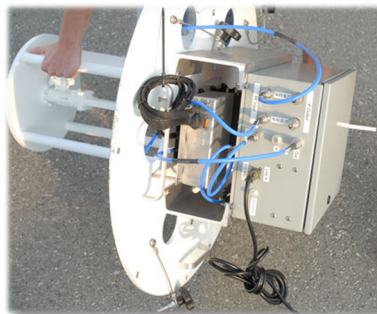
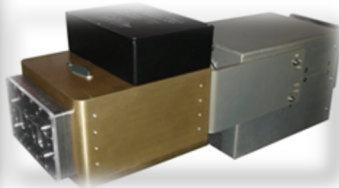
Duration (s)	Detector reading	Measured Allan Deviation	REquired
1	4.50E-013	3.18198E-13	3.30E-13
10	6.00E-014	4.24264E-14	1.00E-13
100	2.00E-014	1.41421E-14	3.30E-14

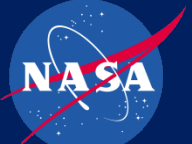


Expected 21m Performance

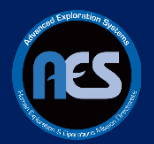


Performance Measure	Current Values	Post-Upgraded Targets
X-Band Frequency Range	7.0 – 7.8 GHz	7.0 – 8.5 GHz
LNA Temperature	70 K	< 20 K
System Temperature T_{sys}	215 K	<100 K
Antenna Gain	62.0 dBi (@7.7 GHz)	62.7 dBi (@8.4 GHz)
System Noise Spectral Density	-175 dBm/Hz	<-178 dBm/Hz
G/T at 5° Elevation	37.5 dBi/K	40.4 dBi/K
Time Standard	GPS (40-ns)	Cesium (2ns/day)
SLE Compliant	No	Yes
CCSDS Capable	No	Yes





DSS-17



Goldstone, California



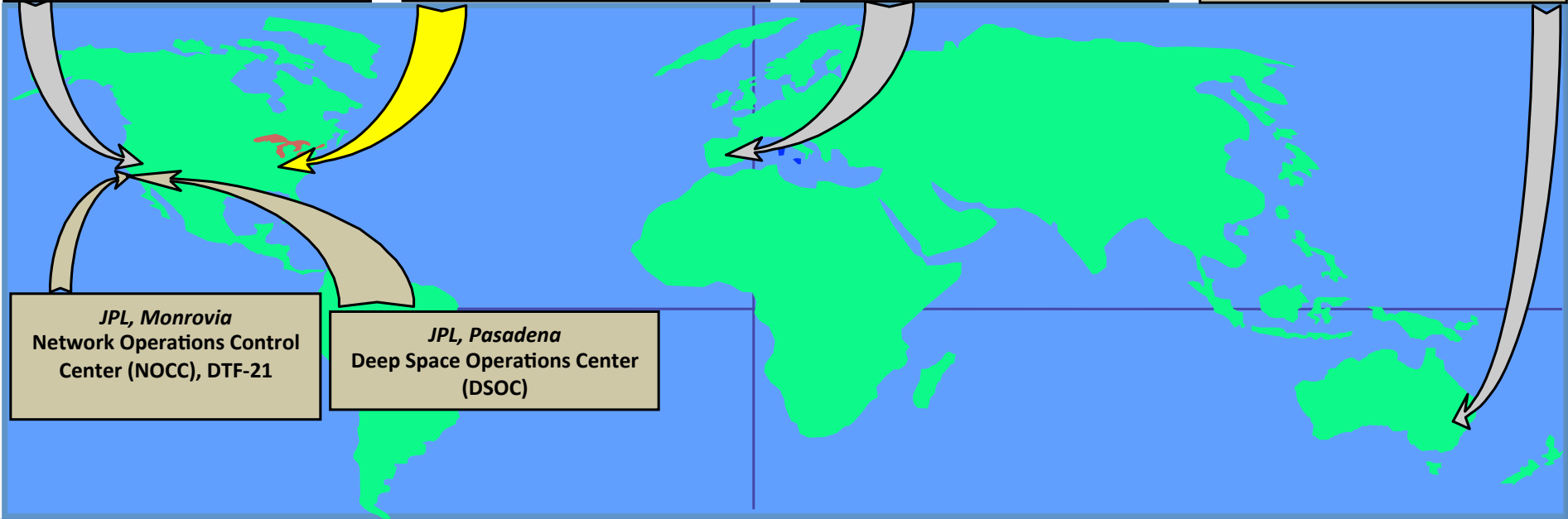
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Canberra, Australia





