

**SpaceTReX**

# Integrated X-band Communication Antenna and Solar Array for Interplanetary CubeSats

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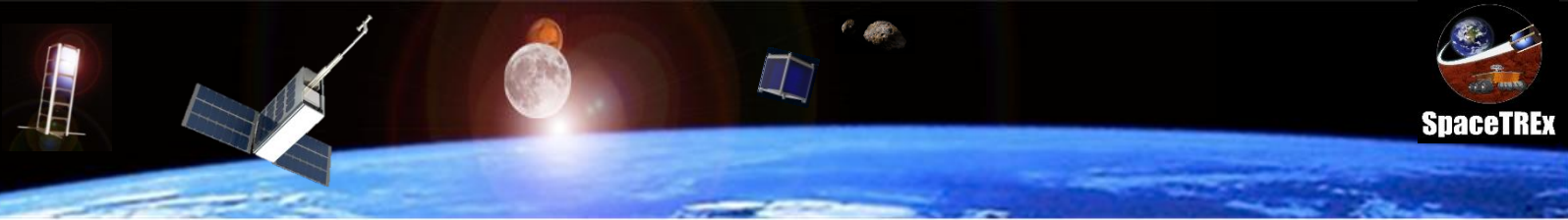
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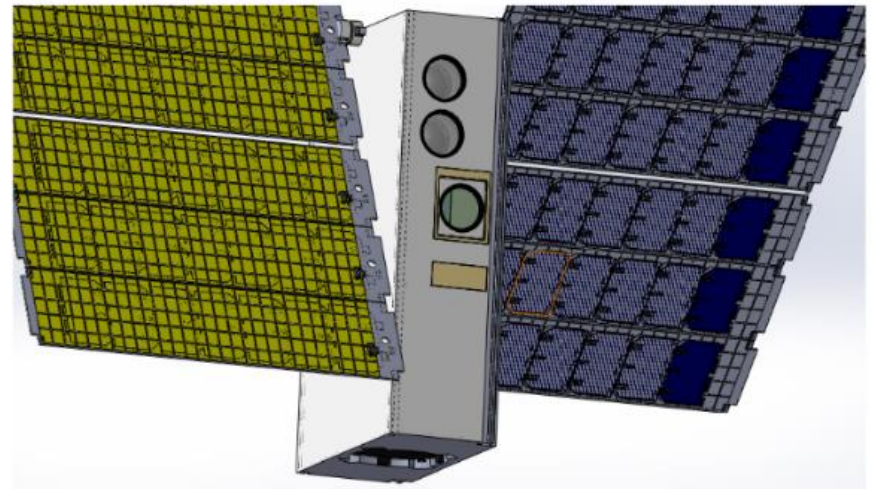


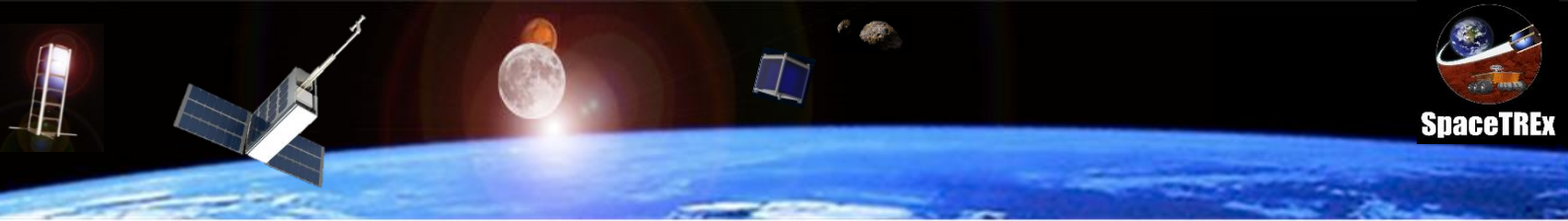
# Hybrid solar array and reflectarray

- HaWK Solar Arrays
  - Dual Gimbals
    - Reaction wheel solar desaturation
    - Variable modes
- X-Band Reflect Array
  - Hybrid on back of HaWK array
- Similar concept – ISARA (Ka-band)

