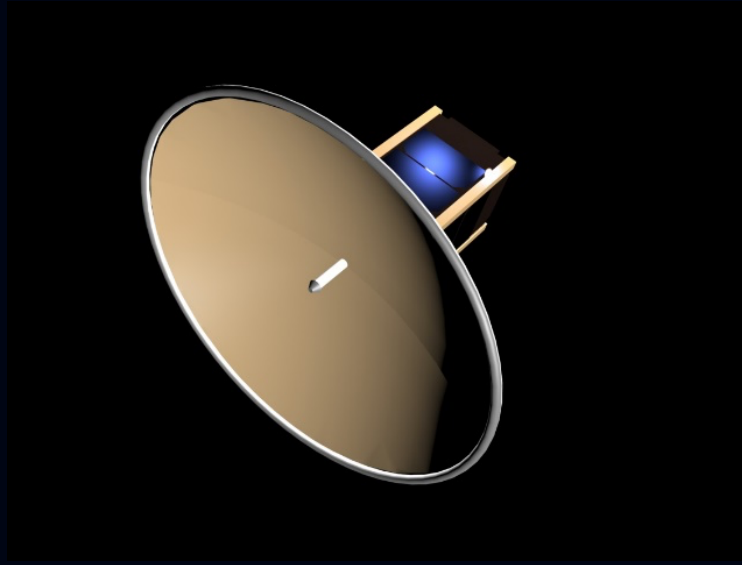
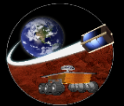
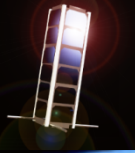


Development of an Inflatable Antenna Prototype for Interplanetary CubeSats



M. Ravichandran, A. Chandra
J. Thangavelautham
SpaceTReX Laboratory
School of Earth and Space Exploration
Arizona State University

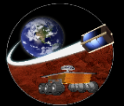
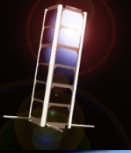
A. Babuscia
Jet Propulsion Laboratory
California Institute of Technology



Motivation

Communication Capability

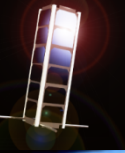
- At LEO or higher
- For Interplanetary missions
- Antenna requirements



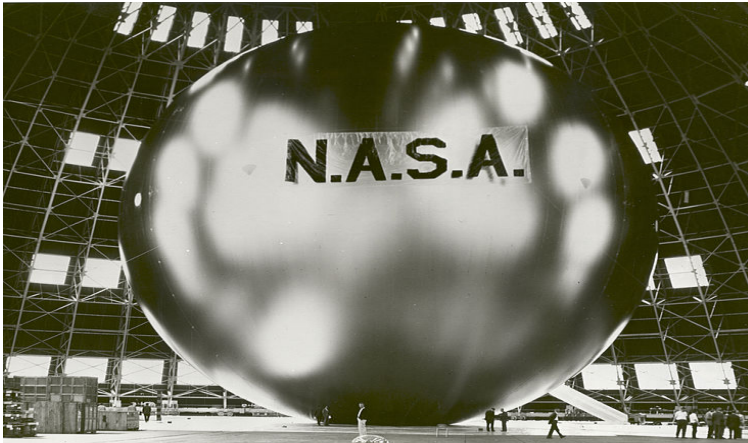
Motivation

Inflatable offer a promising technology pathway for interplanetary CubeSat communication and tracking

- Very high deployed volume
- Very low mass
- High-packing efficiency
- Quick deployment
- Relative simplicity



