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## Opportunistic MSPA: A Low-Cost Downlink Alternative for Deep Space Smallsats

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# Topics

- Challenges to Deep Space Smallsats
- Responding to the Challenges: Opportunistic MSPA
- OMSPA Pros & Cons
- What's Next?
- Appendix: Performance Analysis
  - Data Rates
  - Maximum Smallsat-Host Separation Distances

# Smallsat Comm. Challenges (1/2)

## DSN Communications-Cost Perceptions

\$425M Discovery-Class Mission



\$5M Deep Space Cubesat



“Antenna” Fee = ~1% of Total Mission Cost w/o LV\*

“Antenna” Fee = ~110% of Total Mission Cost w/o LV\*

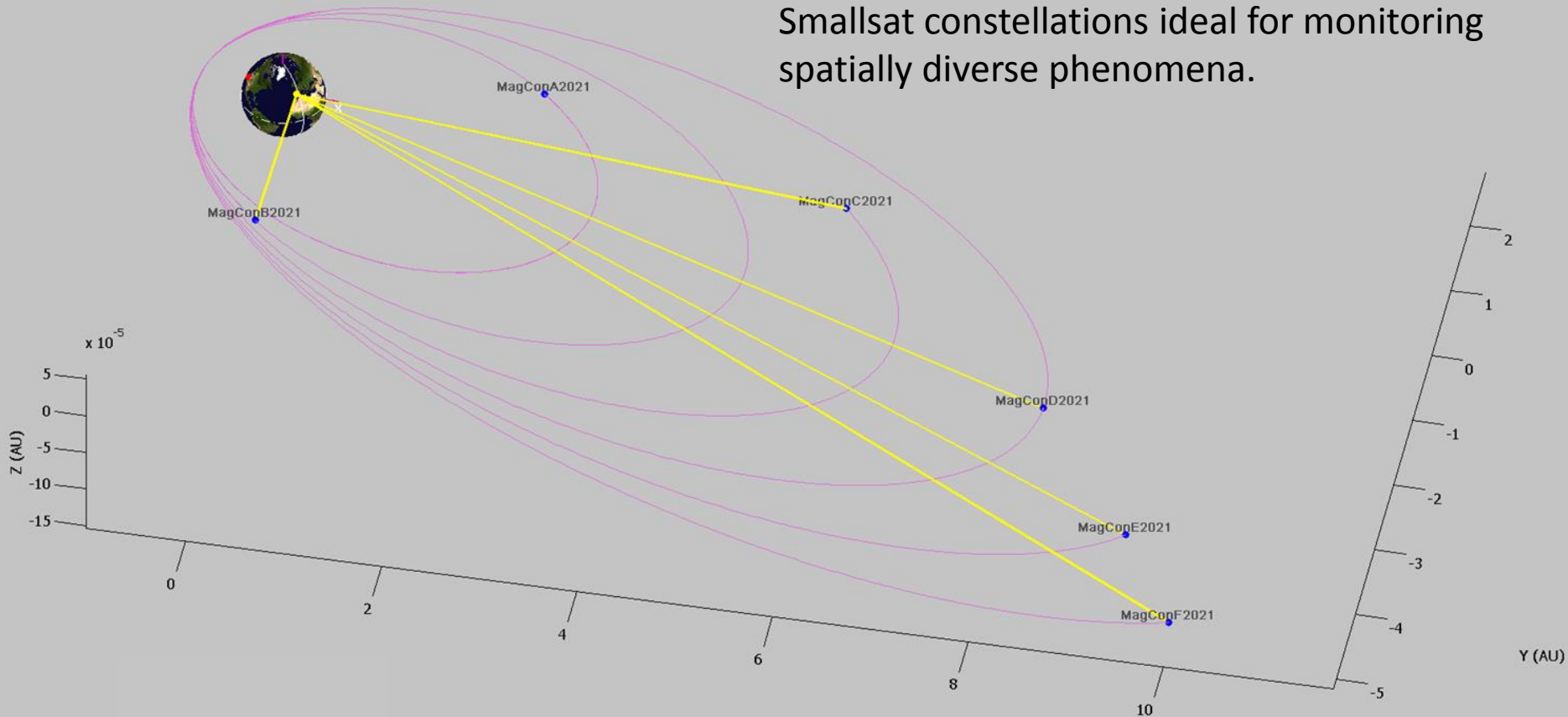
*While antenna time is actually “free” to a NASA mission once it has won a competitively-bid opportunity, the mission may be very concerned with the cost during the competition.*

\*Assumes one 8-hour 34m pass/day for a year.

# Smallsat Comm. Challenges (2/2)

## Smallsat Constellation Support

Smallsat constellations ideal for monitoring spatially diverse phenomena.



But, large inter-spacecraft distances can be problematic for proximity links, and direct communications with each spacecraft can tie up a large number of the DSN's antennas.

