



National Aeronautics and Space Administration

# The *Plume Chaser* Mission:

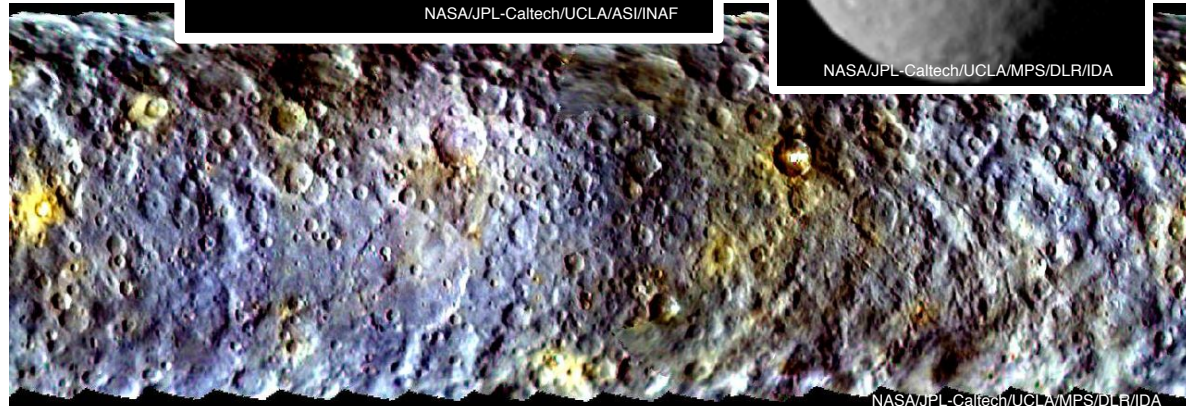
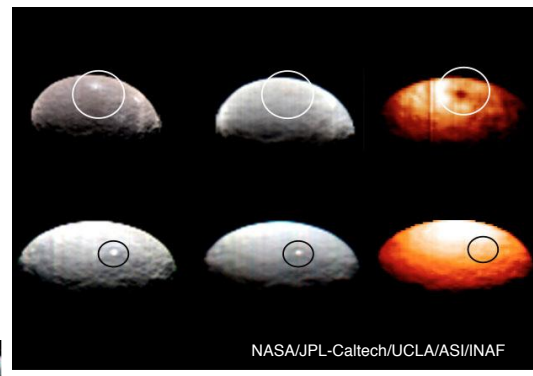
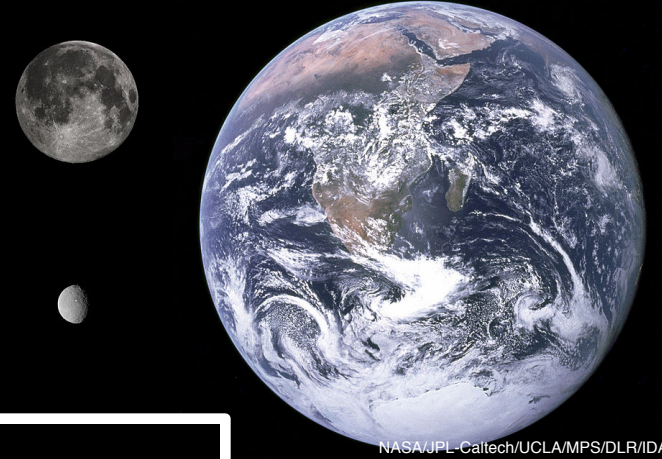
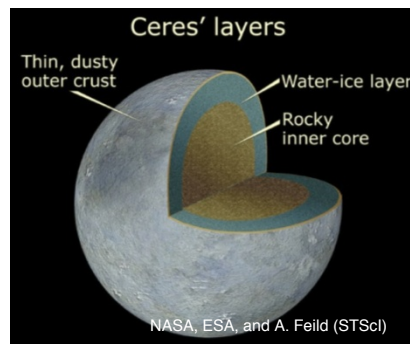
## Two-Spacecraft Search for Organics on the Dwarf Planet Ceres

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# Ceres

- **Aphelion: 2.9773 AU**
- **Perihelion: 2.5577 AU**
- **Orbital period: 4.60 yr**
- **Avg. orbital speed: 17.905 km/s**
- **Mean radius:  $476.2 \pm 1.7$  km**
- **Mean density:  $2.077 \pm 0.036$  g/cm<sup>3</sup>**
- **Surface gravity: 0.28 m/s<sup>2</sup>**
- **Escape velocity: 0.51 km/s**
- **Mean surf. Temp.: 168 K**
- **Max surf. Temp.: 235 K**