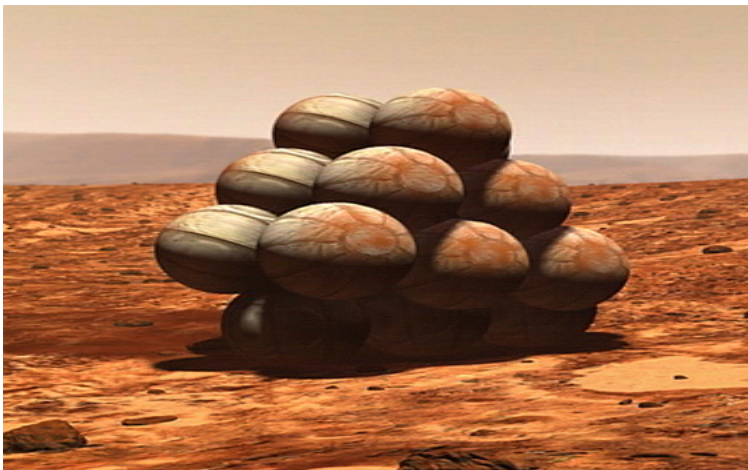
Inflatables



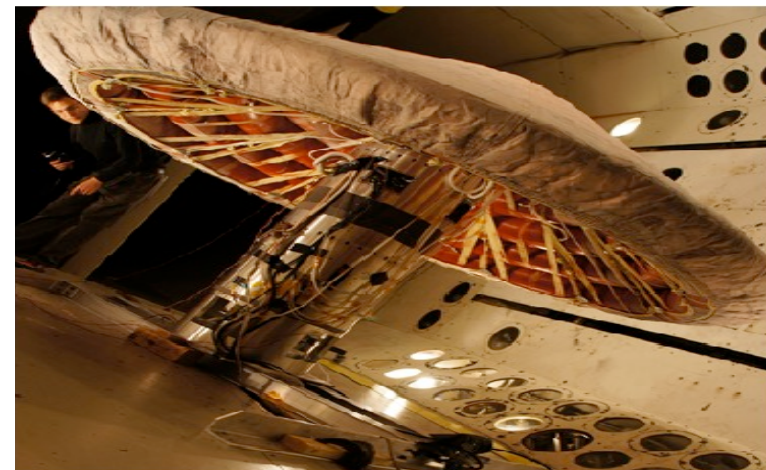
ECHO I MISSION



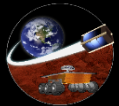
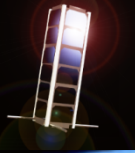
INFLATABLE HABITAT



MARS EXPLORER ROVER



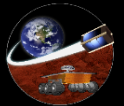
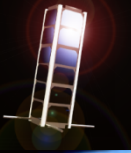
INFLATABLE DECELERATOR



Challenges

Inflation System

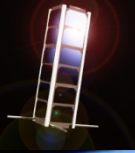
- Optimum Inflation Pressure
- Passive Activation/Inflation
- Rigidization & Shape Conformance
- Micrometeorite Attack



Objective

Research inflation techniques and develop an inflation system for CubeSat platform

- **Low mass, low cost system**
- **Passive activation**
- **Resistance to micrometeorite attacks**
- **Validate with physical models**



Inflation Systems

ECHO I & II Satellite

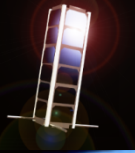
- **Sublimating Powder** [Clemmons, 1964, Talentino, 1966]

Mars Pathfinder

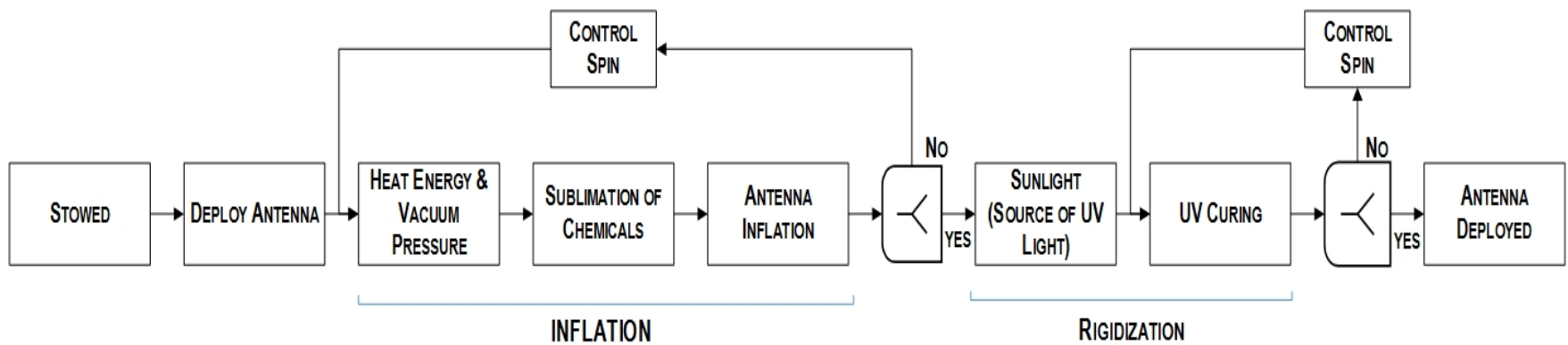
- **Gas Generator** [Cadogen et al., 2002]