# Responding to the Challenges (1/2)

## Traditional MSPA (Multiple Spacecraft Per Antenna)

- Formally scheduled antenna sharing
  - Reduces downlink asset contention
  - Qualifies for lower aperture fees

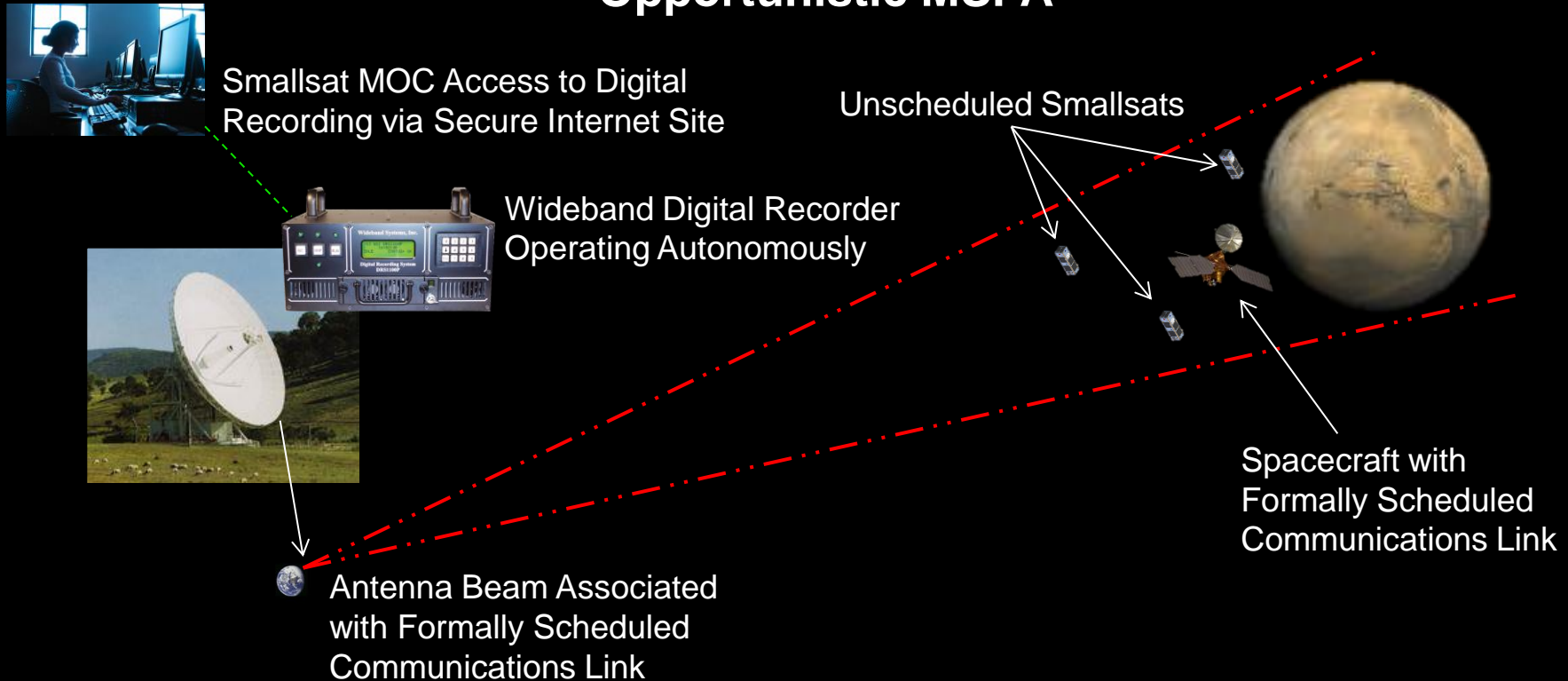


- Constraints
  - Supportable spacecraft number limited by receiver number.
  - Currently 2 applicable receivers per antenna.
  - Adding more receivers is a multi-million dollar endeavor.

*In today's budget-constrained environment, adding lots of receivers to all of the DSN's antennas is probably cost prohibitive.*

# Responding to the Challenges 2/2

## Opportunistic MSPA



*Everything received through the antenna beam is digitally recorded. Smallsats would transmit open loop when in a host spacecraft's beam. Smallsat MOCs would retrieve relevant portion of digital recording for subsequent demodulation and decoding.*

# OMSPA Pros and Cons

- Pros
  - Enables DSN to service a much larger number of missions.
  - Enables support of large constellation missions.
  - Minimizes cost to the smallsat projects.
  - Minimizes cost to DSN to implement and operate.
- Cons
  - Adds data latency.
  - Only applies to routine science downlinks.
  - Requires a “host” spacecraft’s ground antenna beam.
- Performance (see Bruce MacNeal’s analysis in appendix)
  - At average lunar distance, supports up to hundreds of Mbps, with 100-200 km separation from host.\*
  - At average Mars distance, supports up to a few kbps, with several Mars radii separation from host.\*

\*Assumes deep space, X-band smallsat with 0.5m HGA & 10 W transmitter.

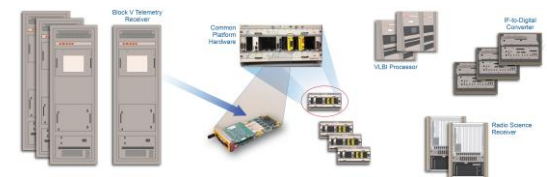
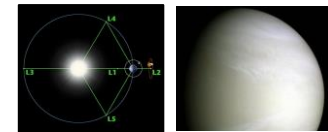
# What's Next?