# Why is Ceres so interesting?

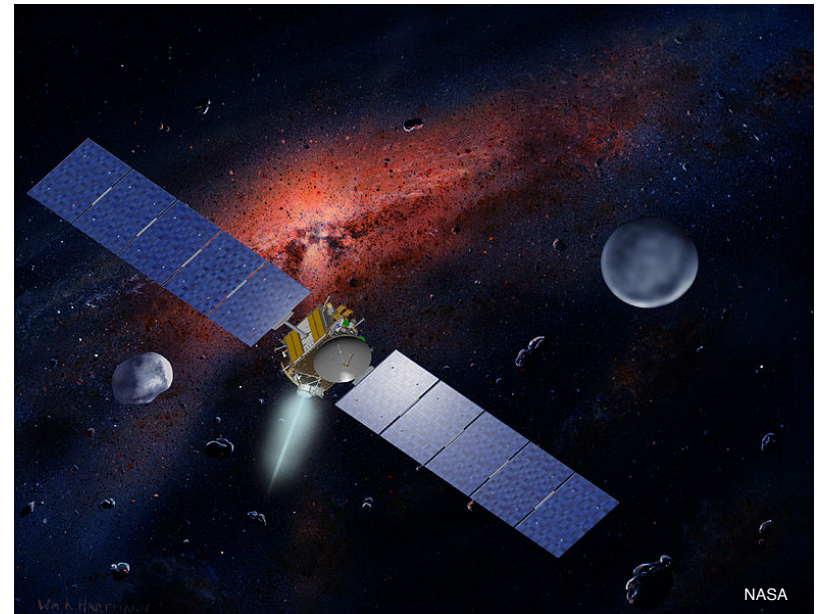
- Differentiated structure with a anhydrous silicate core with water-rich mantle
- The relatively rare CM chondrites are thought to originate from here
- Spared from the effects of the Late Heavy Bombardment (LHB)
- HST detected up to 8 features releasing  $10^{26}$  molecules ( $\sim 6$  kg) of material / s
- HSO found water vapor in the vicinity of 2 mid-latitude dark regions
- Thermal models suggest liquid water below the surface
- Could there be cryovolcanism and liquid water bodies under the frozen crust?



Figure adapted from an illustration by Chris Butler/SPL

# Why another mission to Ceres?

- **Dawn arrives close to aphelion, while plume activity was observed at perihelion**
- **Dawn instruments are not capable of measuring needed critical isotopic composition of volatiles**
- **Dawn has no impactor to enable**
  - **the release subsurface volatiles**
  - **imaging before and after an impact event**



➔ **Plume Chaser has a different, but complementary mission concept to Dawn**



# Science Objectives

## Primary Science Objectives

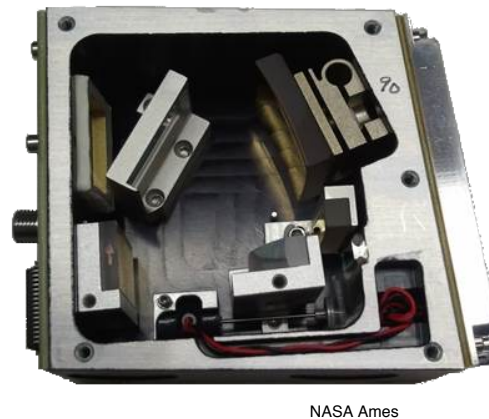
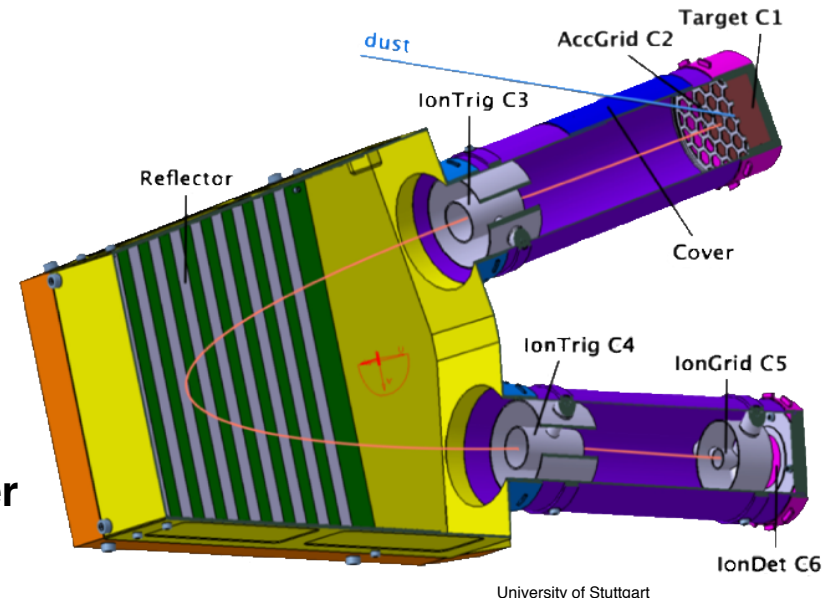
- A.1 Search for evidence of organics in the Ceres water vapor/plume**
- A.2 Determine the column water content and location of highest density**
- A.3 Determine the elemental composition (C, H, O, N) of any carbon rich plume particles**
- A.4 Characterize the particles in the plume for dissolved salts**