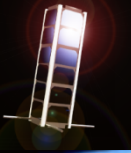
Inflatable Aerodynamic Decelerators

- **Gas Generator & Ram Air** [Cruz et al., 2006, Hughes et al., 2011]



Proposed System





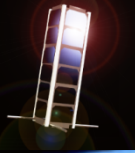
Sublimating Chemicals

Sublimation

- Phase transition from solid state to gaseous state without going through the liquid state

Features

- Chemicals: Powder or Crystalline form, less mass and easy packing
- Sublimate to release large volume of gas



Selection of Chemicals

Operating Temperature of Antenna [Babuscia et al., 2013]

Range:

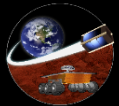
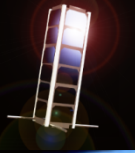
369K – 377K

Average:

373K

Sublimation Enthalpy:

- Depends on operating temperature
- Helps Passive Activation



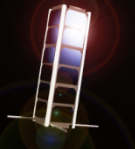
Selection of Chemicals

Vapor Pressure

Pressure exerted by the gas on the inflatable surface

- **High Pressure: Rapid inflation and rupture**
- **Low Pressure: Insufficient internal pressure for inflation**

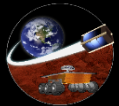
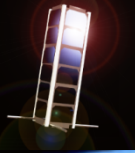
[Clemmons, 1964, Talentino, 1966]



Candidate Chemicals

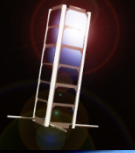
From an extensive list of sublimating chemicals the following were shortlisted [Jones, A.H, 1960]

SUBLIMATING POWDERS	CHEMICAL FORMULA	MOLECULAR MASS (g)	MASS REQUIRED (mg)
Benzoic acid	C ₇ H ₆ O ₂	122.12	0.86
Butyramide	C ₄ H ₉ NO	87.12	0.62
Diformylhydrazine	C ₂ H ₄ N ₂ O ₂	88.07	0.62
Hexachloroethane	C ₂ Cl ₆	236.74	1.68
O-methoxybenzoic acid	C ₈ H ₈ O ₃	152.15	1.08
Oxalic acid (anhydrous)	H ₂ C ₂ O ₄	90.03	0.64
Pyrene	C ₁₆ H ₁₀	202.25	1.43
Salicylic acid	C ₇ H ₆ O ₃	138.12	0.98
Urea	CH ₄ N ₂ O	60.06	0.43



Rigidization Techniques

- **UV setting resins** [Keller et al., 1964]
- **Thermosetting resins** [Cadogen et al., 2001]
- **Glass transition resins** [Lichodziejewski et al., 2005]
- **Gas cured resins** [Bernasconi et al., 1990]
- **Stretched metal laminates** [James et al., 1964, Bahiman et al., 1965]
- **Evaporation hardened rigidizing foams** [Schnell et al., 2002]



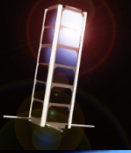
Rigidization Techniques

Proposed Technique

- UV Curable Ink sprayed over the antenna surface
- Curing through rapid polymerization

Advantages

- Rapid Curing (<20s) [Chattopadhyay et al., 2005]
- Flexible coating post curing [Schwalm et al., 1997]
- No hazardous vapors [Chattopadhyay et al., 2005]
- Low UV Energy requirement [Chattopadhyay et al., 2005]
- Clear coating possible [Seubert et al., 2004]
- Strong mechanical property [Chattopadhyay et al., 2005]