- **Series of Demonstrations**

- FY'14 (Funded) – “Shadow” tracking of Mars spacecraft via Opportunistic MSPA.
  - Objective: Show readiness to proceed with demonstrations involving real smallsats.
- FY'15 – FY'17 (Not Yet Funded) – assuming deep space smallsat missions available in this timeframe, demonstrate ability to support.

- **Operational Capability?**

- Common Platform Era – DSN switch to selectable distribution of digitized IF; will enable dedicated antenna recording -- i.e., no competition for time on science and wideband VLBI receivers.





# Appendix

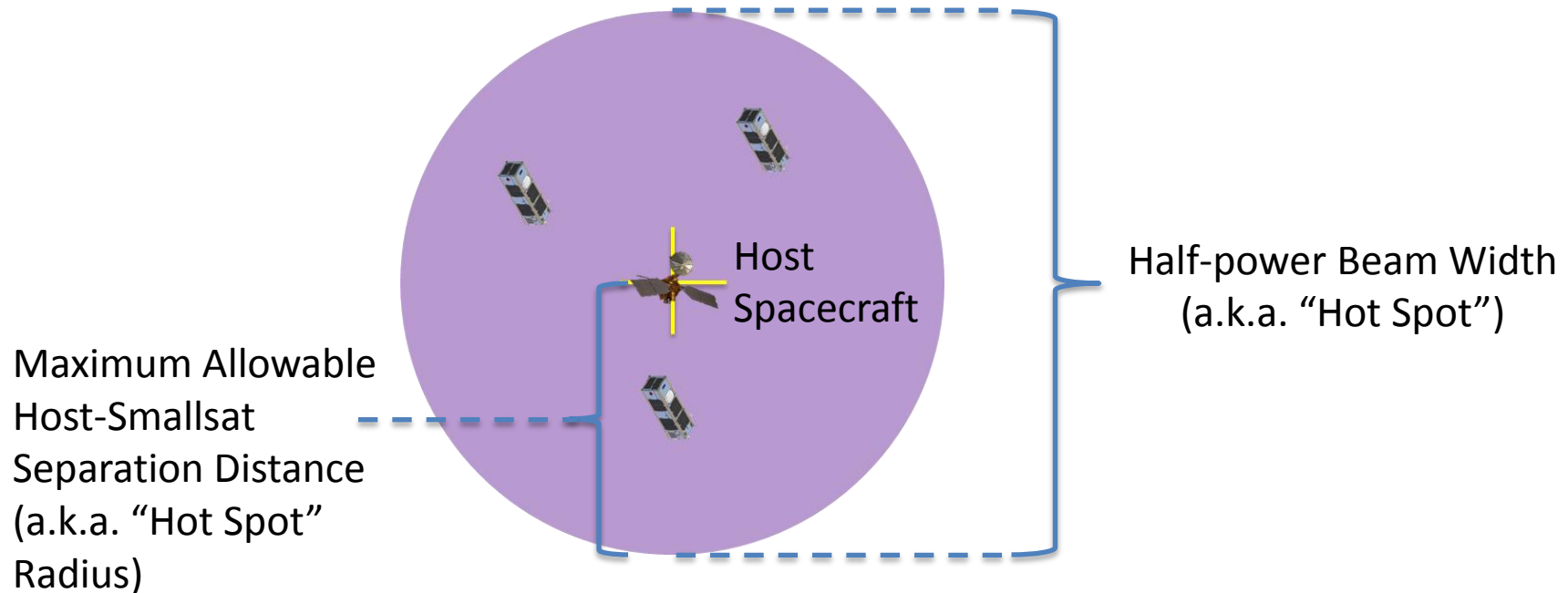
## **Performance Assessment**

by

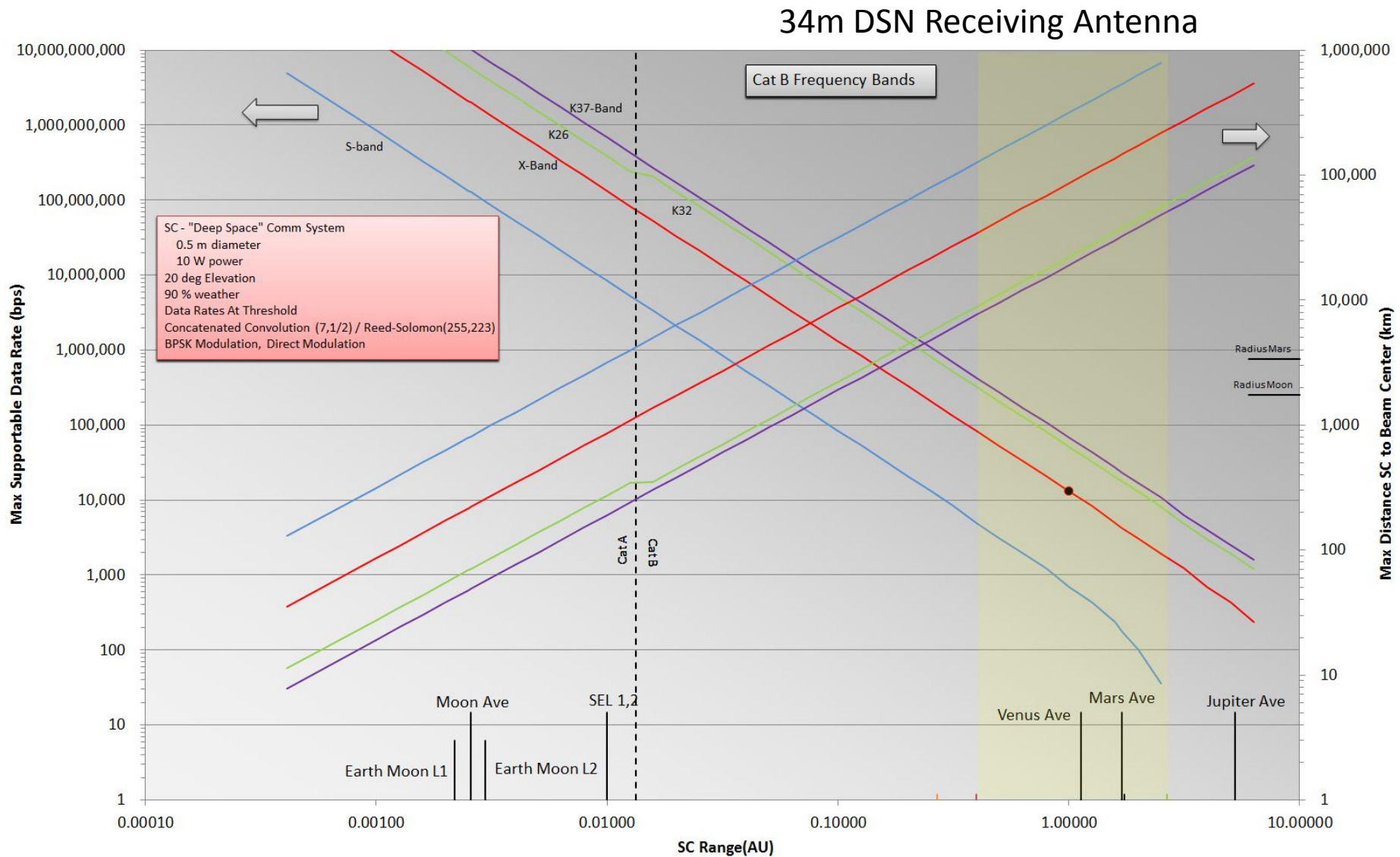
**Bruce E. MacNeal**

# Key Performance Questions

- Supportable Data Rate?
- Max. Distance from Tracked Target?



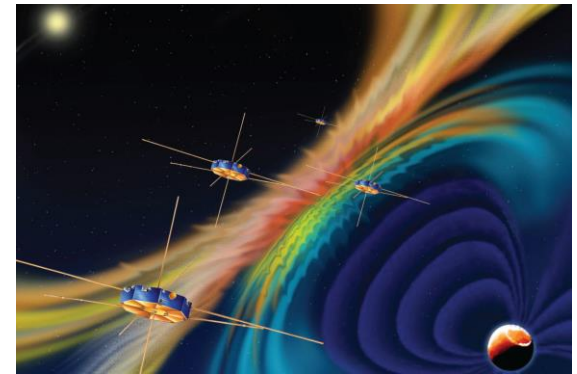
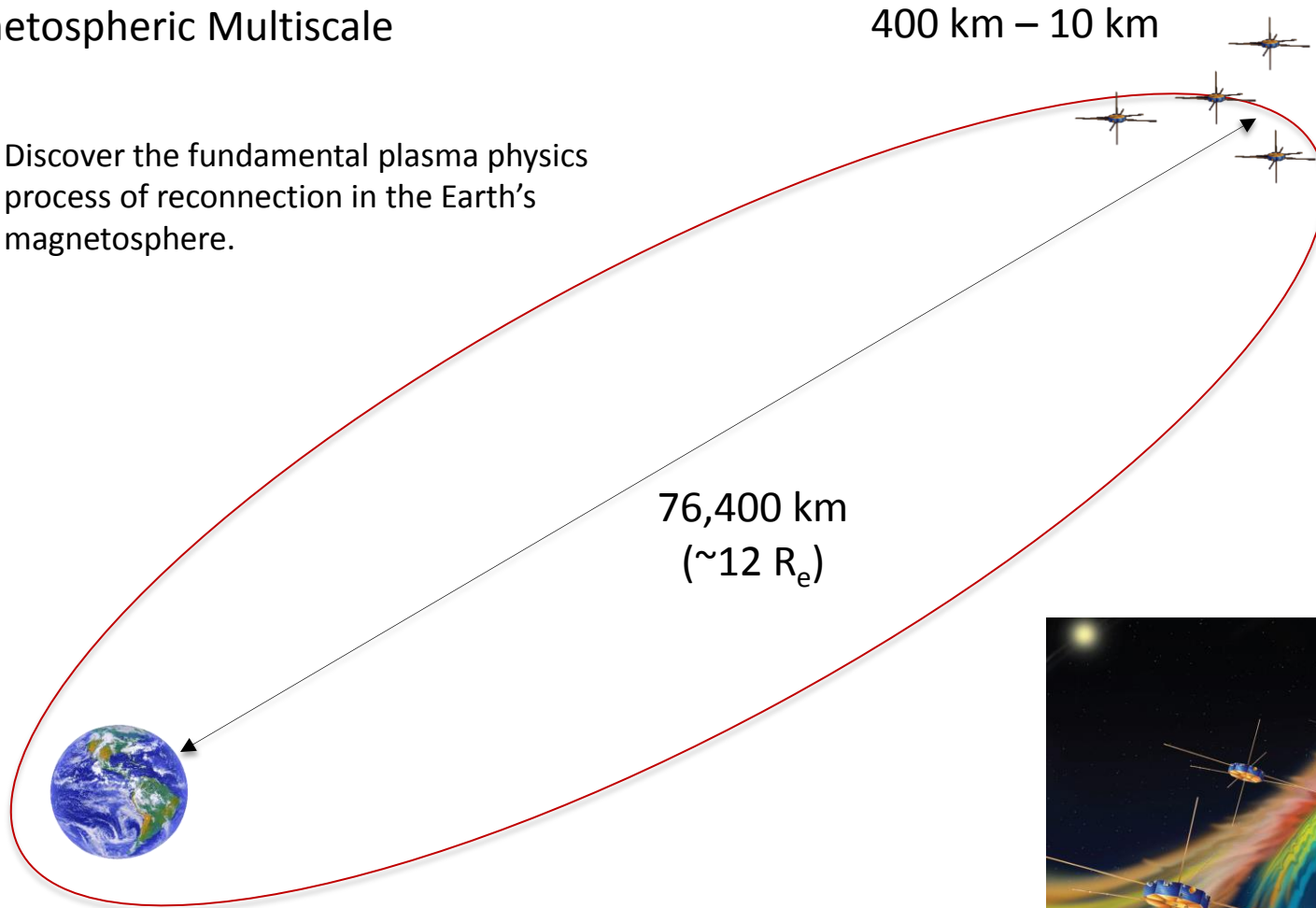
# Supportable Data Rate and "Hot Spot" Radius