## Secondary Science Objectives

- B.1 Characterize the silicate dust in the plume and relate it to other solar system objects, such as meteorites and comets**
- B.2 Characterize the mechanical, chemical and geological properties of the crust by observing the impact site before and after impact**
- B.3 Look for surface compositional changes on the surface associated with venting; characterize composition differences involving water or possible organics**

# Instrumentation

- ENIJA Mass Spectrometer**  
 TRL 4  
 Particle size: 10 nm – 100  $\mu\text{m}$   
 Particle mass: 1 – 2000 amu
- NASA ARC/Draper Lab. NIR spectrometer**  
 TRL 4  
 Range: 1.6 – 3.4  $\mu\text{m}$   
 Resolution: 0.15  $\mu\text{m}$
- GomSpace NanoCam C1U camera**  
 TRL 9  
 Spectrum: 400 – 1000 nm  
 FOV: 9.22 deg  
 Resolution: < 80 m/pixel from 650 km



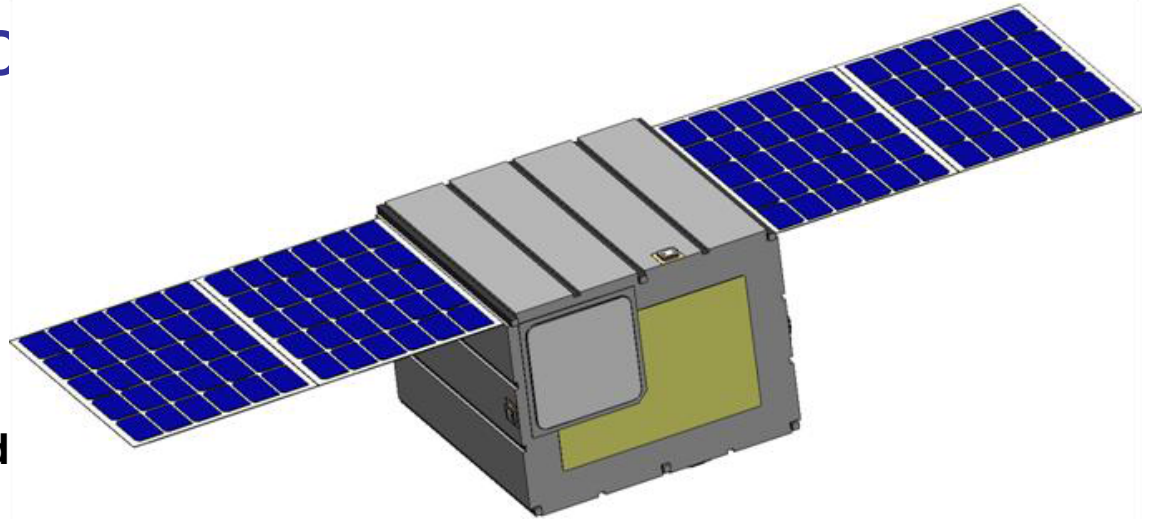


# Mission Concept: *Sniff, Wait, and Impact*

- **Sniff**
  - Venting from the surface of Ceres is common
  - Permanent exosphere
  - Densest parts will be localized and studied
- **Wait**
  - Venting is not common on Ceres, but likely at perihelion
  - S/C must remain in orbit until a plume event occurs and can subsequently be maneuvered through it
- **Impact**
  - Venting is rare on Ceres
  - Sporadic occurrences under extremely favorable conditions
  - Trace exosphere might be beyond detection limits
  - To guarantee scientific return, impactor will create a plume to fly through