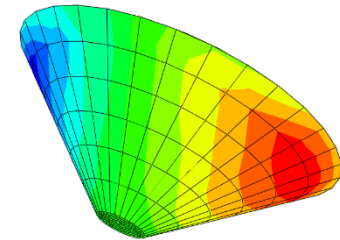
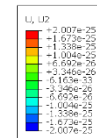


Analysis

Verified structural integrity of 2 prototypes

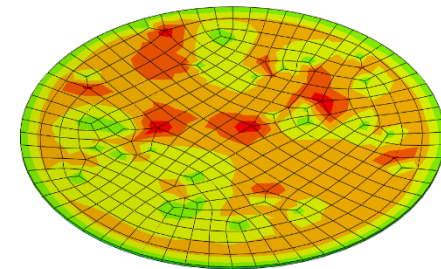
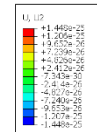
Forces applied

- Centrifugal Force
- Radiation Force
- Internal Vapor Pressure



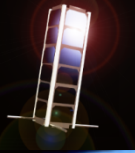
CDB: Spherical_Antenna.odb Abaqus/Standard Student Edition 6.14-2 Mon Apr 06 23:37:19 US Mountain Standard Time 2015
 Step: Step-1
 Increment: 1 Step Time = 2.2200E-16
 Primary Var: U, U2
 Deformed Var: U, U2 Deformation Scale Factor: +1.000e+09

Prototype-I



Y
Z X
CDB: Antenna_Shell.odb Abaqus/Standard Student Edit on 6.14-2 Mon Apr 06 22:57:21 US Mountain Standard Time 2015
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 Increment: 1 Step Time = 2.2200E-16
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 Deformed Var: U, U2 Deformation Scale Factor: +1.000e+09

Prototype-II

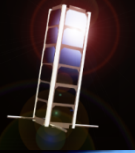


Results

Max. Deformation of Mylar at Break

$$5.226 \times 10^{-25} \text{ mm}$$

SUBLIMATING CHEMICAL	VAPOR PRESSURE (Pa)	MAX DEFORMATION (mm)	
		Prototype-I	Prototype-II
Benzoic Acid	9.33E-02	1.19E-26	1.01E-25
Butyramide	5.21E-01	2.007E-25	5.66E-25
Diformylhydrazine	6.67E+01	3.01E-27	7.24E-27
Hexachloroethane	5.33E+01	1.69E-27	5.69E-27
Methoxybenzoic acid	2.00E-01	1.018E-26	2.17E-25
Oxalic Acid	1.33E-01	5.04E-26	1.45E-25
Pyrene	6.00E-04	3.28E-28	6.51E-28
Salicylic Acid	1.09E-02	4.93E-27	1.19E-26
Urea	1.60E-03	3.49E-28	1.74E-27



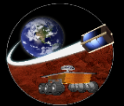
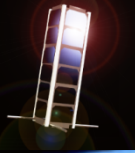
Discussion

Structural Feasibility:

- No structural failure (except Butyramide)

UV Curable Rigidity:

- Curing increases mechanical strength as temperature [Chattopadhyay et al., 2005]



Future Work

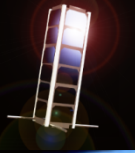
Experimental Validation in Vacuum Chamber

Sublimating Chemicals:

- Validation Pressure & Temperature points for Sublimation
- Time taken for Inflating the model

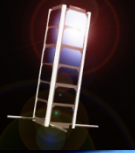
UV Curable Ink:

- Observation of chemical reactions with Mylar sheet
- Rigidity achievable measured with FTIR method
- Transparency of coat to electromagnetic radiations