# Example: MMS Constellation\*

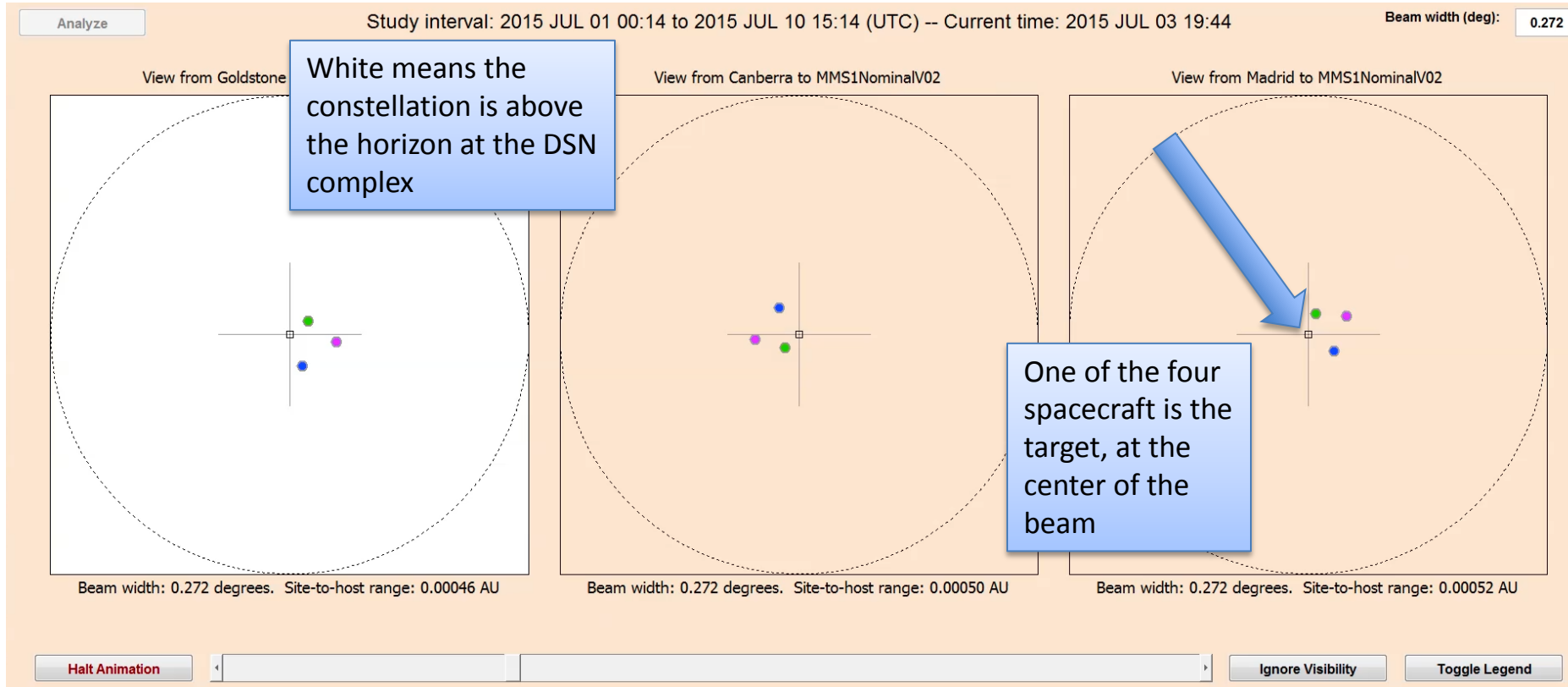
## Magnetospheric Multiscale

Discover the fundamental plasma physics process of reconnection in the Earth's magnetosphere.