ISARA REFLECT ARRAY

HAWK GIMBALLED SOLAR ARRAY





## Pros and Cons

### Pros

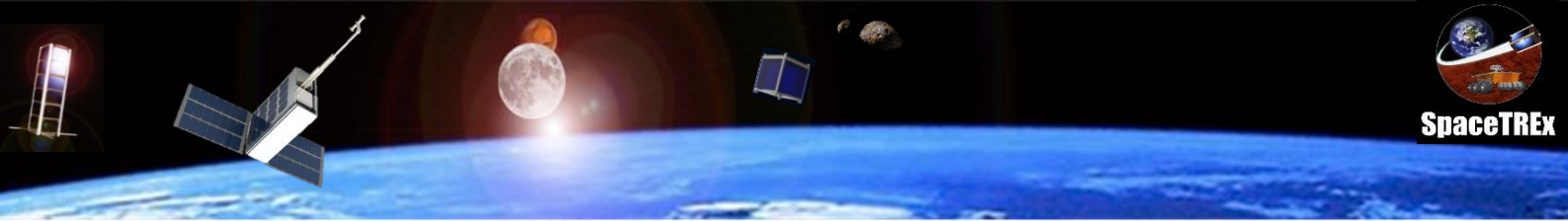
- Reduced mass and after deployment volume
- Reduced number of deployments
- Reduced effective surface area
- Simultaneous operation of solar array and antenna
- Redundancy
- Reduced Comms link limitations during tumbling
- Improvement in antenna gain\*

\* – assuming both arrays can be synchronized

### Cons

- Solar array and Reflectarray integration process complexity
- Two feeds required
- Complexity of synchronizing two Reflectarray signals (and feeds)

❖ Longer communication link time with CubeSat



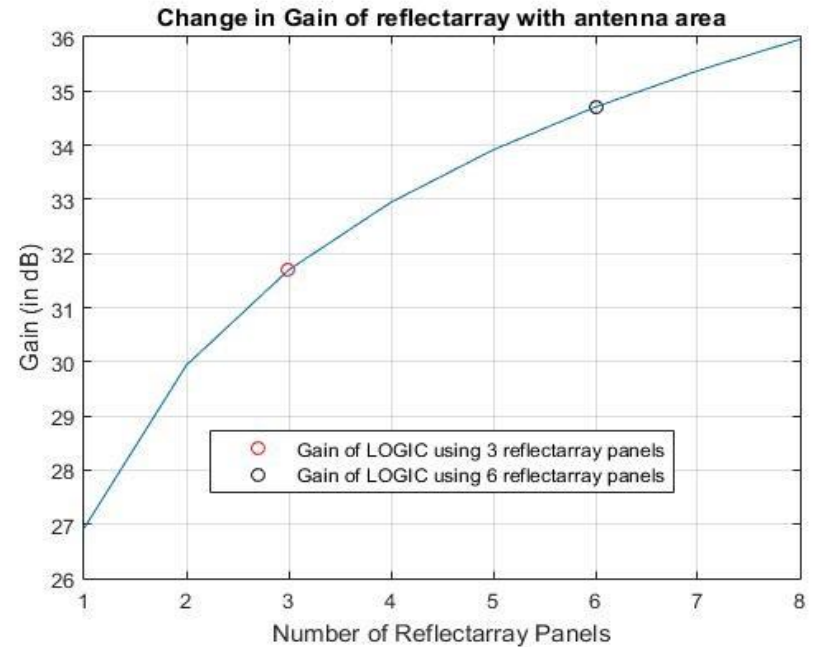
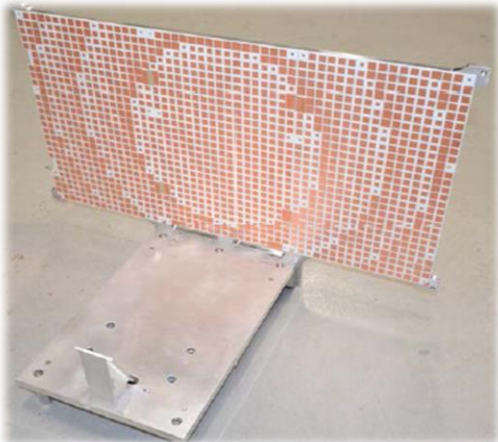
# Properties

Specification	X-band reflectarray
Mass	0.5 kg
Gain	31.69 dB (single array) 34.7 dB (double array)