References

1. Jones, A. H. "Sublimation Pressure Data for Organic Compounds." *Journal of Chemical and Engineering Data* 5.2 (1960): 196-200.
2. Clemmons, Dewey L. *The Echo I inflation system*. National Aeronautics and Space Administration, 1964.
3. Talentino, J. P. "Development of the fabrication and packaging techniques for the ECHO II satellite." (1966).
4. Cadogan, D., C. Sandy, and M. Grahne. "Development and evaluation of the Mars Pathfinder inflatable airbag landing system." *Acta Astronautica* 50.10 (2002): 633-640.
5. Calomino, Anthony M. "Hypersonic Inflatable Aerodynamic Decelerator (HIAD) Technology Development Overview." (2011).
6. Cruz, Juan R., and J. Stephen Lingard. "Aerodynamic decelerators for planetary exploration: past, present, and future." *AIAA Paper* 6792 (2006): 21-24.
7. Stein, James, and Charles Sandy. "Recent developments in inflatable airbag impact attenuation systems for Mars exploration." *AIAA* 1900 (2003): 7-10.
8. NASA Factsheet, HIAD, <http://www.nasa.gov/HIAD>
9. Babuscia, Alessandra, et al. "CommCube 1 and 2: A CubeSat series of missions to enhance communication capabilities for CubeSat." Aerospace Conference, 2013 IEEE. IEEE, 2013.
10. Mylar 814 Datasheet, Dupont Teijin Films, <http://www.dupontteijinfilms.com/FilmEnterprise/Datasheet.asp?ID=301&Version=US>
11. Klinkrad, H., and B. Fritsche. "Orbit and attitude perturbations due to aerodynamics and radiation pressure." *ESA Workshop on Space Weather, ESTEC, Noordwijk, Netherlands*. 1998.
12. L.B. Keller and S. Schwartz. Rigidization techniques for integrally woven composite constructions. *Technical Report ML-TDR-64-299, Hughes Aircraft Corp.*, 1964.



References

13. D.P. Cadogan and S.E. Scarborough. Rigidizable materials for use in Gossamer Space Inflatable Structures. In 42nd AIAA/ASME/ASCE/AHS/ASC Structures, Structural Dynamics, and Materials Conference, number AIAA-2001-1417, Seattle, WA, USA, April 16–19 2001. doi: 10.2514/6.2001-1417
14. D. Lichodziejewski, R. Cravey, and G. Hopkins. Inflatably deployed membrane waveguide array antenna for space. In 44th AIAA/ASME/ASCE/AHS Structures, Structural Dynamics, and Materials Conference, number AIAA-2003-1649, Norfolk, VA, USA, April 7–10 2003. doi: 10.2514/6.2003-1649.
15. D. Lichodziejewski, B. Derb`es, R. Reinert, K. Belvin, K. Slade, and T. Mann. Development and ground testing of a compactly stowed scalable inflatably deployed solar sail. In 45th AIAA/ASME/ASCE/AHS/ASC Structures, Structural Dynamics and Materials Conference, number AIAA-2004-1507, Palm Springs, CA, USA, April 19–22 2004. doi: 10.2514/6.2004-1507.
16. F. H. Redell, D. Lichodziejewski, J. Kleber, and G. Greschik. Testing of an inflation-deployed sub-tg rigidized support structure for a planar membrane waveguide antenna. In 46th AIAA Structures, Structural Dynamics and Materials Conference, number AIAA-2005-1880, Austin, Texas, April 18–21 2005. doi: 10.2514/6.2005-1880.
17. D. Lichodziejewski, B. Derb`es, and T. Mann. Vacuum deployment and testing of a 4-quadrant scalable inflatable rigidizable solar sail system. In 46th AIAA/ASME/ASCE/AHS/ASC Structures, Structural Dynamics and Materials Conference, number AIAA-2005-2122, Austin, TX, USA, April 18–21 2005. doi: 10.2514/6.2005-2122.
18. Schenk, Mark, et al. "Review of inflatable booms for deployable space structures: Packing and rigidization." *Journal of Spacecraft and Rockets* 51.3 (2014): 762-778.
19. Chattopadhyay, D. K., Siva Sankar Panda, and K. V. S. N. Raju. "Thermal and mechanical properties of epoxy acrylate/methacrylates UV cured coatings." *Progress in Organic Coatings* 54.1 (2005): 10-19.
20. R. Schwalm, L. Ha` Eussling, W. Reich, E. Beck, P. Enenkel, K.Menzel, *Prog. Org. Coat.* 32 (1997) 191
21. Seubert, C. M., and M. E. Nichols. "Alternative curing methods of UV curable automotive clearcoats." *Progress in organic coatings* 49.3 (2004): 218-224.