# OMSPA Tool from OTIE: MMS\* Animation

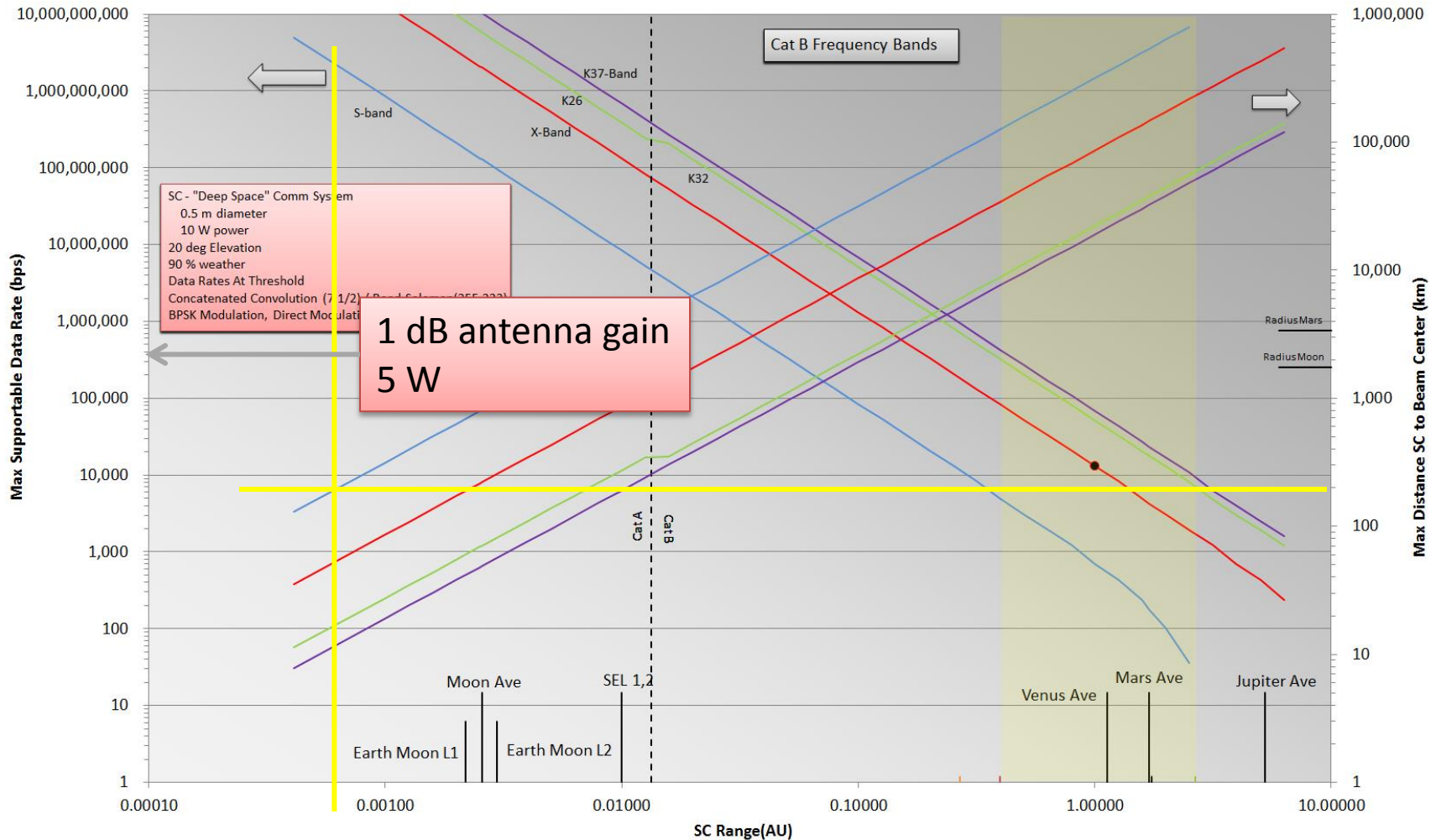
Credit: Dave Heckman



Beam intercept tool shows that all four spacecraft would be in the beam long enough to employ OMSPA for MMS\*