# The Two Spacec

- **Same design for both S/C**  
**Size: 40x40x30 cm<sup>3</sup>**  
**Dry mass: 40 kg**  
**Wet mass: <190 kg**  
**Fit within ESPA Grand frame**
- **3D printed Ti tank/structure**
- **MMA Design solar array**  
**200 W EOL**
- **Propulsion**  
**CubeSat Ambipolar Thruster**  
**(CAT)**  
**ISP: 1200**  
**Size: 1U**  
**Propellant: Iodine**



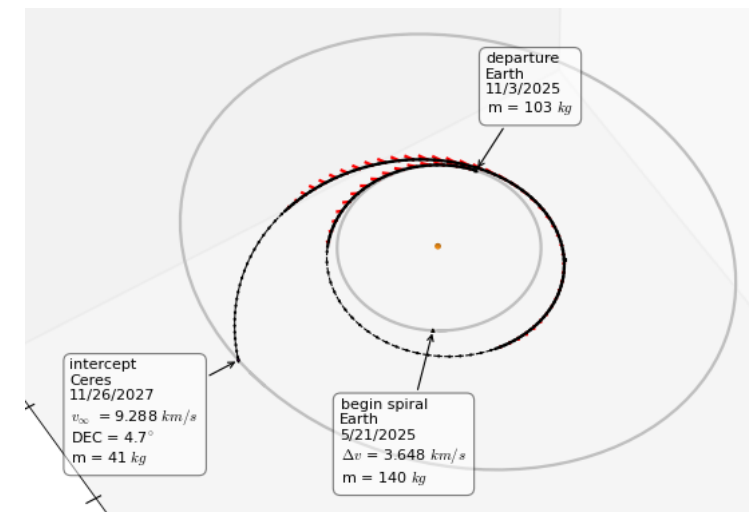
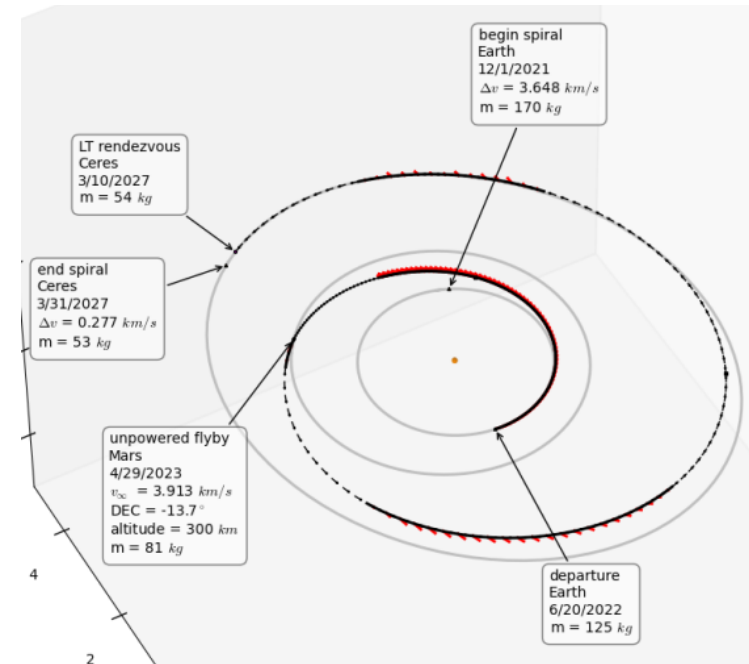
University of Michigan