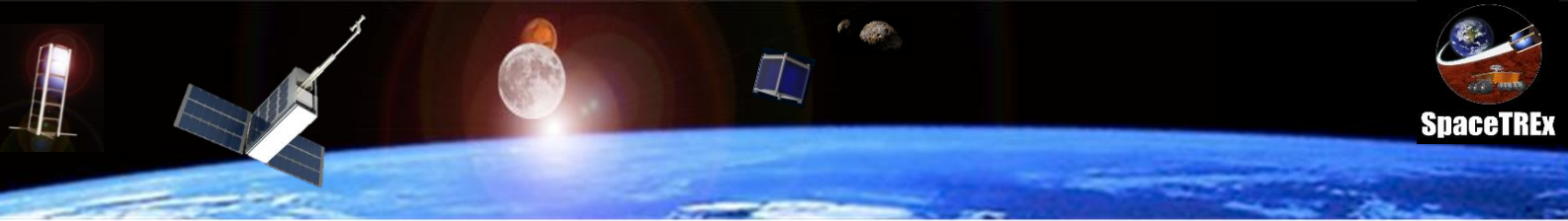
Specification	MMA E-HaWK (3 panels/ wing)
Mass	0.85 kg
Power	44 W OAP

- 1 – estimated mass of only electronics (no chassis)
- 2 – assuming both arrays can be synchronized

High gain X-band Reflectarray  
NASA JPL (MarCO)

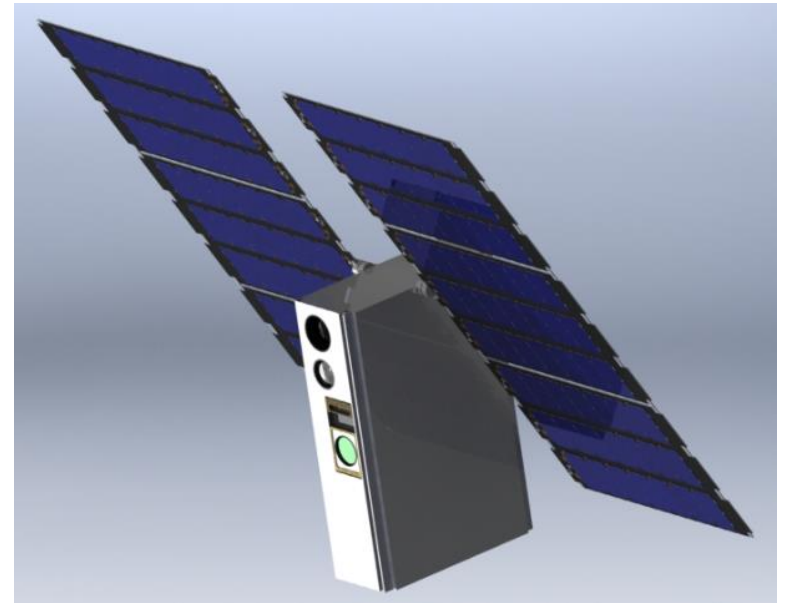


❖ Total area of 12U x 2U shared by Reflectarray and solar panels



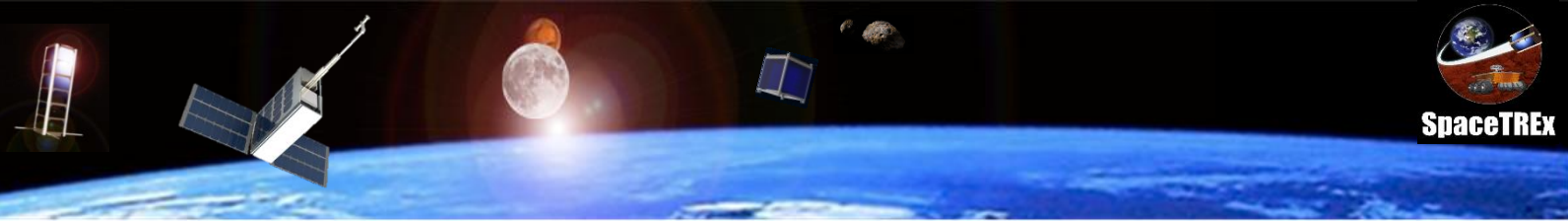
# Devils LOGIC: Low Orbit Geothermal Imaging CubeSat

- 6U CubeSat
- Hosted payload science mission to Phobos
  - Nominal Launch: Mars 2020
- 2 year mission life
  - Potential for extended mission (3 years)
- Thermal and visible camera payload
- Impulsive maneuver for Mars capture
  - 15+ flyby's achievable within 7 months
- 7 month science mission
  - Minimum of 5 flyby's science mission