# Supportable Data Rate and "Hot Spot" Radius for MMS\*

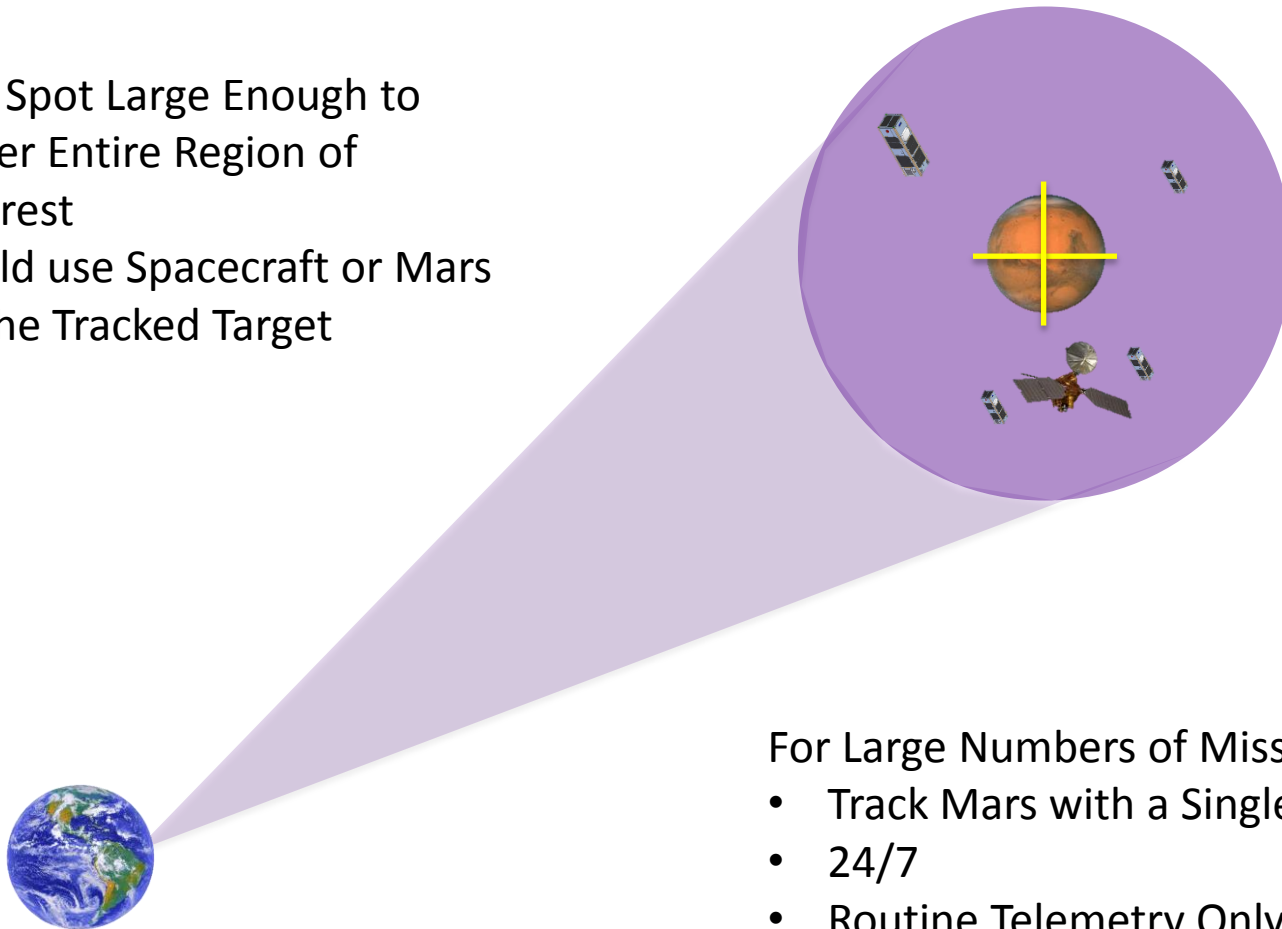
## 34m DSN Receiving Antenna



\*Pre-decisional – for Planning and Discussion Purposes Only

# Example: Mars

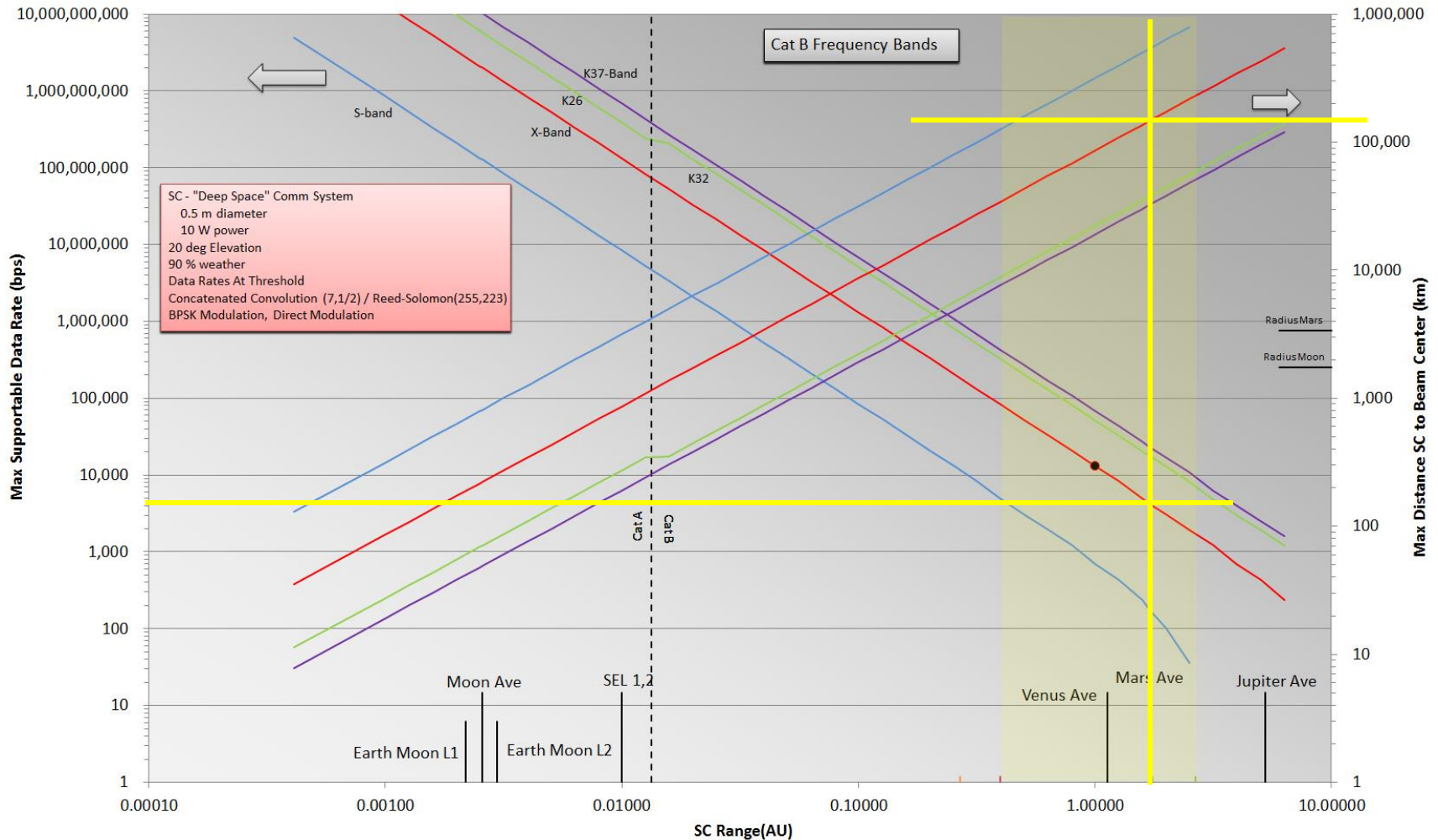
- Hot Spot Large Enough to Cover Entire Region of Interest
- Could use Spacecraft or Mars as the Tracked Target



- For Large Numbers of Missions at Mars:
- Track Mars with a Single 34m Antenna
  - 24/7
  - Routine Telemetry Only