DSS-17 Next Steps



Critical Milestones

Δ SRR	IoNet Connection	Downlink Demo	Uplink Demo	Ranging Demo	ORR	Operational	Mission Ops	Mission Duration	Project Closure
01/15/2016	06/30/2017	12/15/2017	01/15/2018	03/15/2018	06/30/2018	10/15/2018	10/15/2018	EM-1 CubeSats Duration	EM-1 CubeSats Closure





Becoming a DSN Compatible Station



Provide Support for NASA EM-1 CubeSats

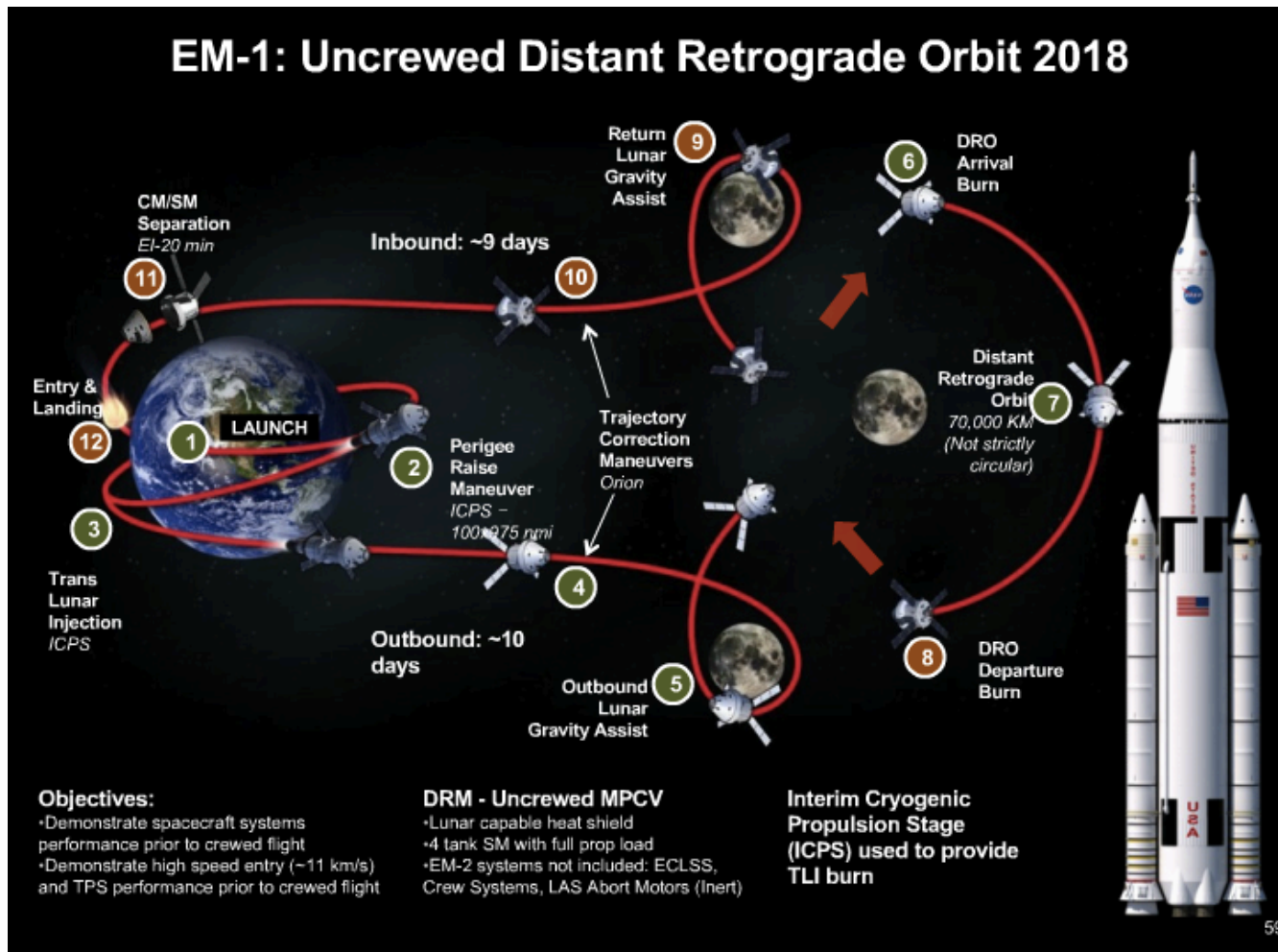
Lunar IceCube

NEA Scout

Lunar Flashlight

Biosentinel

LunaH Map





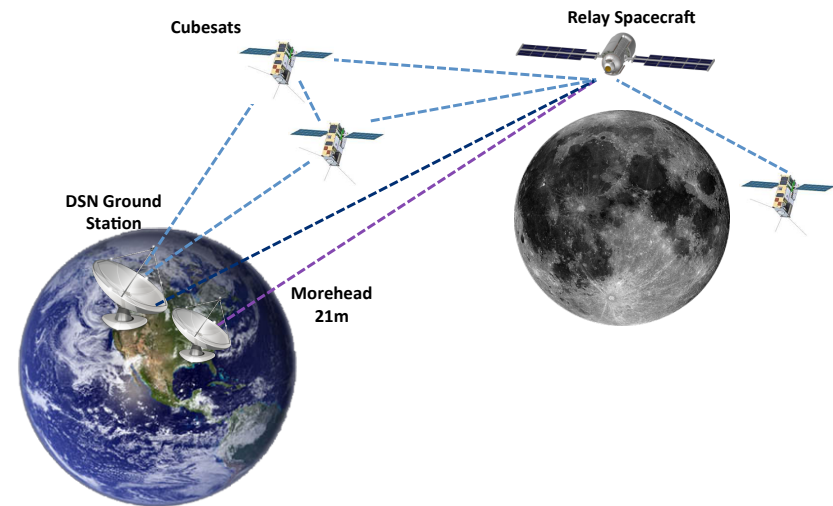
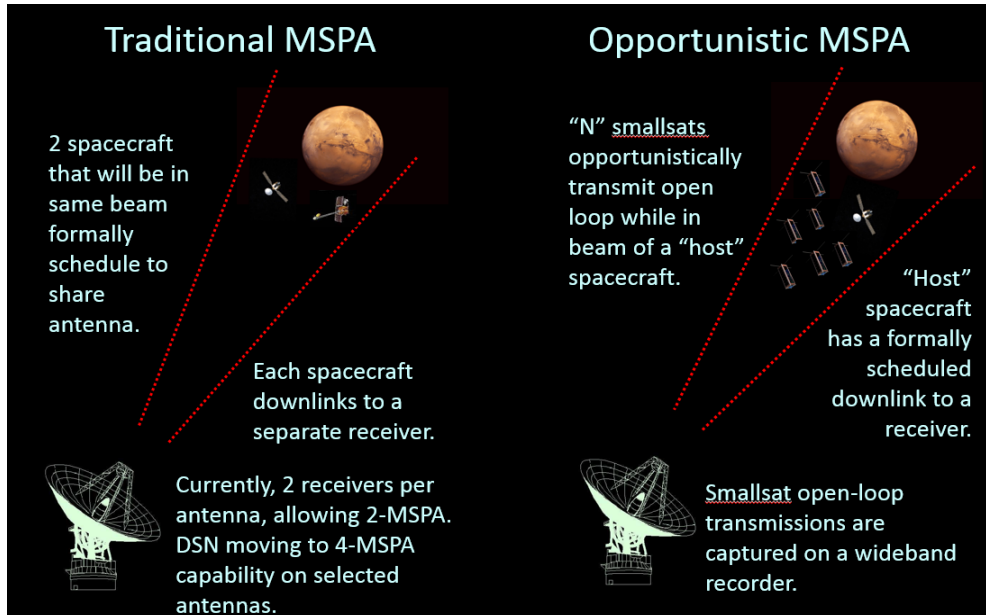
Future of DSS-17



Possible Extensions and Adaptations

- Multiple Spacecraft per Aperture (MSPA) and Opportunistic MSPA
- Delay/Disruption Tolerant Networking

Support for Space Networking





Questions?