# Trajectory Design

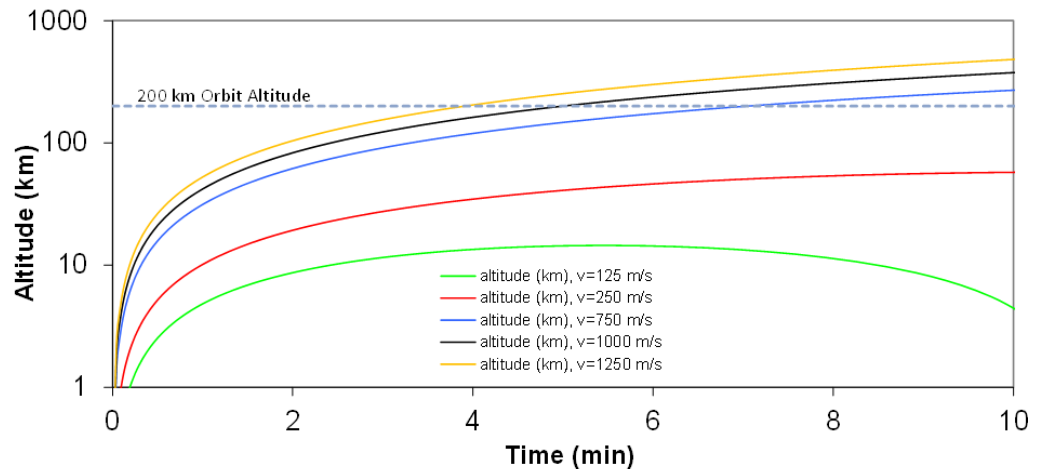
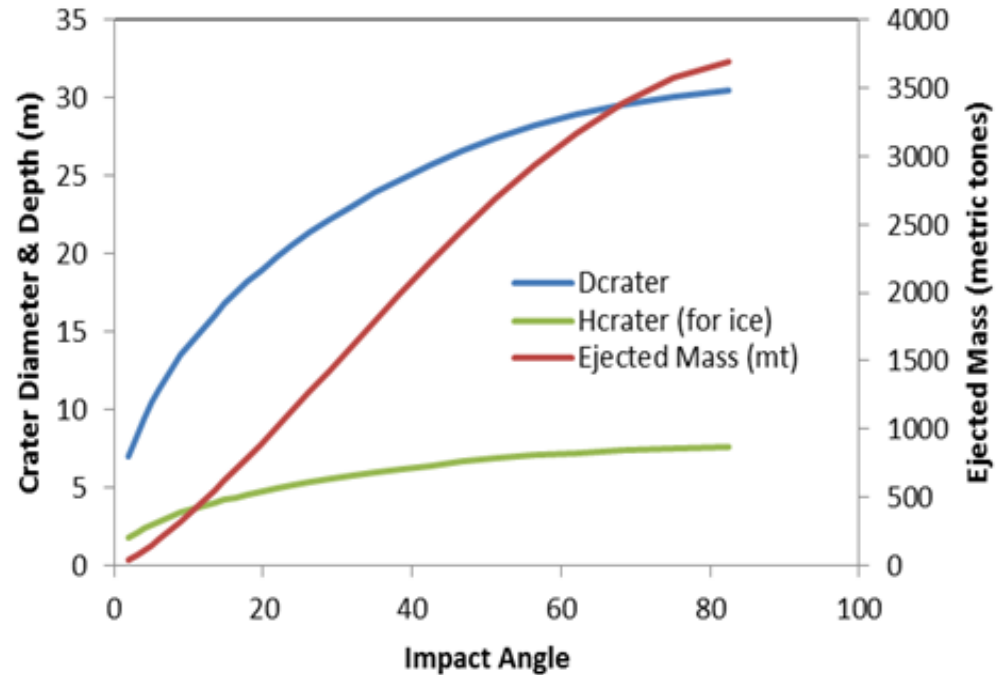
- Starting from GTO
- Plume Chaser
  - Dry mass: 43 kg
  - Propellant: 120 kg
  - Overall duration: 4.8 yrs
  - Delta-V: 14.4 km/s
  - 1 year science orbit before decommission
- Plume Maker
  - Impact mass: 41 kg
  - Propellant: 99 kg
  - Overall duration: 2.1 yrs
  - Delta-V: 14.5 km/s
  - Impact speed: 9.3 km/s





# Impacting

- **To create a crater**
  - **Speed: 9.3 km/s**
  - **Mass: 40.9 kg**
  - **Energy: 1764.5 MJ**
  - **Crater**
    - **> 10 m wide**
    - **> 3 m deep**
- **To create a plume**
  - **Reach above 200 km**
  - **With particle velocity 750-2000 m/s**





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# Conclusions



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