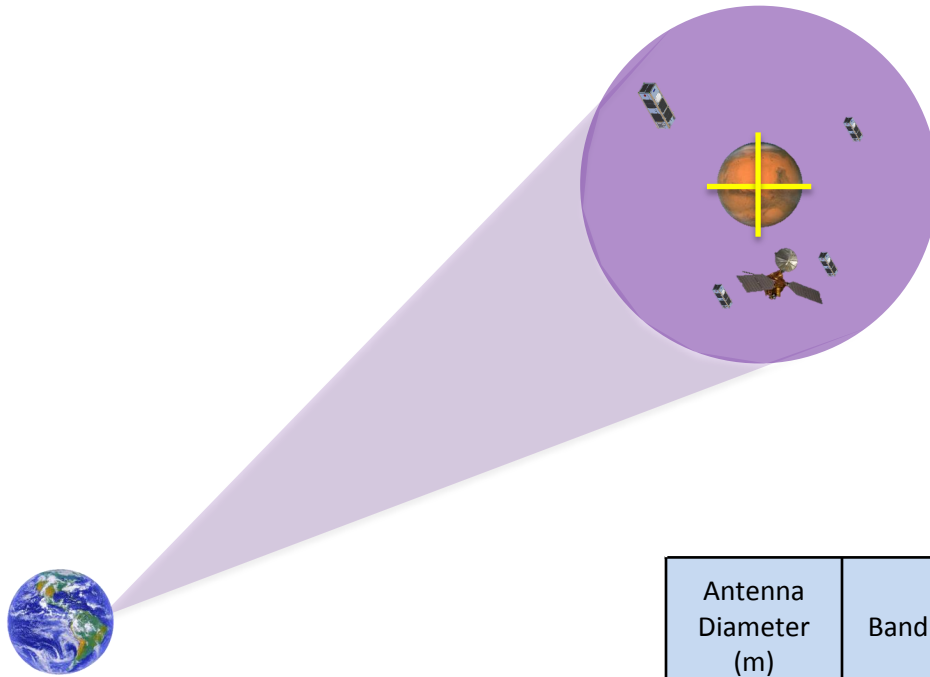
# Supportable Data Rate and "Hot Spot" Radius for Mars

## 34m DSN Receiving Antenna





# “Hot Spot” Mars



Antenna Diameter (m)	Band	Downlink Frequency (GHz)	HPBW (deg)	“Hot Spot” Diameter (Mars Diameters)		
				Mars Min	Mars Ave	Mars Max
70	S	2.295	0.1180	17	78	122
70	X	8.420	0.0320	5	21	33
34	S	2.295	0.2420	35	159	249
34	X	8.420	0.0660	10	43	68
34	K32	32.000	0.0170	2	11	18

HPBW – 810-005  
 Mars distances determined  
 Over next 50 years (Horizons).