**LOGIC Deployed View**

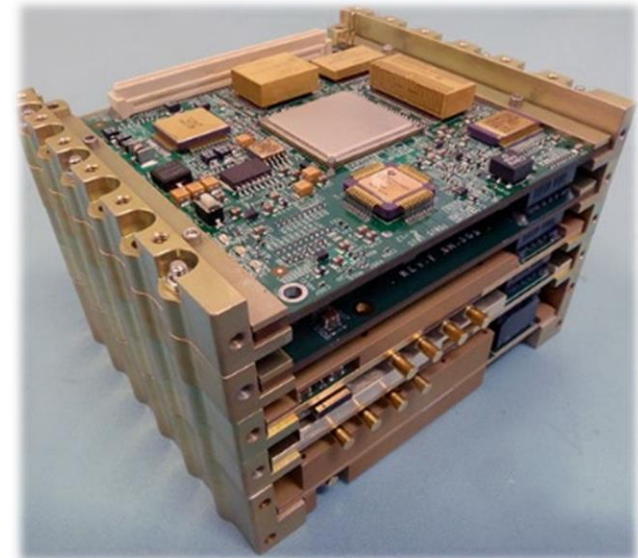




# LOGIC - Communications Subsystem

- DSN compatibility
- EIRP - 35 dB
- UHF relay (investigation in progress)

Iris V2 X band Transponder

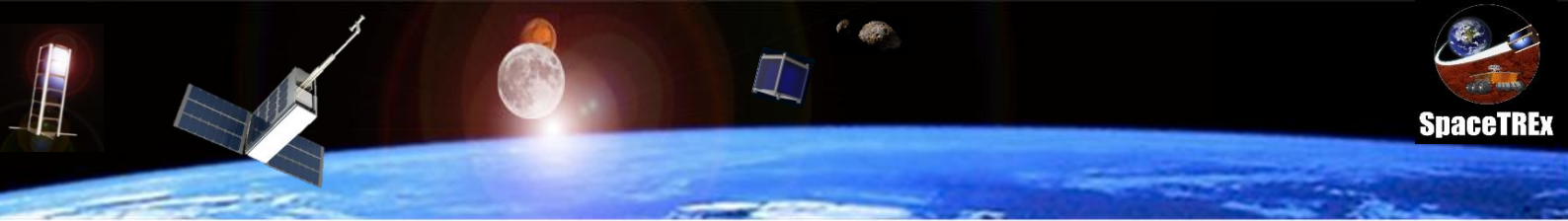


NASA JPL (MarCO)

Link budget to Earth (X band)

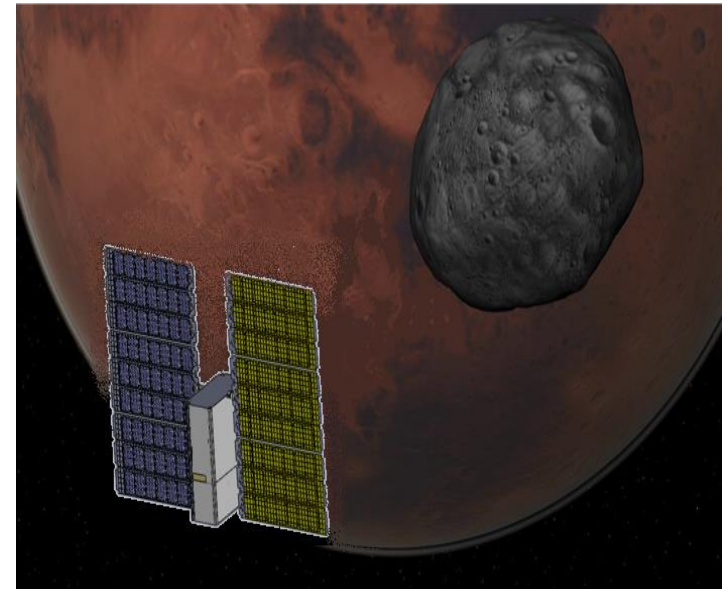
Scenario (Data rate)	Distance (million km)	Downlink
Worst	300	6.4 kbps
Nominal	250	9.3 kbps
Best	200	14.5 kbps

- ❖ Integrated (solar array and) X-band Reflectarray with Iris v2 for Interplanetary missions



## Summary

- Reduced risk during complex maneuvers (aerobraking)
- Simultaneous solar array and comms operation -> Reduced comms blackout
- CubeSat Electra (UHF) payload -> proximity comms to existing Mars orbiter assets



LOGIC spacecraft with Phobos and Mars